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INTERACTIONAL AERODYNAMICS OF THE SINGLE ROTOR HELICOPTER CONFIGURATION

VOLUME VI-A - One-Third Octave Band Spectrograms of Wake Single Film Data Bildup to Baseline

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Approved for public release; distribution unlimited.

Prepared for

APPLIED TECHNOLOGY LABORATORY

U. S. ARMY RESEARCH AND TECHNOLOGY LABORATORIES (AVRADCOM)
Fort Eustis, Va. 23604

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APPLIED TECHNOLOGY LABORATORY POSITION STATEMENT

In 1975 a wind tunnel test program was conducted in the Boeing-Vertol 20-foot V/STOL Wind Tunnel on a 1/5th-scale UTTAS model to investigate and find solutions for several aerodynamic problems encountered during the UTTAS flight-testing. Specifically, these tests focused upon (a) the structure of the hub/rotor wake in the vicinity of the empennage, (b) the formulation of the ground vortex and its relation to hub loads and fuselage loads during transition, and (c) the occurrence of vibratory air pressures from the blade passing over the fuselage. Only portions of the above-mentioned wind tunnel test data were reduced and analyzed in addressing the flight-test problems of the UTTAS aircraft.

Under Contract DAAJ02-77-C-0020, Boeing-Vertol completed analyses on the data to understand more completely the aerodynamic interactions that are involved and to formulate instructions for the guidance of designers in these respects. The results of these studies are applicable to all existing and future single-rotor/tail rotor helicopters. The data have been segregated according to aerodynamic interactions and associated phenomena/problem areas. From this body of knowledge, a generalized set of design guidelines meaningful to the single-rotor helicopter design concept formulation were developed and are included in these reports.

Mr. Robert P. Smith of the Aeronautical Technology Division, Aeromechanics Technical Area, served as project engineer for this effort.

DISCLAIMERS

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PREFACE

The entire report describing the investigation of INTERACTIONAL AERODYNAMICS OF THE SINGLE-ROTOR HELICOPTER CONFIGURATION comprises eight numbered volumes bound as 33 separate documents. The complete list of these documents is as follows:

Volume I, Final Report

Volume II, Harmonic Analyses of Airframe Surface Pressure Data

- A Runs 7-14, Forward Section
- B Runs 7-14, Mid Section
- C Runs 7-14, Aft Section
- D Runs 15-22, Forward Section
- E Runs 15-22, Mid Section
- F Runs 15-22, Aft Section
- G Runs 23-33, Forward Section
- H Runs 23-33, Mid Section
- I Runs 23-33, Aft Section

Volume III, Flow Angle and Velocity Wake Profiles in Low-Frequency Band

- A Basic Investigations and Hubcap Variations
- B Air Ejector Systems and Other Devices

Volume IV, One-Third Octave Band Spectrograms of Wake Split-Film Data

- A Buildup to Baseline
- B Basic Configuration Wake Explorations
- C Solid Hubcaps
- D Open Hubcaps
- E Air Ejectors
- F Air Ejectors With Hubcaps; Wings
- G Fairings and Surface Devices

Volume V, Harmonic Analyses of Hub Wake

Volume VI, One-Third Octave Band Spectrograms of Wake Single Film Data

A - Buildup to Baseline

This volume is

B - Basic Configuration Wake Exploration

C - Hubcaps and Air Ejectors

Volume VII, Frequency Analyses of Wake Split-Film Data

- A Buildup to Baseline
- B Basic Configuration Wake Explorations
- C Solid Hubcaps

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D - Open Hubcaps

E - Air Ejectors

F - Air Ejectors With Hubcaps; Wings

G - Fairings and Surface Devices

Volume VIII, Frequency Analyses of Wake Single Film Data

A - Buildup to Baseline

B - Basic Configuration Wake Exploration

C - Hubcaps and Air Ejectors

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INTRODUCTION

Volume VI presents spectrograms of the six velocity measurements from the single film transducers which were located outboard on the wake rake to the left and right of the split film transducers. These plots are similar to those of Volume IV E, being machine plotted spectrograms in the one-third octave band format. They relate directly to the standard spectrograms that appear in Volume VIII for the same set of runs.

The sub-volumes of Volume VI display data derived from the following test runs:

Volume VI-A - 149, 150, 160, 156, 158, 159 Volume VI-B - 111 -119, 121,

Volume VI-C - 135, 136, 188, 211, 168, 167, 194, 161, 154, 172, 174, 176, 203, 205, 197

The runs follow the order of the logical arrangement of the Outline of Wake Investigations, Table 1, from which they have been selected. The Table I outline and other material is included for reference and as context to the work of each sub-volume. Table 2, the List of Test Runs, arranges the runs in numerical order and gives pertinent text parameters.

The Index of Rake Positions, Table 3, lists the hot film transducer rake positions in the model coordinate system for each run and its test points. The main feature of Table 3 is the indexing of the test point number to the model water line station and butt line as it varied from run to run. The table groups the runs as they shared the indexing correspondence of point with position. It is emphasized that the runs in a group do not necessarily all share the same number of test points but they do have same correspondence within their respective ranges of test points.

The orientation of the rake is shown pictorially in Figures 1 through 6 for the various test runs. Figure 7 presents a scaled drawing of the model with reference to the three-axis coordinate system. Table 4 lists the center frequency and the upper and lower band limits for each of the numbered one-third octave bands.

TABLE 1			
OUTLINE OF WAKE IN Description	Configuration Code		Base- line
Build-up to Baseline			garden II
1. Nacelles removed	K ₁₃ +H ₁ -N	149	150
2. Blades off, rotating hub	K ₁₃ -M+H _{1.0}	160	156
3. " , non-rotating hub	$K_{13}-M+H_{1.0}$	158	156
4. " , hub off	K ₁₃ -M-H _{1.0}	159	156
Basic Configuration 1. Wake Explorations near Empennage (a) 15" Long. + traverse at T/R C.L. (b) 9" Vert. + " above T/R " (c) 2" " " in vortex (d) 8" " " (continue 112) (e) 13" " behind stab. (f) Lateral traverse, left stab. (One T.P. only) (g) Same continued (h) Same continued (One T.P. only) (i) Lateral traverse right stab. (j) T/R effect on wake	K ₁₁ " " " " " " " K ₁₁ +T ₂ ⁰	111 112 113 114 115 116 117 118 119 121	 115
2. Climb/Descent Studies (a) Climb 900 FPM (b) Descent 800 FPM	K _{1 1}	135 136	===
Effect Of Hub Caps			
1. Solid Caps on Canister			
(a) 7.6" diam. 2.17" ht. soft Pitch Arms	K ₁₁ -H _{1.0} +H _{1.2}	137	136
(b) 7.6" diam. 2.17" ht. stiff Pitch Arms	K ₁₃ +H _{1.2}	153	156
(b) 7.6" diam. 2.45" ht. flt. test config.	K ₁₃ +H _{1.2.1} +I ₁ +E _{1.0}	207	188

TABLE 1 (CONTIN			
OUTLINE OF WAKE INVES	STIGATIONS		
Description	Configuration Code*	Run No.	Base- line
Effect of Hub Caps (Continued)			
2. Solid Caps Raised Above Canister		read.	
(a) 7.6" diam. 2.45" ht. 70"	H _{1.2.2} +I ₁ +E _{1.0}	208	188
depth, .55 gap (b) 10.0" diam. 3.25" ht. 1.55" depth, .50" gap	H _{1.8.1} +I ₁ +E _{1.0}	189	188
(c) 10.0" diam. 4.125" ht. 2.05" depth, .875" gap	H _{1.8.2} +I ₁ +E _{1.0}	190	188
(d) Repeat of 189	" " "	210	188
3. Open Caps Without Underbody			
(a) 10.0" diam. 1.25" gap,blades (b) " " gap, no blades	$H_{1.0.1}^{1.0.2} + I_{1.0}^{+E_{1.0}}$	193 166	188/166 158
(c) " " 2.05" gap, blades (d) " " 1.75" gap, no	H _{1.14.1} +I ₁ +E _{1.0} H _{1.0.1} -M	211 165	188 158
(e) " " 1.87" gap,blades (f) 16" diam. 2.00" gap,blades (g) " " gap, no blades	H _{1.0.3} +I ₁ +E _{1.0} H _{1.7.1} -M	191 168 167	188 156/167 158
	H _{1.7.2}	169	156
4. Open Caps with Underbody	8000 CO 476		
(a) 7.6" diam. 1.25" gap (b) " " " " center post	1	194 198 202	188 188 194
(d) 10.0" diam5" gap, no blades	H _{1.5.1} -M	164	158
(e) " " 1.25" gap, no blades	H _{1.5.2} -M	161	158
(f) " " 2.0" gap, no blades	H _{1.5.4} -M	163	158
(g) " " 4.0" gap, no blades	H _{1.5.3} -M	162	158
(h) " " 1.25" gap	H _{1.5.2}	154	156/161
*Basic Code is K13.	, m.2.2,500 d	165	

TABLE 1 (CONTINUED)

OUTLINE OF WAKE INVESTIGATIONS

			
Description	Configuration Code*	Run No.	Base- line
5. Miscellaneous Hub Covers (a) Hub fairing 16" diam. (b) Wham-O-Frisbee 10" diam. (c) Fab. glass Frisbee 16" diam.	H _{1.3} H _{1.9.0} +E _{1.2} H _{1.9.1} +E _{1.2}	151 182 183	150 181 181
Effect of Air Ejectors			
1. Basic system no blowing 2. " " 40 psi 3. " " 150 psi 4. Wide chord shroud 40 psi 5. Wide " " 150 psi 6. W/C shroud w. lip 40 psi 7. Same Contoured Parallel 150 psi 8. Bifurcated duct 0 psi 9. " " 40 psi 10. " " 150 psi	H _{1.0} +E _{1.0} H _{1.0} +E _{2.5.1} H _{1.0} +E _{3.5.2} H _{1.0} +E _{3.5.4}	173 174 175 176 184 187 203 204	156 156/172 156/173 156/174 156/173 156/174 156 156/203 156/203
Air Ejectors with Open Hub Caps with Underbodies			
7. " " " 150 psi	H _{1.11.1} +1 ₂ +E _{1.0} """ H _{1.11.1} +I ₂ +E _{4.0} """ H _{1.11.2} +I ₂ +E _{4.6} H _{1.5.4} +E _{2.5.1}	195 196 197 198 199 200 201	188/172 188 188/173 188/174 188/194 188/196 188/196 188/200 156/176
Effect of Wings and Misc.			
1. Wings (a) Nacelle-mounted stub wing (b) Single slotted flapped wing (c) Dougle slotted flapped wing (d) Boom-mounted stub wing	H _{1.0} +W _{1.0} +E _{3.1} H _{1.0} +W _{3.0} +E _{1.0} H _{1.0} +W _{2.0} +E _{1.0} H _{1.0} +W _{4.0}	178 180 179 186	181 181 181 156

TABLE 1 (CONTINUED)

OUTLINE OF WAKE INVESTIGATIONS

	Description	Configuration Code*	Run No.	Base- line
2.	Crown Fairings (a) Flat top behind shaft (b) Round top behind shaft (c) Extended flat top fairing (d) Flat top + 16" cap, 4" gap (e) Forward fairing/nacelle fairing	K _{1 1} +D ₁ K _{1 1} +D ₂ H ₁ +D ₄ H _{1 • 7 • 2} +D ₄ P _{1 • 0}	140 141 170 171 152	138 138 156 170 156
3.	Surface Devices (a) Vortex generators (b) Guidevane between nacelles (c) Longitudinal strakes (d) 14% porosity spoiler	K ₁₁ +VG ₂ , 1 K ₁₁ +FV ₁ H _{1•5,3} +S ₄ K ₁₁ +X ₁	139 142 155 143	138 138 156 138

*Basic Code is Kl3 unless noted otherwise.

TABLE 2
LIST OF TEST RUNS
BASIC INVESTIGATIONS OF THE HUB WAKE

								-
RUN	CONFIGURATION	Vrun	RPM	DISK	MODEL	MODEL	MR HT.	TAIL
NO.		KNOTS	MR/TR	psf psf	a a	o /h	h/d	ROTOR
111	$ m K_{11}/15"$ Long, wake traverse at TR center line	80	1433/0	8	0.9	-2.0	8	Off
112	" /9" Vert. wake traverse above TR center line			=	E	=	=	=
113	" /2" Vert traverse through MR vortex	=	-	"	=			
114	" /8" Vert. traverse below TR center line	=			=	=	=	
115	" /13" Vert. traverse behind stabilizer		=	=	=	=	=	=
116	" /Lateral traverse - left stabilizer			u u	=	=	=	=
117	" /116 continued	п	=	"	=	=	-	=
118	" /116 continued			=	-	=		=
119	" /Lateral traverse - right stabilizer	H	=	-	=	E	=	=
121	K ₁₁ +T ₂ /Effect of tail rotor flow on wake	=	1433/ 4500	=	=	E	=	u0
135	K ₁₁ /Wake in 900 fpm climb	=	=	=	0.9-	-4.5	=	Off
136	" /Wake in 800 fpm descent	=	=	=	0.9	-2.0	=	=

TAGLE 2 (CONTINUED)

LIST OF TEST RUNS EVALUATION OF WAKE-ALTERING DEVICES

RUN	CONFIGURATION/CONDITTION	Vrun	RPM	DISK	MODEL	MODEL	MR HT.	TAIL
NO.		KNOTS	MR/TR	psf.	0 8	• 1	p/q	ROTOR
137	K11-H1.0+H1.2/Effect of 7.6 inch diam. solid hub cap	80	1433/0	8	9	-3.8	8	Off
138	K ₁₁ /Repeat of base run	=	=	=				
139	K ₁₁ +VG _{2,1} /Effect of vortex gener- ators on aft crown	=	=	=	=	=	=	
140	K ₁₁ +D ₁ /Flat-topped "doghouse" fair- ing on aft crown		=	=	=	=	=	
141	K ₁₁ +D ₂ /Rounded-top fairing	=	=	=			=	
142	K ₁₁ +FV ₁ /Deflection vane on crown between nacelles	=	=	=	=		=	=
143	κ_{11} + κ_{1} /Variable porosity spoiler		=			=		
149	$\kappa_{13}^{+H_1-N_1}$ /Effect of nacelles off also add stiff pitch arms (κ_{13})	60	1075/0	4.5	í	=	=	•
150	$K_{13}+H_{1}/60$ knot baseline			-		=	=	
151	K ₁₃ +H _{1.3} /16 inch diam. helmet fair- ing		=	=	=	=	=	-
152	K ₁₃ +P _{1.0} /Pylon and intake fairings	80	1433/0	8	=	=	=	
153	$\mathrm{K}_{13}{}^{+\mathrm{H}}_{1.2}/\mathrm{Repeat}$ 137 with K_{13} pitch arms		Е		=	=	=	

	LIST OF TEST RUNS					
EVALUATION OF WAKE-AI	WAKE-ALTERING DEVICES	CES				
VATORION/COMPTETON	Vrun	DISK	MOI	MODEL	MR HT.	TATT
	<u>Σ</u>	LDG.	e	0 =	h/d	ROTOR
open hub cap, 7" 8	80 1433/0	8	9	-3.8	8	Off
	=	=	=	=		=
K ₁₃ +H _{1.0} /Baseline with K ₁₃ ,i.e., stiff pitch arms	=	=	=	=	-	=
studies with blades hub not rotating	0/0	=	=	=	=	=
studies with hub off	:	=	=	=	=	=
K ₁₃ -M+H _{1.0} /Same as 158 except hub is rotating	" 1433/0	=	=	2	=	=
K ₁₃ -M+H _{1.5.2} /Repeat of 154 without blades	0/0 "	E			=	=
as 161 except 4"	=		=			=
as 161 except 2"	=	=	•	=	=	=
as 161 except	=	:	-			=
K ₁₃ -M+H _{1.0.1} /10" open hub cap, no underbody, same cap vert, position as Run 154	=	=	=	=	=	=
K ₁₃ -M+H _{1.0.2} /Same as 165 with cap lowered by 0.5"	=	=	-	=	=	-

		DEVICES
(UED)	RUNS	RING
TABLE 2 (CONTINUED	LIST OF TEST RUNS	-ALTE
LE 2 (OF	WAKE-
TA ₃	LIST	OF
		EVALUATION OF WAKE-ALTERING DEVICES

				-	1			
RUN	CONFIGURATION/CONDITITION	VTUN	RPM	DISK	MODEL	MODEL	H.	TAIL
NO		KNOTS	MR/TR	psf.	a a	ф ф	h/d	ROTOR
167	K ₁₃ -M+H _{1.7.1} /16" open cap, no underbody, 2" gap	- 80	0/0	8	9	-3.8	8	Off
168	K ₁₃ +H ₁ ,7,1/B ₁	=	1433/0	=		=	=	=
169	K ₁₃ +H _{1.7.2} /16" open cap, no under- body, 4" gap	=	=		=	=	=	=
170	K ₁₃ +H _{1.0} +D _{4.0} /Extended flat top fairing on aft crown	=		н	=	=	=	=
171	K ₁₃ +H _{1.7.2} +D _{4.0} /Same fairing as 170 same cap as 169		=	=	=	=	=	
172	K13+H1.0+E1.0(Opsi)/Basic air ejector zero blowing baseline	и	=	n.	=	-	=	
173	K ₁₃ +H _{1.0} +E _{1.0} (40 psi)/Same as 172 with 40 psi supply	=	=	ш	=	ш	=	=
174	K ₁₃ +H _{1.0} +E _{1.0} (150 psi)/Same as 172 with 150 psi supply	=	u	=	=	=	=	
175	Kl3+Hl, 0+E2,5,1(40 psi)/ wide chord shroud		=			=	=	
176	K13+H1.0+E2.5.1(150 ps: with 150 p	=	=	=	=		=	=
177	K ₁₃ +H ₁ .5 ₁ 4+E ₂ 5 ₁ (150 psi)/Same as	=	=	=	=	=	=	=
178	K ₁₃ +H _{1.0} +W _{1.0} +E _{1.1} (0 psi)/Nacelle mounted wing	=	=	=	=	=	=	=

	TABLE 2 LIST OF	TABLE 2 (CONTINUED)	(G					
	EVALUATION OF WAKE-ALTERING DEVICES	-ALTERING	3 DEVICE	S				
NCC	CONFIGURATION	VTUN	RPM	DISK	MODEL	MODEL	MR HT.	TAIL
		KNOTS	MR/TR	rbg.	a,	o /h	þ/q	ROTOR
K13+H1.0+	K ₁₃ +H _{1.0} +W _{2.0} +E _{1.0} (0 psi)/Double slotted flapped wing	80	1433/0	8	9	-3.8	8	Off
K13+H1.0+	K ₁₃ +H _{1.0} +W _{3.0} +E _{1.0} (0 psi)/Single slotted flapped wing	=	=		-	=	=	=
K13+H1.0+	K ₁₃ +H _{1.0} +E _{1.2} (0 psi)/Baseline with ejector tube moved aft	=	н	-		=	=	=
K13+H1.9.0+E1	0 ^{+E} 1.2 (0 psi)/Standard 10" frisbee			=			=	=
K13 ^{+H} 1.9.1 ^{+E} 1.	,2 (0 psi)/ cated	=	ш	=	-	=	=	=
K13+H1.0+E3.5.	2 Wi	ш		-	=	=	=	=
K ₁₃ +H _{1.0} +E _{3.5} .	E3.5.2 (150 psi)/Same as 184 with 150 psi air	u		=	=	=	=	=
K13+H1.0+W4.0/	-W4.0/Boom mounted stub wing	=			=	=	=	=
K13+H1.0+E3.5.	+E3.5.4 (150 psi)/Like 185 with modified shroud			=	=	=		=
K13+H1.0+I1+E1	$+I_1+E_1$, (0 psi)/Baseline with I, instr. ring			E	=	=	=	=
K ₁₃ +H _{1.8}	K ₁₃ +H _{1.8.1} +I ₁ +E _{1.0} (0 psi)/Solid cap, 10" diam. 3.25" height			=	=	=	=	-
K13+H1.8.2+I1+	.2+I1+E1.0 (0 psi)/Same as	= 			=			
	TAN EXCEPT 4.12 HELGING	NC.						

			TAIL	ROTOR) Jjo	=		u	=		=		=	=	=	=
			MR HT.	h/d	8			u			=		•	=	=	=
			MODEL	0	-3.8	=		u	=		=	=	=		=	=
			MODEL	• 8	9	=	-	=	=		=		=	=	=	=
		Ø	DISK	rDG,	œ	=		=	=					=		=
0	S	DEVICE	RPM	MR/TR	1433/0	=		E	=		=			=	=	=
TABLE 2 (CONTINUED)	FEST RUN	ALTERING	V _{TUN}	KNOTS	80	=	=							=		=
TABLE 2	LIST OF TEST RUNS	EVALUATION OF WAKE-ALTERING DEVICES	NOTH TOWOO, NOTH ROLL ON CO.		K13+H1.0.2+I1+E1.0 (0 psi)/10" cap, no underbody, 1.87" gap	K13+H1.0.2+I1+E1.0 (0 psi)/10" cap, no underbody, 1.25" gap	Kl3+H1.11.1+I2+E1.0(0 psi)/7.6" cap, underbody, 1.25" gap	K13+H1.11.1+I2+E1.0(20 psi)/Same as 194 with 20 psi air	K13+H1.11,1+12+E1,0(40 psi)/Same as 194 with 40 psi air	K13+H1.11.1+I2+E1.0(150 psi)/Same as 194 with 150 psi air	K13+H1.11.1+I2+E4.0 (0 psi)/Same as 194 except blowing tube 2" aft	K13+H1.11.1+I2+E4.0 (40 psi)/Same as 198 with 40 psi air	K13+H1.11.1+I2+E4.0 (150 psi)/Same as 198 with 150 psi air	K ₁₃ +H _{1.11.2} +I ₂ +E _{4.0} (150 psi)/Same as 200 except center support cap	.2+I2/Baseline	K ₁₃ +H _{1.0} +E _{5.0} (0 psi)/Bifurcated air duct baseline
			RUN	NO.	191	193	194	195	196	197	198	199	200	201	202	203

TABLE 2 (CONTINUED) LIST OF TEST RUNS EVALUATION OF WAKE-ALTERING DEVICES

ROTOR	Off	-	=	=	=	=						
h/d	8	=	=	=	=	=		1.44				
9	-3.8	=	=	=	=	=						
8	9	=	=	=	=	=						
LDG. psf	8	=	=	=	-	•						
MR/TR	1433/0	=	=		=							
KNOTS	80	=	ت ت	•	-	=						
NOTITION (NOTITION THOS	Kl3+Hl.0+E5.0 (150 psi)/Bifurcated duct with 150 psi air					Kl3+Hl.14.1+I1+El.0 (0 psi)/Like 189 and 210 except cap is open						
NO.	204	205	207	208	210	211		3.L 3.L				
	MR/TR DSf a w	K ₁₃ +H _{1.0} +E _{5.0} (150 psi air Riot with 150 psi air duct with 150 psi air duct with 150 psi air duct with 150 psi air ω ω ω ω ω ω ω ω ω	Kl3+H1.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8 π π π π π π π π π	Kl3+H1.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8	NO. Control of the	Kl3+H1.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8	NO. CONTINUAL LONG KNOTS MR/TR LDG. φ* h/d 204 K13+H1.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8 ∞ 205 K13+H1.0+E5.0 (40 psi)/Same as 204 " " " " " " " 207 K13+H1.2.1+I1+E1.0 (0 psi)/7.6" solid "	NO. Contribution Contribution KNOTS MR/TR LDG. φ* h/d 204 Kl3+H1.0+E5.0 (150 psi air 205 Mith 150 psi air " " " " h/d 205 Kl3+H1.0+E5.0 (40 psi)/Same as 204 "<	NO. CONTINUAL TOTAL TOT	NO. KNOTS MR/TR psf. d** \psi** h/d 204 Kl3+H1.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8 \sigma 205 Kl3+H1.0+E5.0 (40 psi)/Same as 204 " " " " " " 206 Kl3+H1.0+E5.0 (40 psi)/Same as 204 " " " " " " 207 Kl3+H1.2.1+I1+E1.0 (0 psi)/T.6" solid " " " " " " 208 Kl3+H1.2.2+I1+E1.0 (0 psi)/Same as " " " " " " 210 Kl3+H1.15.1+I1+E1.0 (0 psi)/Repeat " " " " " " 211 Kl3+H1.15.1+I1+E1.0 (0 psi)/Like " " " " " " " 212 Kl3+H1.14.1+I1+E1.0 (0 psi)/Like " " " " " " " " " " " " " " " " " " "	NO. CARIATORICALIDON KNOTS MR/TR psf a° ψ° h/d 204 Kl3+H1.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8 ∞ adact with 150 psi air " " " " " " " " " " " " " " " " " " "	NO. CONTROLLING RNOTS MR/TR psf a° ψ° h/d 204 Kl3+Hl.0+E5.0 (150 psi)/Bifurcated 80 1433/0 8 6 -3.8 ∞ duct with 150 psi air 205 Kl3+Hl.0+E5.0 (40 psi)/Same as 204 " " " " " " 207 Kl3+Hl.2.1+I+E1.0 (0 psi)/7.6" solid " " " " " " 208 Kl3+Hl.2.1+I+E1.0 (0 psi)/Same as 204 " " " " " " 208 Kl3+Hl.2.1+I+E1.0 (0 psi)/Repeat " " " " " " " 210 Kl3+Hl.15.1+I+E1.0 (0 psi)/Repeat " " " " " " " " 211 Kl3+Hl.14.1+I+E1.0 (0 psi)/Like " " " " " " " " " " " " " " " " " " "

TABLE 3

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
111	20 21 22 24 26 28 30 32 34 36	53.5	103.1 105.0 107.0 109.0 111.0 112.9 114.9 116.9 118.9	-7.25	1
112	2 4 6 8 10 12	48.9 50.8 52.7 54.5 56.2 57.2	107.3	-7.25 " " "	1
113	2 4 6 8 10 11	51.7 52.3 52.8 53.3 53.9 53.3	103.3	-3.25 " " "	1
114	2 4 6 8 10	44.5 46.4 48.2 50.0 51.9	103.0	-3.25 " "	1
115	3 4 6 9 10 12 14 16 18 20	52.9 52.0 50.0 48.0 46.0 44.1 42.1 53.0 54.0	124.7	-3.25 " " " "	1

TABLE 3 (CONTINUED) INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT LINE	LOCATION FIGURE
116	7	36.9	100.5	-17.5	1
117	2 4 6 8 10	37.6 37.3	100.5 99.6	-16.0 -14.0 -12.0 -10.0 - 8.0	1
118	2	37.6	100.5	- 6.0	1
119	2 5 8 9 14 16 20 25	37.3 "" "" 51.5 52.3	99.6 " " " 102.5 101.7	+ 6.0 8 10 " 14 16 17.5 -17.5	1
121	3 4 6 8 10	62.9 53.5 50.1 46.0 42.1	129.0	+ 5.7	2
135	2 4 6 8 10 12 14	56.9 54.5 52.5 50.5 48.5 46.5 44.5	106.3	- 5.7 "	3
136	2 4 6 8 10 12 14 17 18 19	56.5 54.5 52.5 50.6 48.5 46.5 44.5 37.1 39.0 41.0	104.0	- 8.0	4

TABLE 3 (CONTINUED) INDEX TO RAKE POSITIONS

	•	DEX TO KAI			
RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT LINE	LOCATION FIGURE
137	3 5 7 9 11 13 15 17	38.7 39.9 42.0 44.0 46.0 48.0 50.0 52.0 54.0	98.4 100.5 103.6	- 8.0	5
138-41, 143	2 3 4 5 6 7 8 9	38.8 40.0 42.0 44.0 46.0 48.0 50.0 52.0 54.0	98.4 100.5 103.6	- 8.0	5
142	7 8 9 10 11 12 13 14 15 16	37.8 40.2 42.0 44.0 46.0 48.0 50.0 52.0 54.0 56.8	98.4 " 100.5 103.6	- 8.0 "" "" ""	5
			2.25 2.25 2.25 2.25 2.25 2.25 2.25 2.25		

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST POINT	WATER LINE	MODEL STATION	BUTT	LOCATION FIGURE
149-151	2 3 4 5 6 7 8 9	38.8 40.0 42.0 44.0 46.0 48.0 50.0 52.0 54.0	98.5 100.6 103.5	- 8.0 " " "	5
152-6, 158 161-4, 166 167, 169-71 175, 177-9 180,182,184 186-8, 190 191,193,194 196,198,201 204,207,208	5 6 7 8 9	42.9 44.9 46.9 48.9 50.9 52.9 54.9 56.9	97.9 100.6 104.6	0.0	6
159	1 2 3 4 5	54.9 52.9 50.7 48.6 46.7	104.6	0.0	6
160,203	5 6 7 8 9 10	42.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 100.6 104.6	0.0	6
165	3 4 5 6 7 8	44.9 42.9 46.9 48.9 50.9 52.9	97.9 100.6 104.6	0.0	6

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER	TEST	WATER LINE	MODEL STATION	BUTT LINE	LOCATION FIGURE
168, 183	4 5 6 7 8 9	42.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 100.6 104.6	0.0	6
172	3 4 6 7 8 9 10	42.9 44.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 " 100.6 104.6	0.0	6
173,174,176 185,195,197 199,200,205 210	1 2 3 4 5 6 7	42.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 100.6 104.6	0.0	6
181	2 3 4 5 6 7 9 10 11 12 13	42.9 44.9 46.9 48.9 50.9 52.9 54.9	97.9 100.6 104.6	0.0	6
		9.391	0.52 9.58 0.05 4.52		

TABLE 3 (CONTINUED)

INDEX TO RAKE POSITIONS

RUN NUMBER 189	TEST POINT	WATER LINE	MODEL	BUTT	LOCATION
100		LINE	STATION	LINE	FIGURE
109	29 30 31 32 33 34 35 36 37 38 39	42.9 44.9 46.9 48.9 50.9 50.9 52.9 54.9	97.9 100.6 " 104.6 100.6 104.6	0.0	6
202	3 4 5 6 7	43.4 44.9 46.9 48.9 50.9	97.9 100.6 104.6	0.0	6

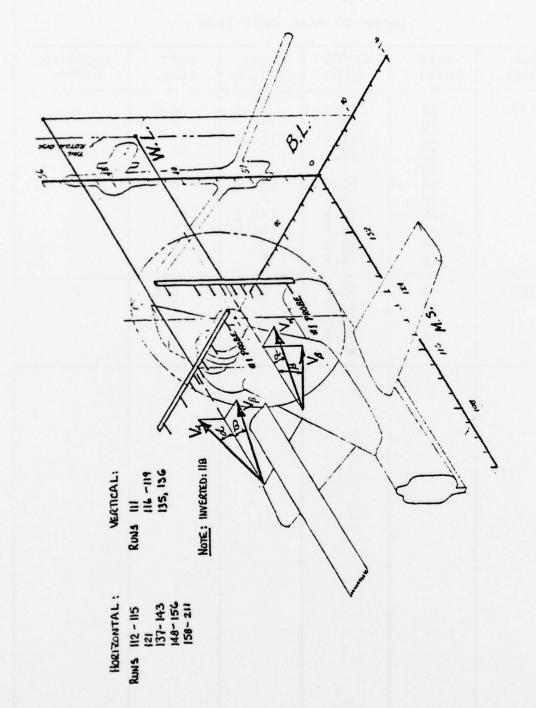


FIGURE 1 - RAKE ORIENTATION DIAGRAM

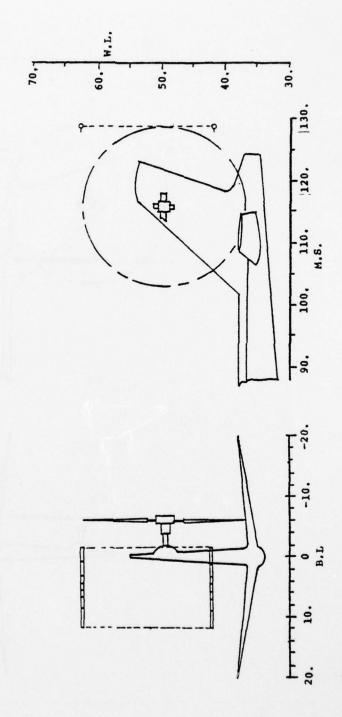


FIGURE 2 -HOT FILM RAKE LOCATIONS

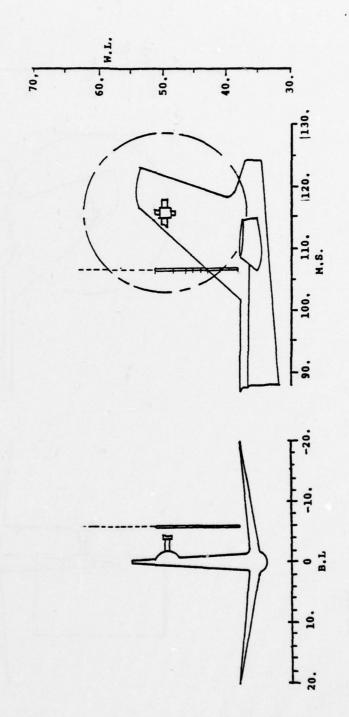


FIGURE 3 -HOT FILM RAKE LOCATIONS

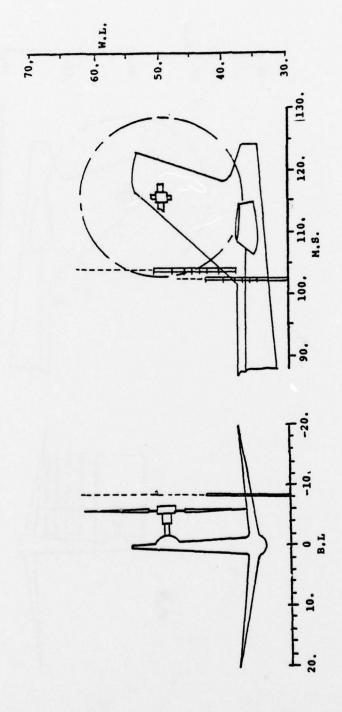


FIGURE 4 -HOT FILM RAKE LOCATIONS

RUN 137, 138, 139, 140, 141, 142, 143, 148, 149, 150, 151

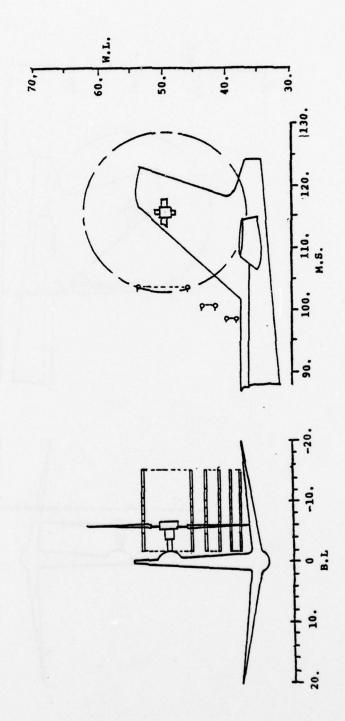


FIGURE 5 -HOT FILM RAKE LOCATIONS

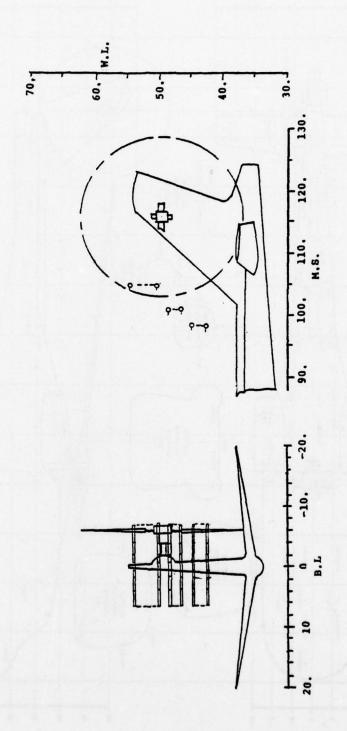


FIGURE 6 -HOT FILM RAKE LOCATIONS

THE COLUMN TO TH

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MODEL STATION

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EUNE CINE

SECTION THROUGH STATION 23.0

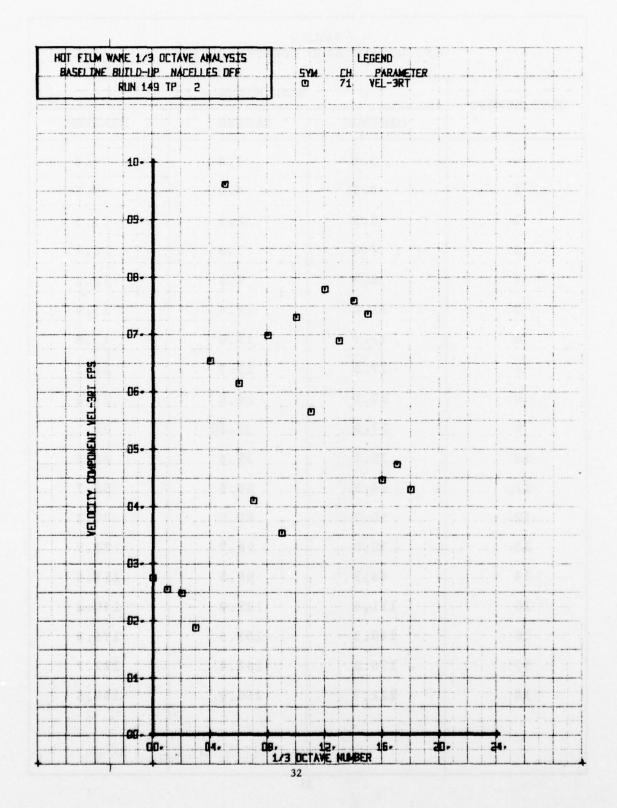
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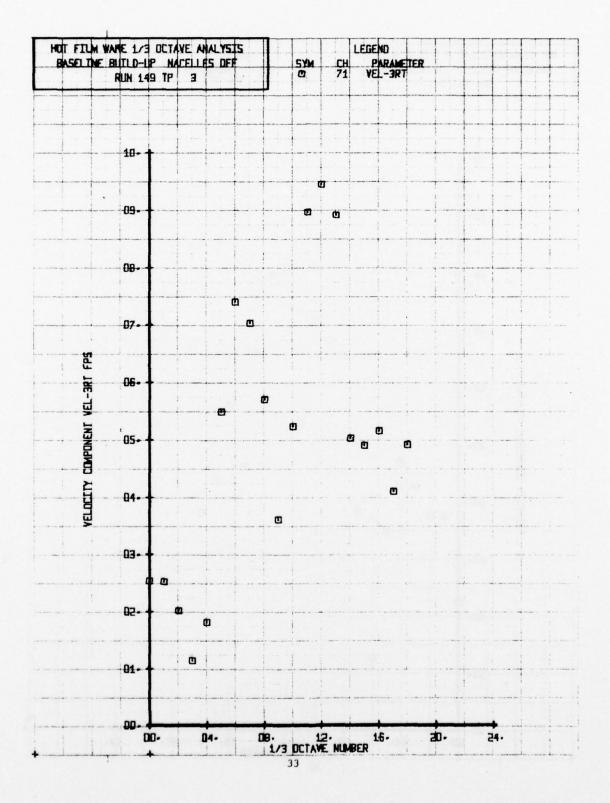
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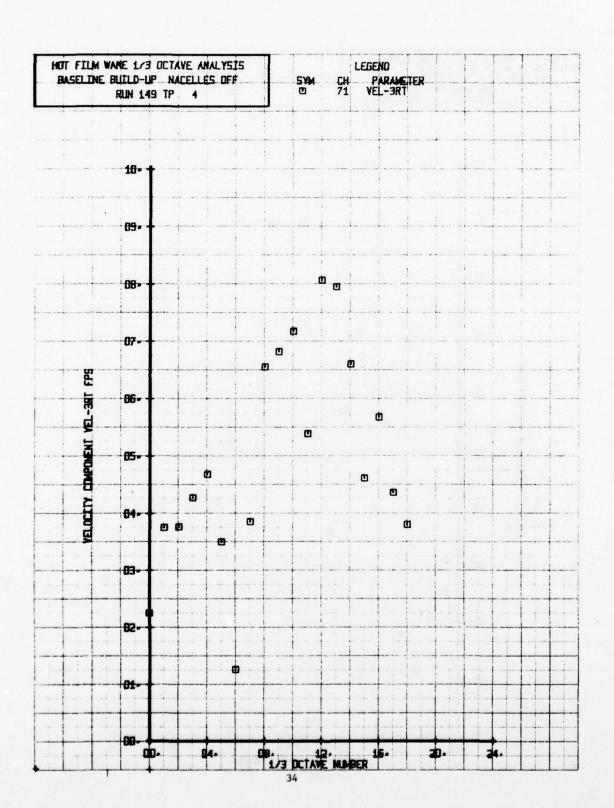
WATER

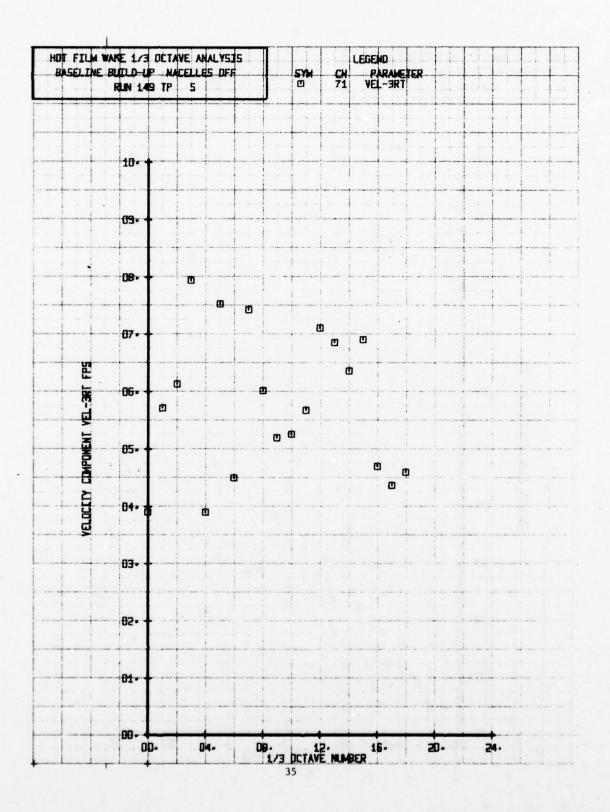
TABLE 4
1/3 OCTAVE BAND IDENTIFICATION

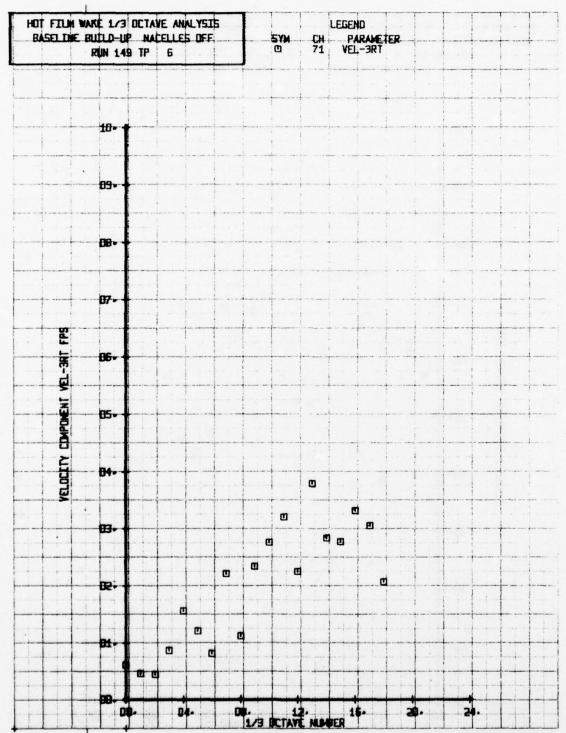
AND NUMBER	BAND WIDTH - Hz						
ZAVD NOIDDA	MINIMUM	CENTER	MAXIMUM				
0	3.5	3.4	4.4				
1	4.4	4.9	5.5				
2	5.5	6.2	7.0				
3	7.0	7.8	8.7				
4	8.7	9,8	11.0				
5	11.0	12,4	13.9				
6	13.4	15,6	17.5				
7	17.5	19.7	22,1				
8	22.1	24.8	27.8				
9	27.8	31.25	35,1				
10	35.1	39.4	44.2				
11	44.2	49.6	55.7				
12	55.7	62.5	70.2				
13	70.2	78.7	88.9				
14	88.9	99.2	111.4				
15	111.4	125.0	140.3				
16	140.3	157.5	176.8				
17	176.8	198.4	222.7				
18	222.7	250.0	280.6				

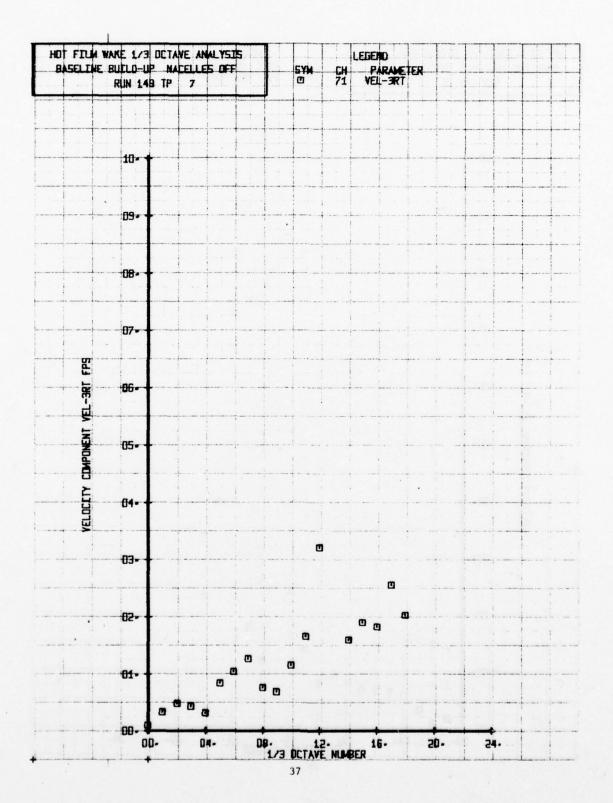


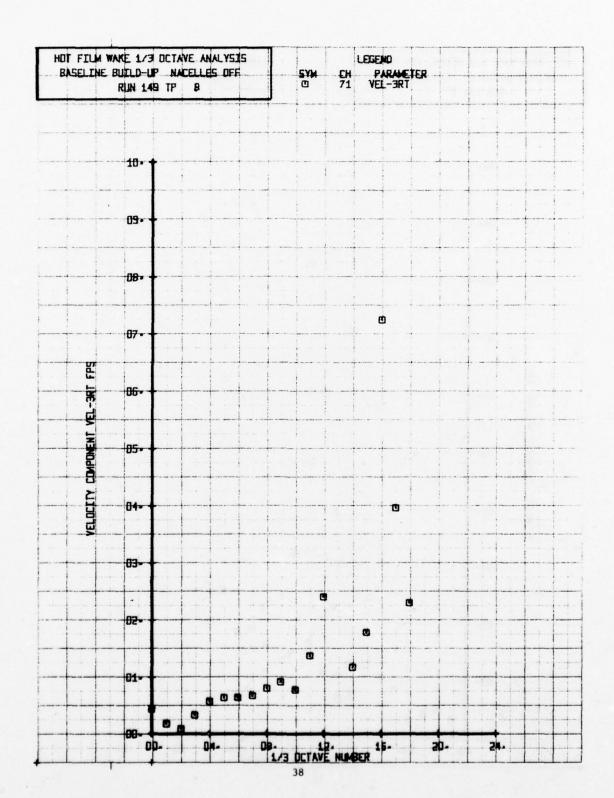


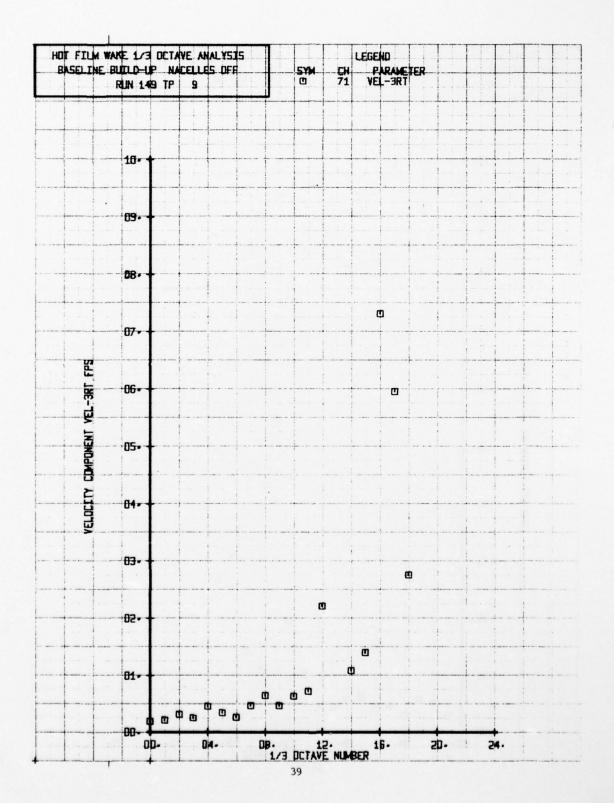


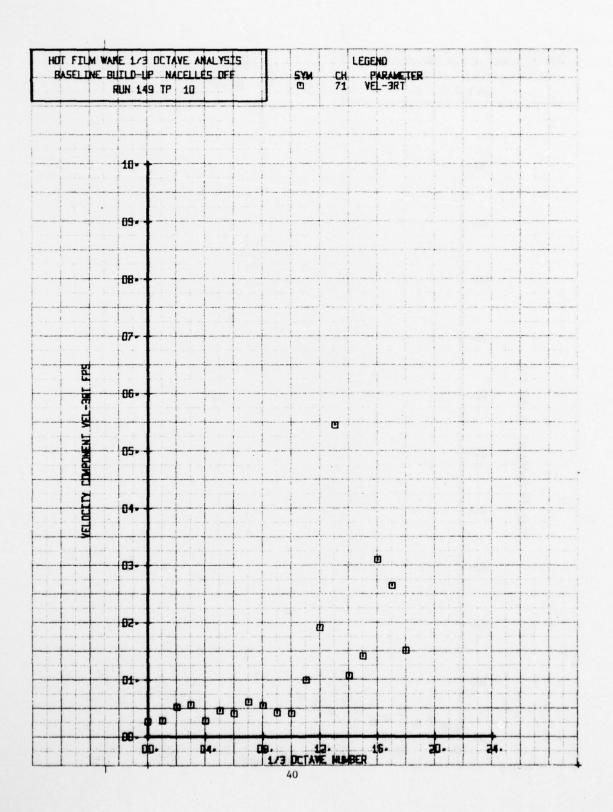


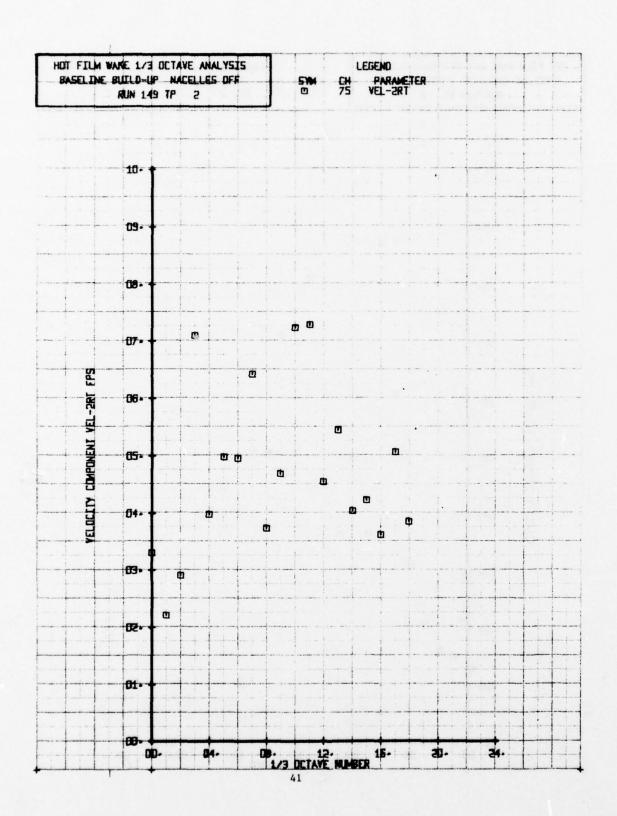


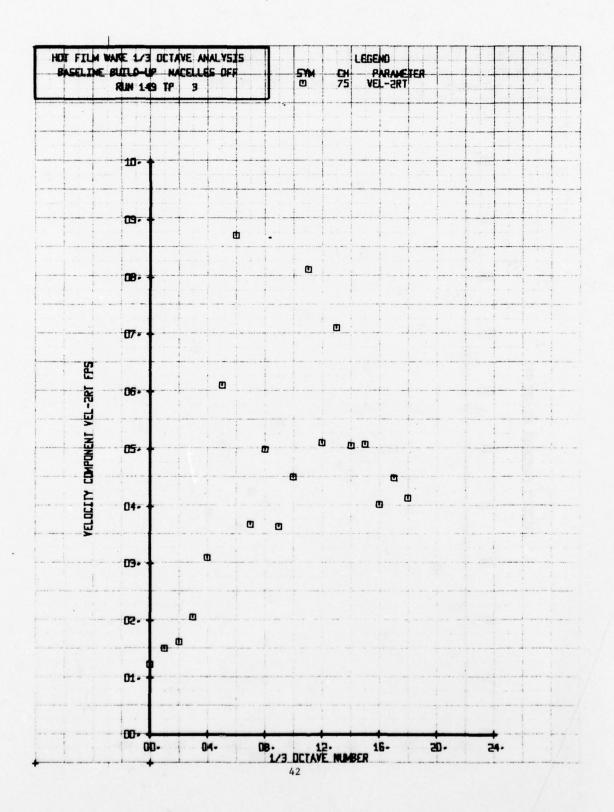


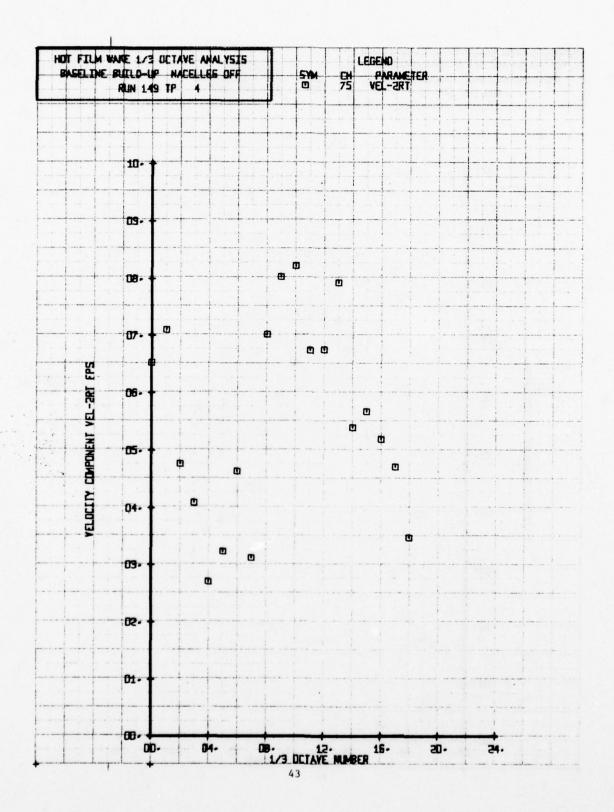


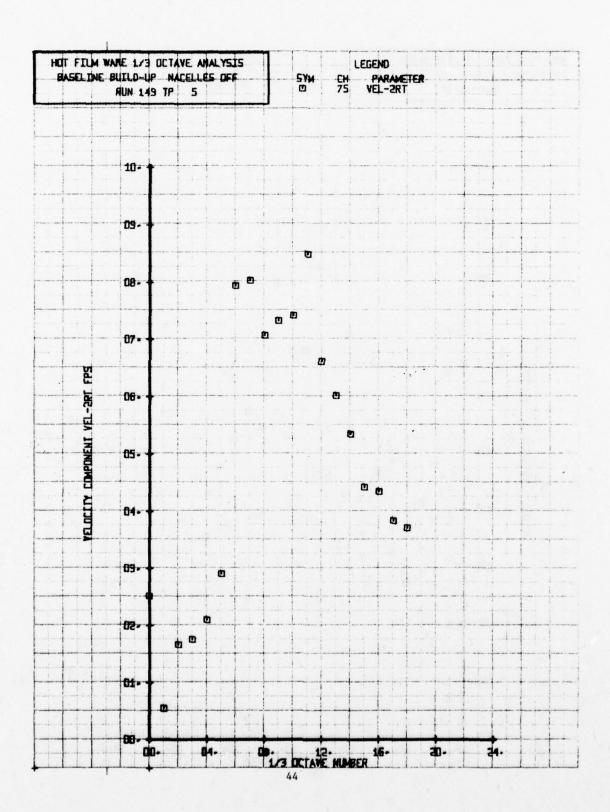


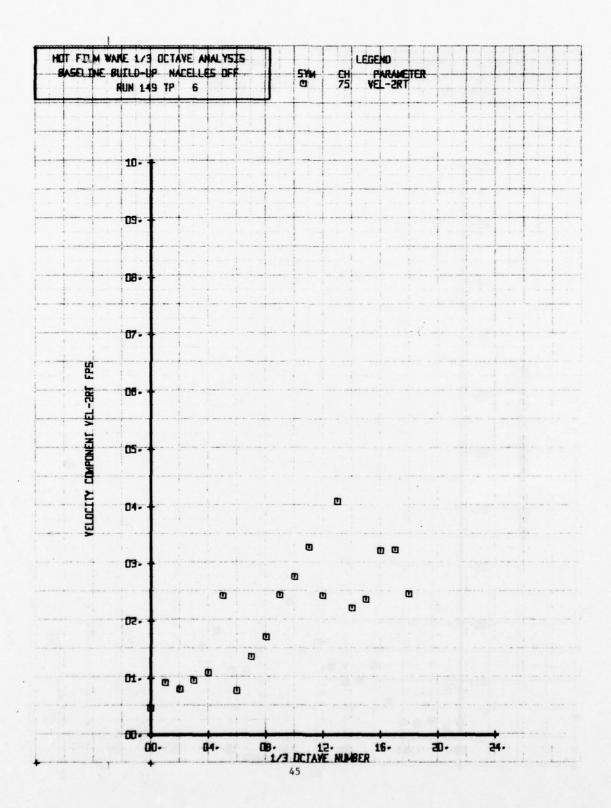


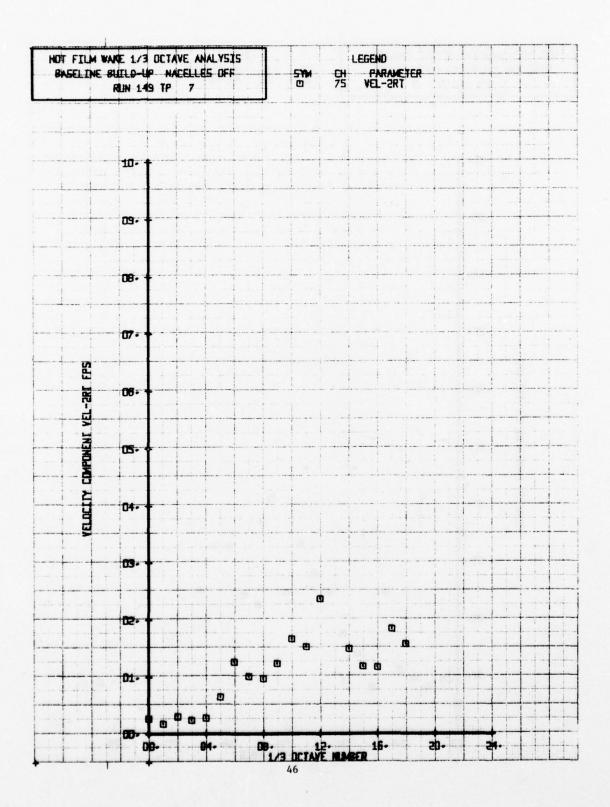


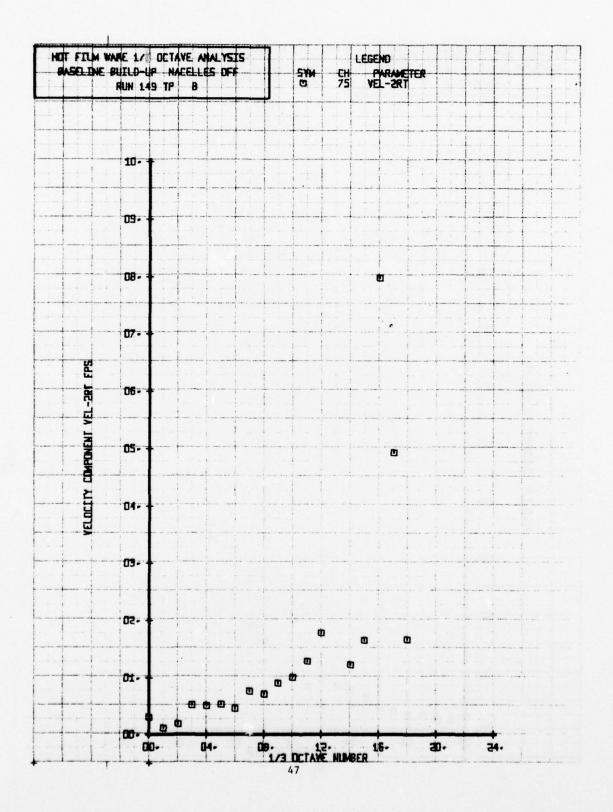


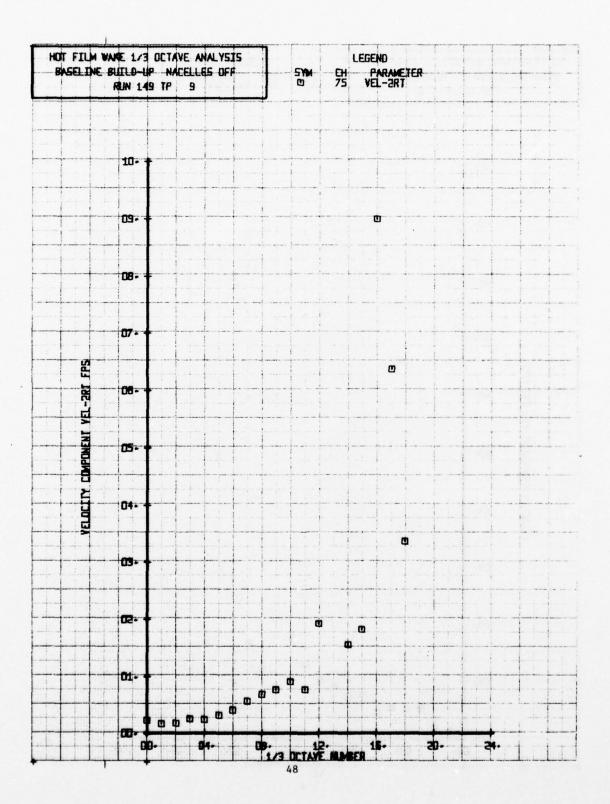


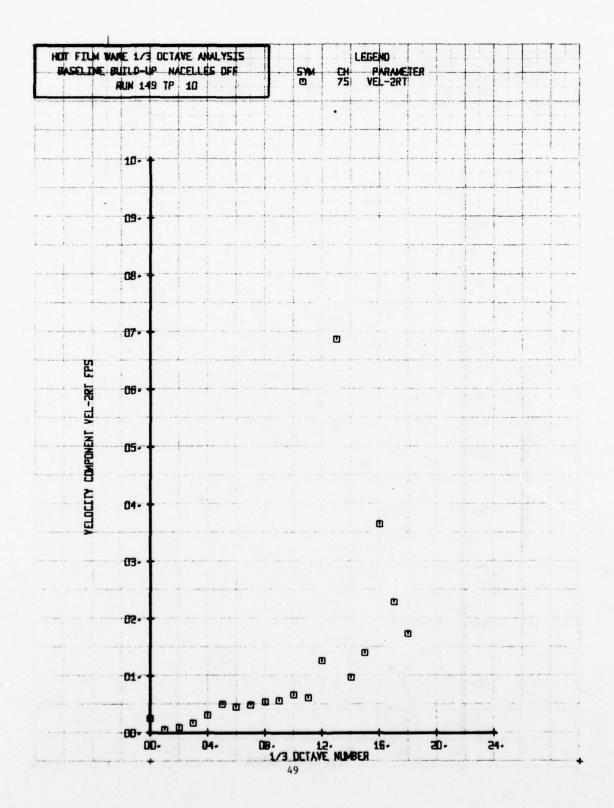


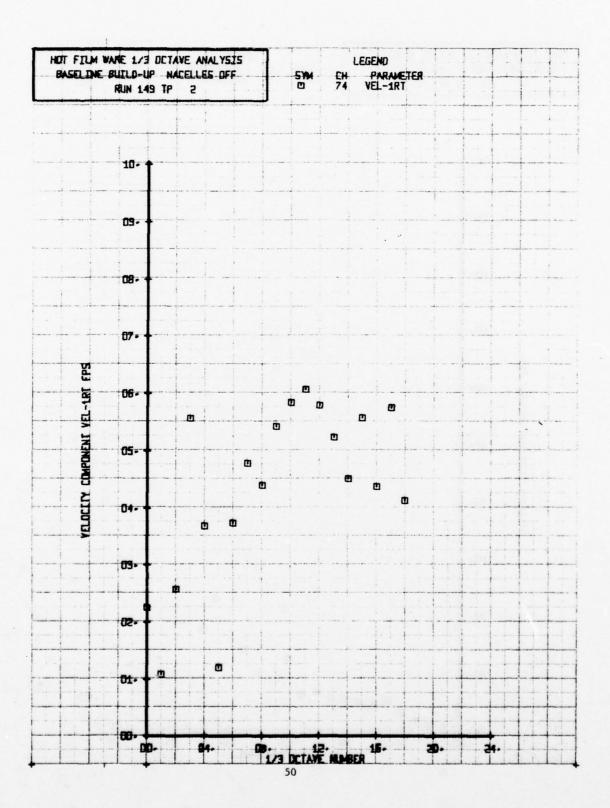


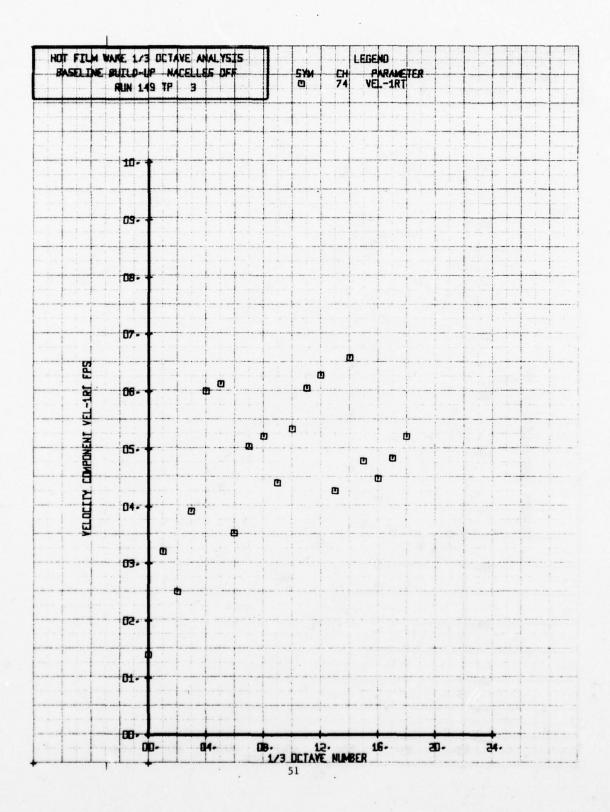


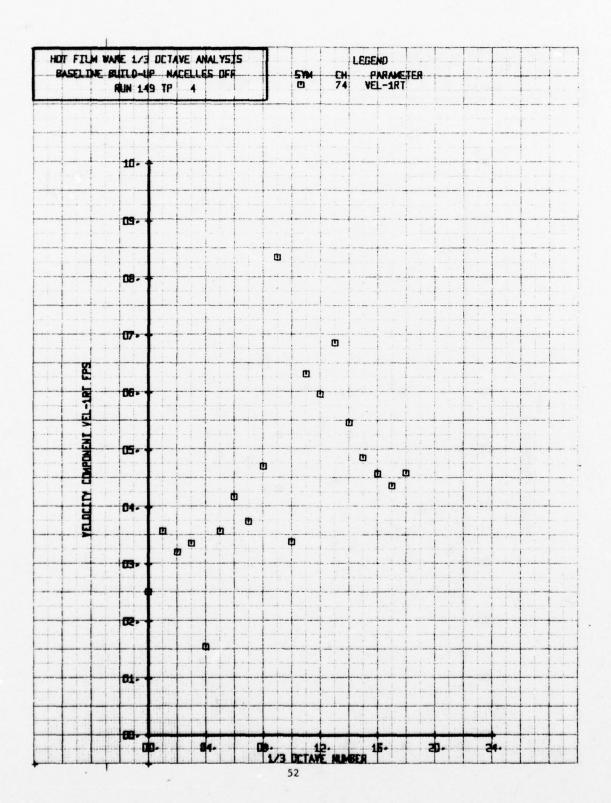


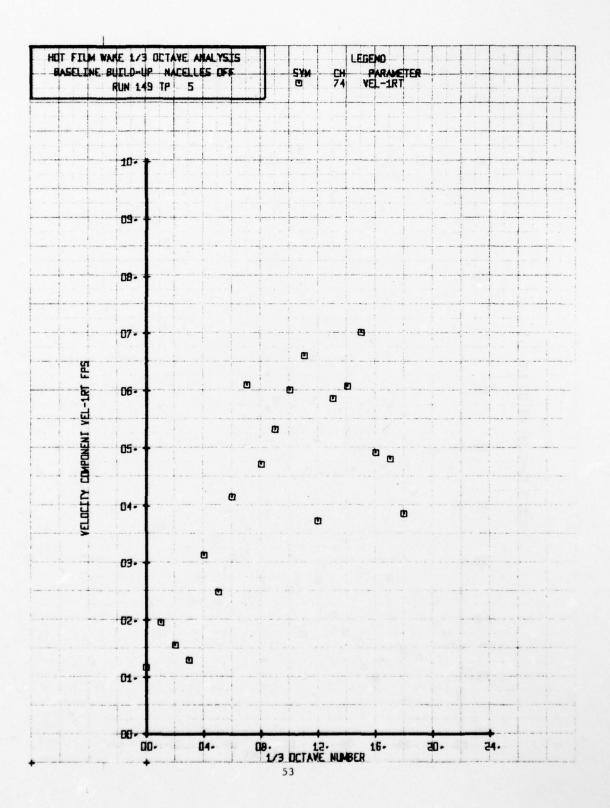


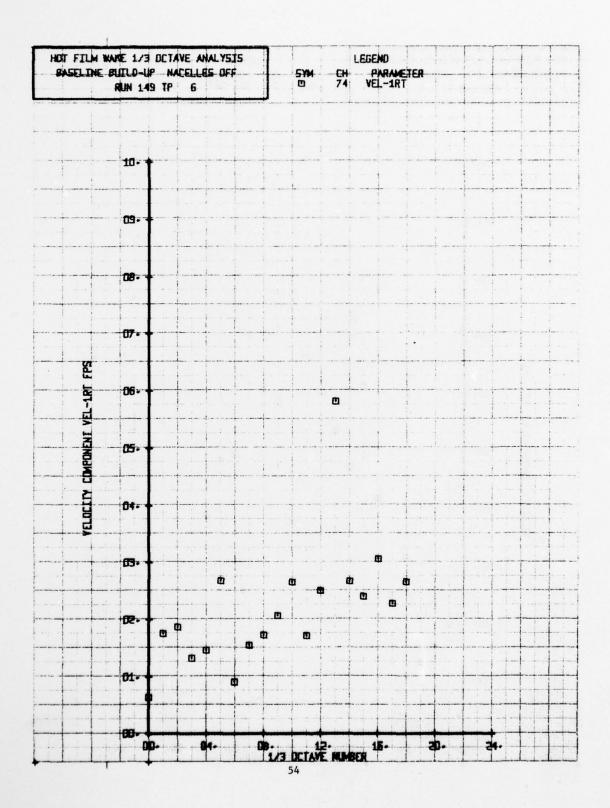


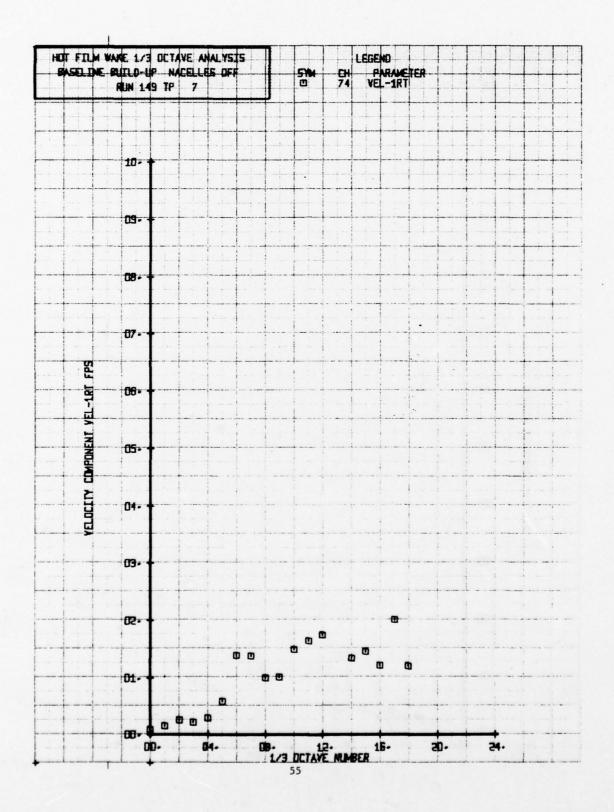


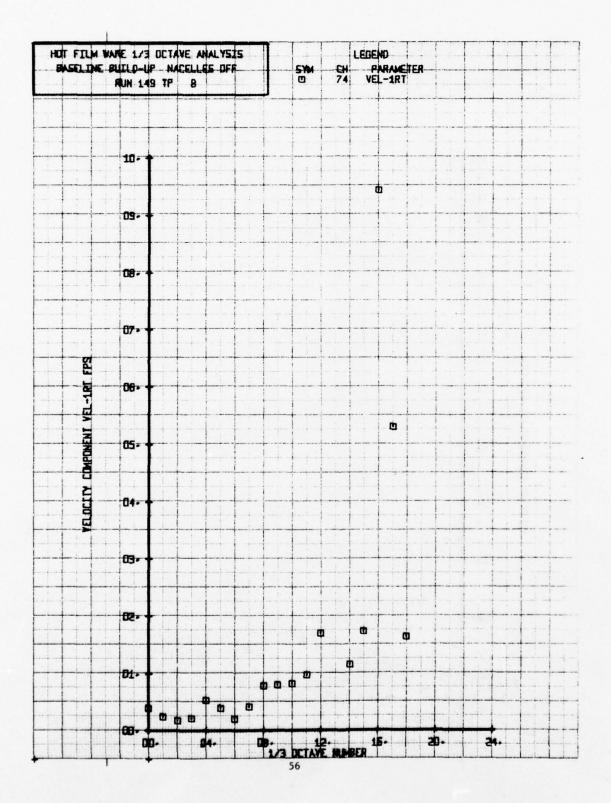


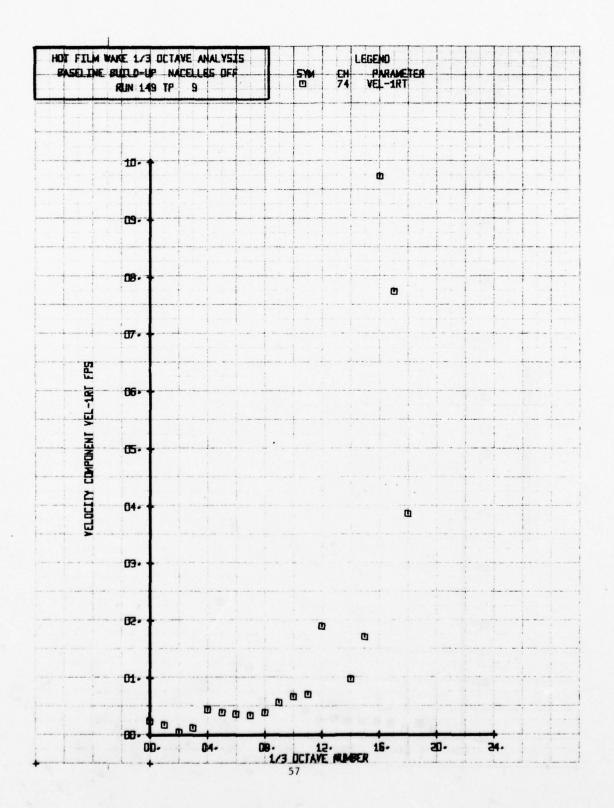


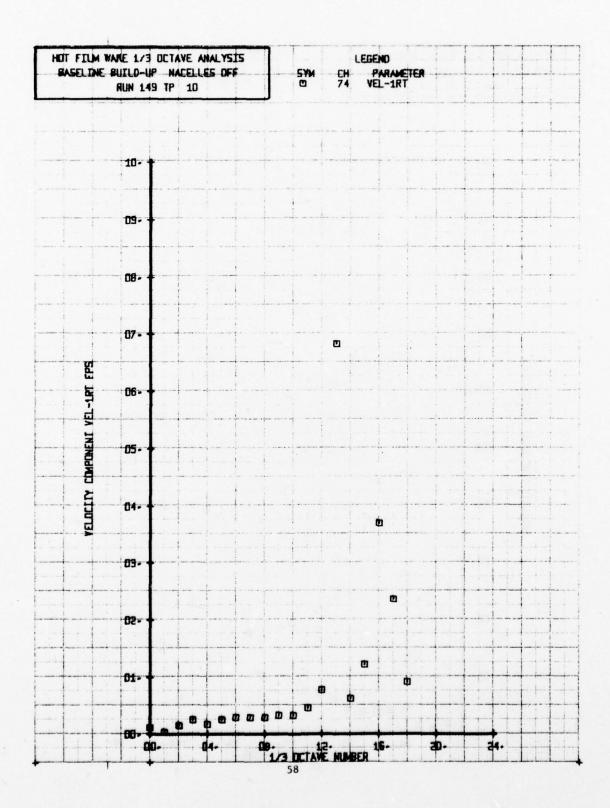


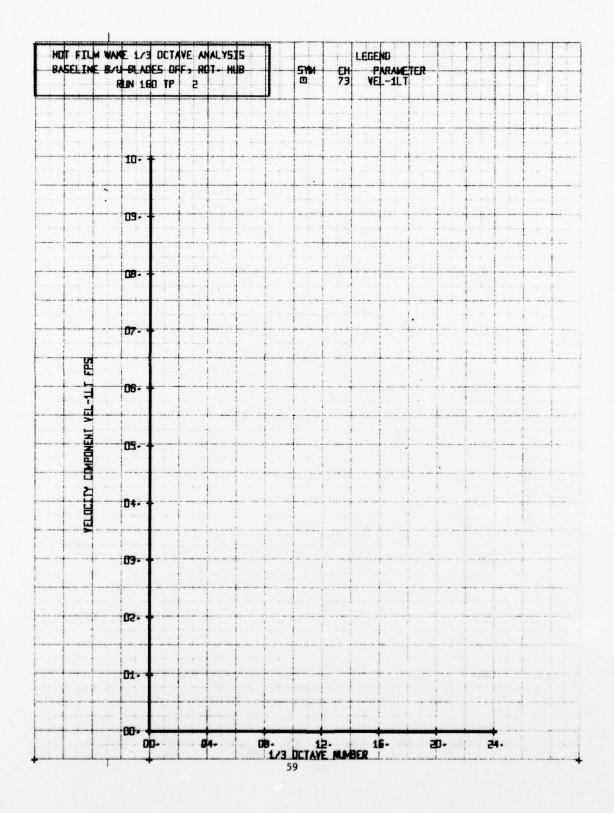


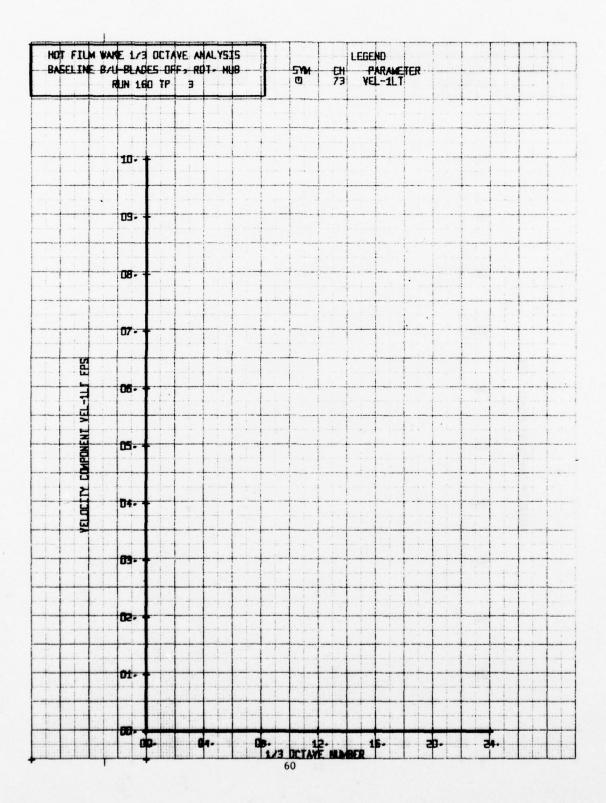


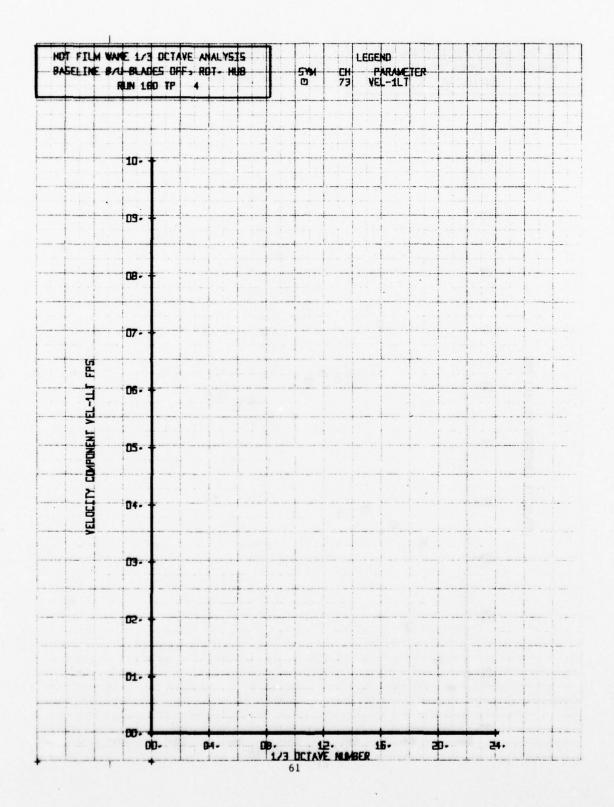


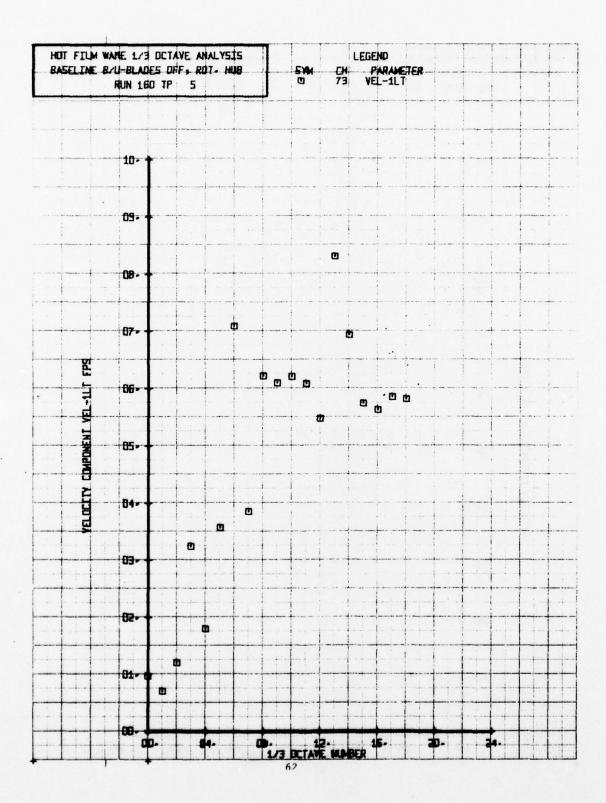


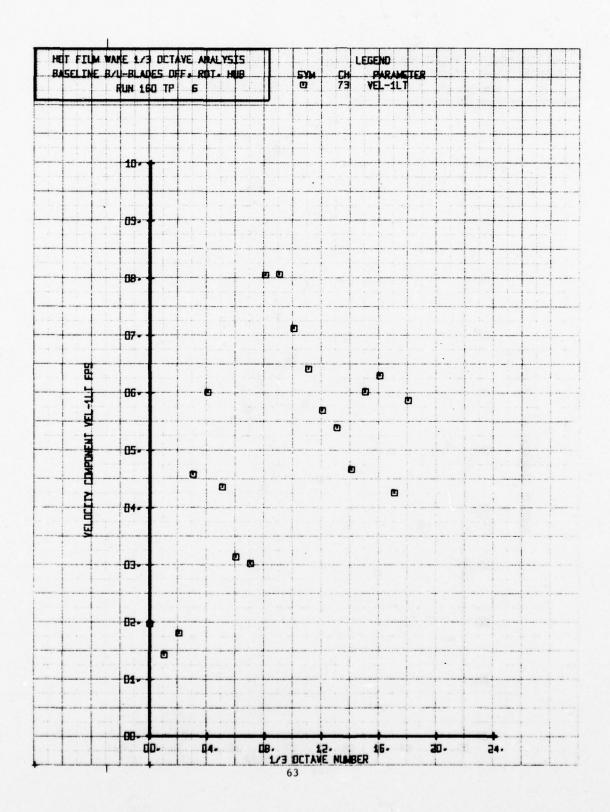


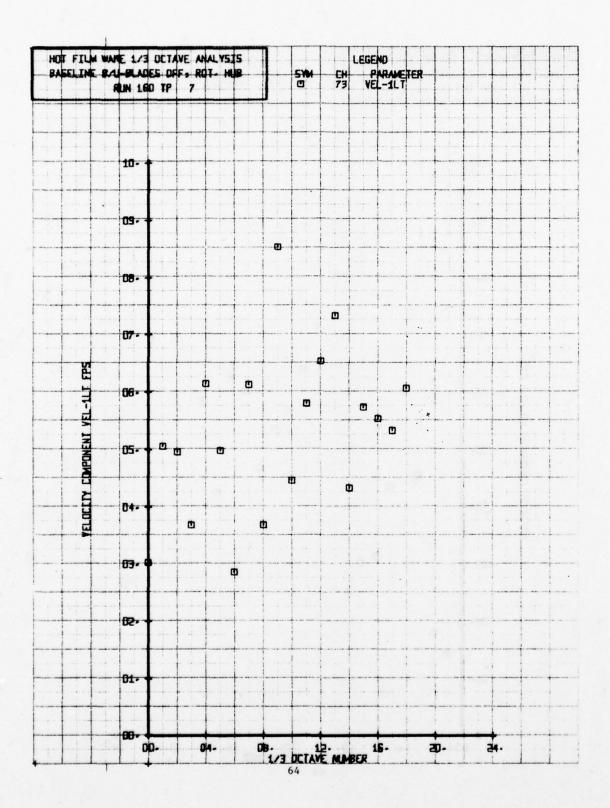


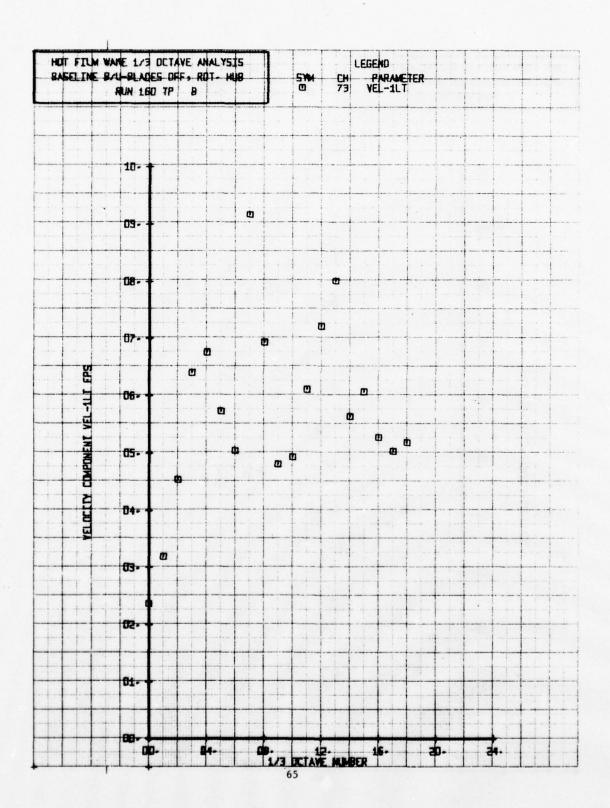


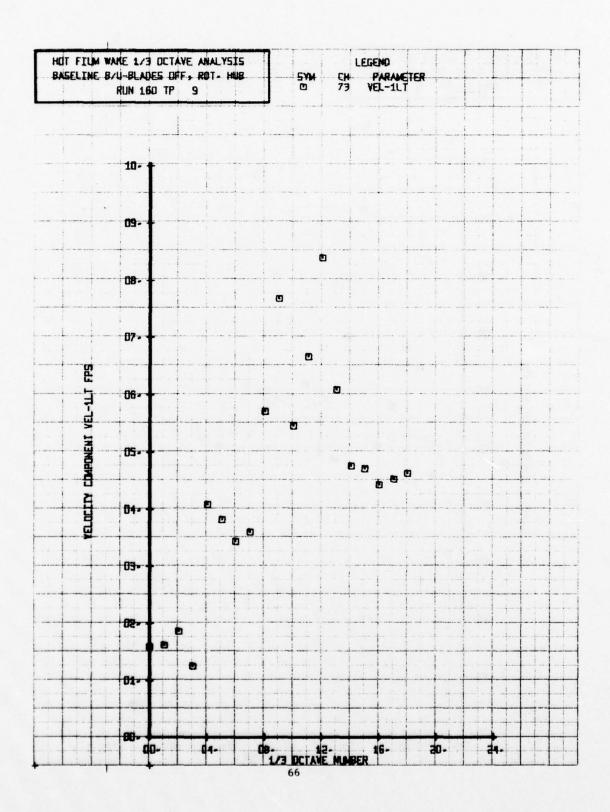


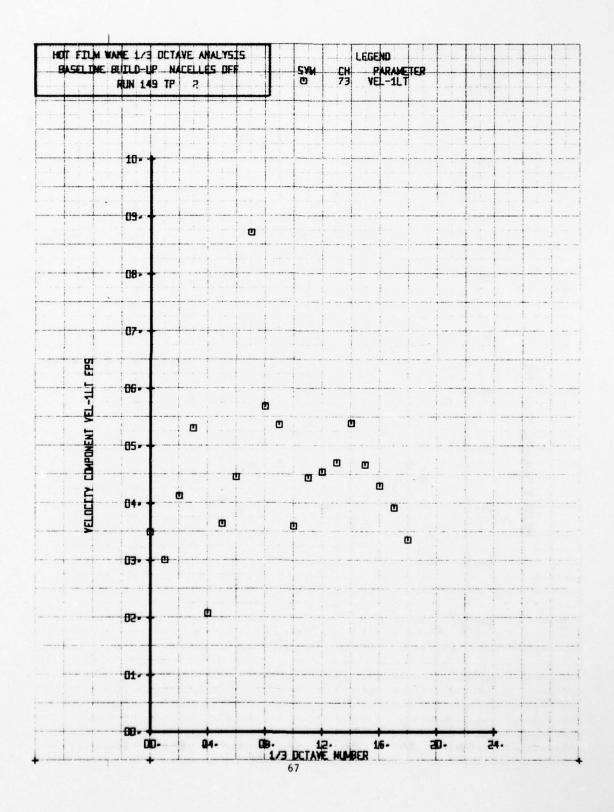


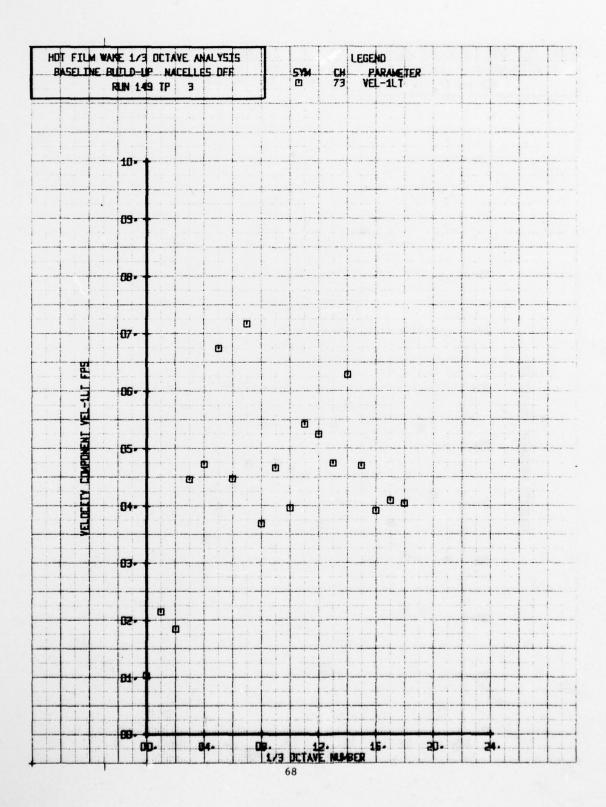


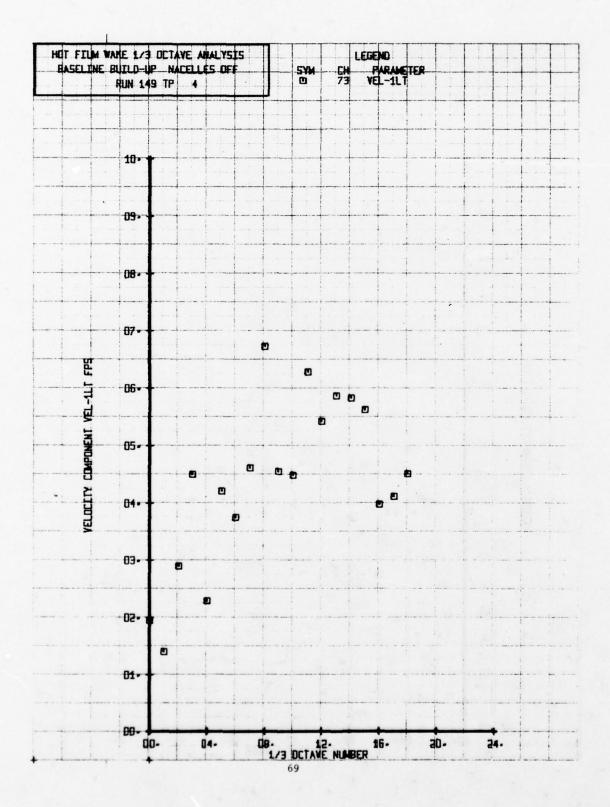


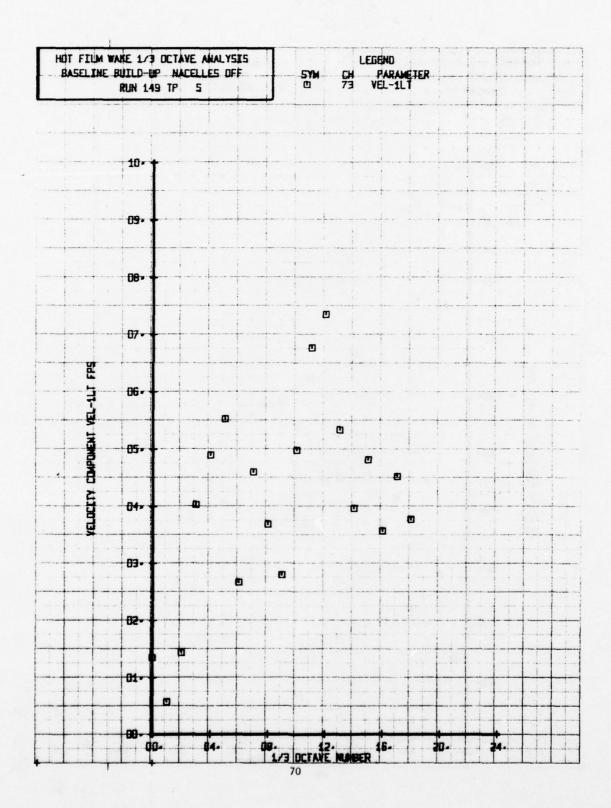


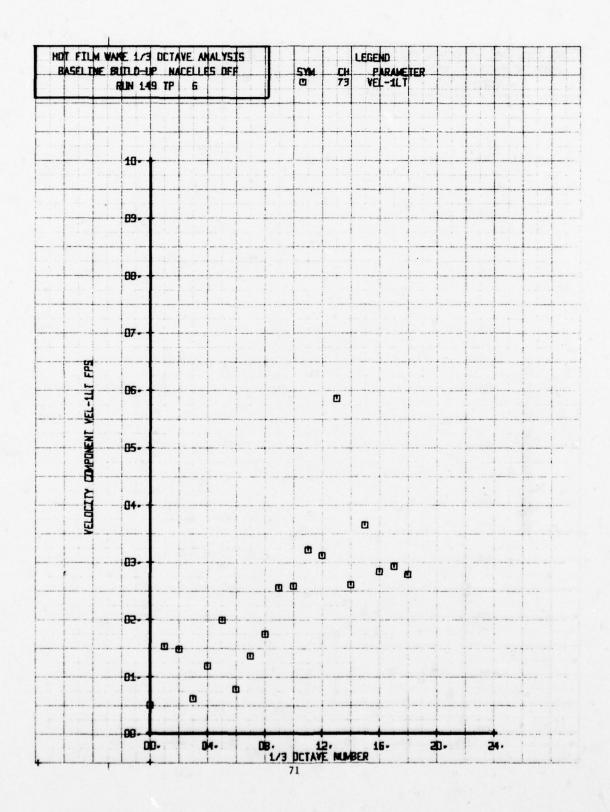


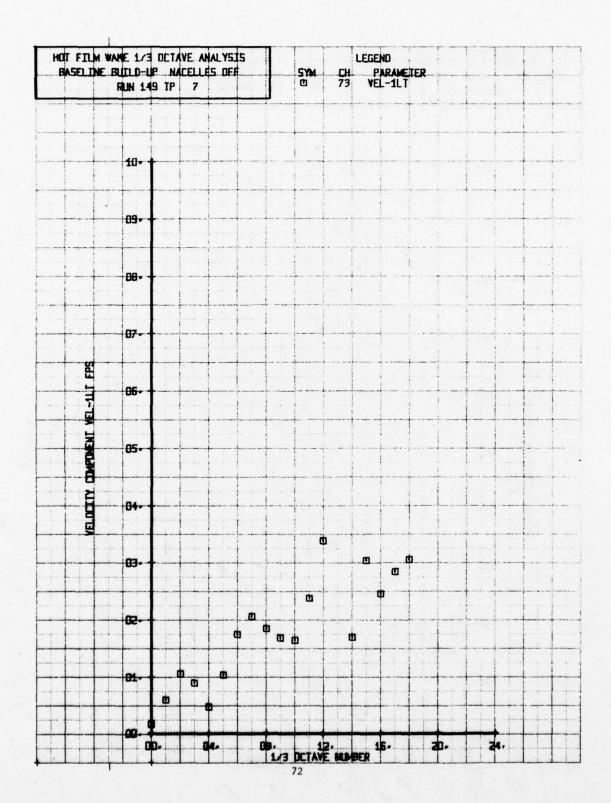


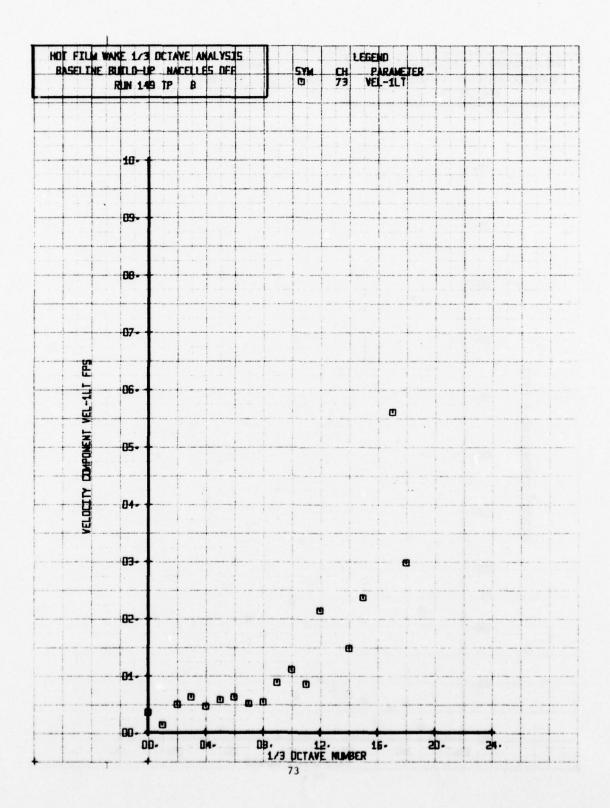


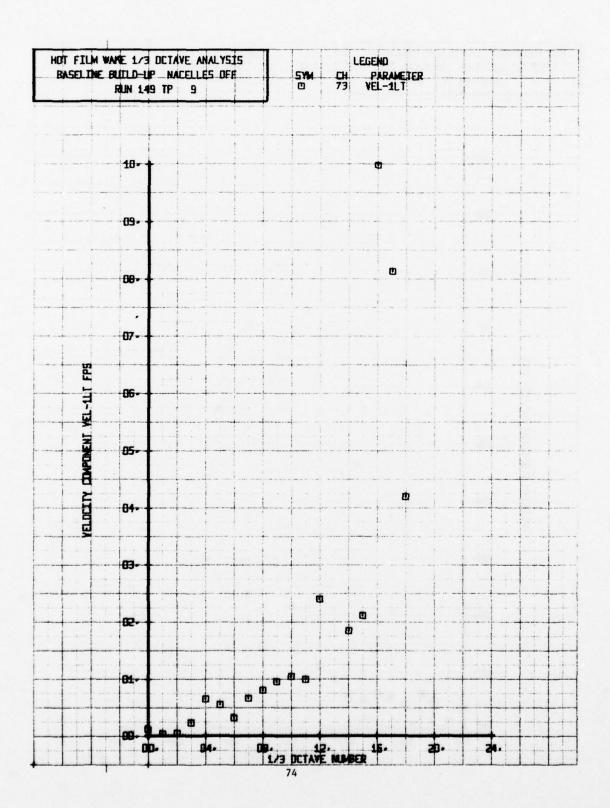


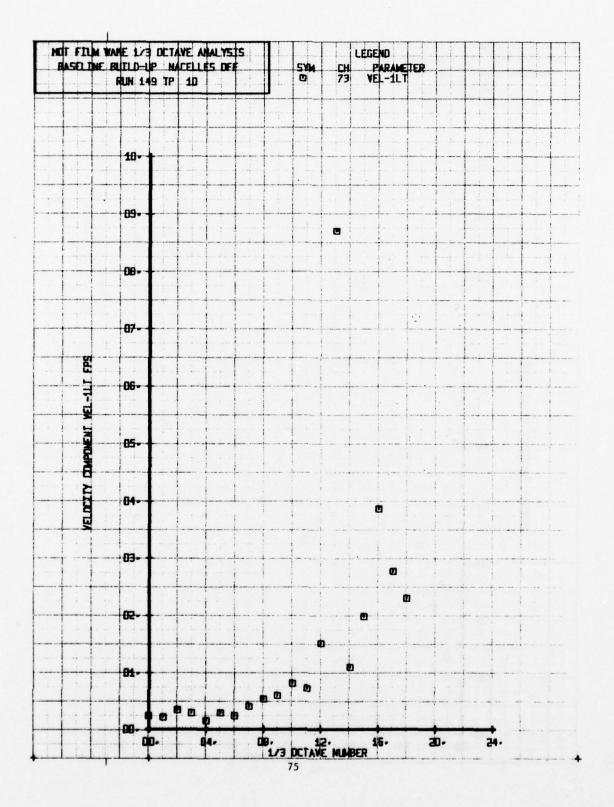


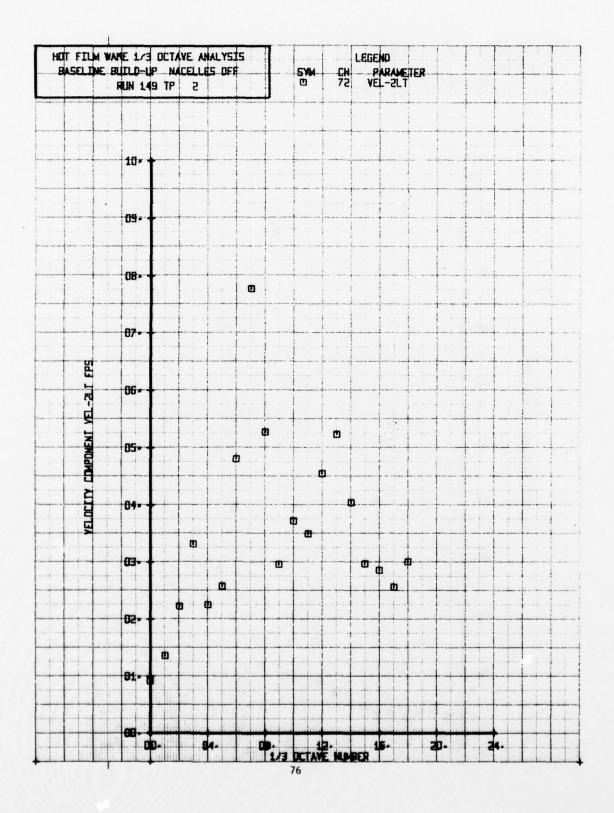


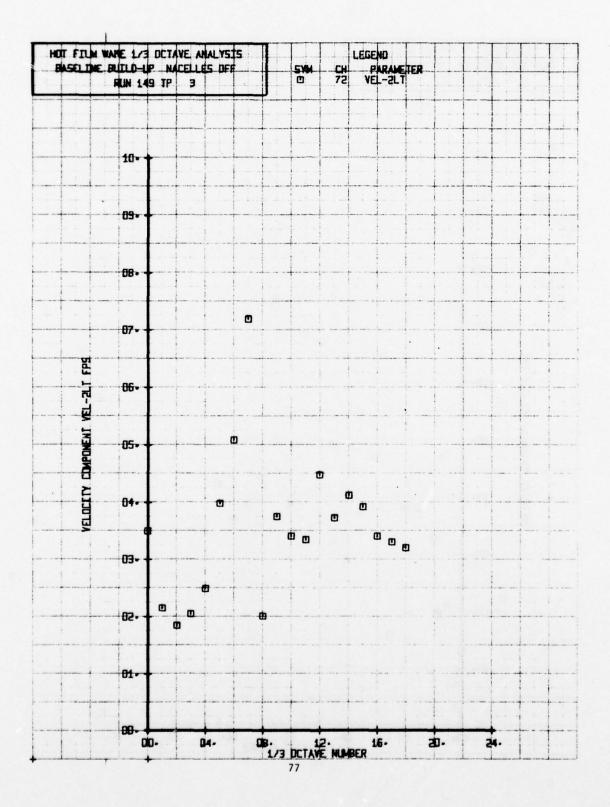


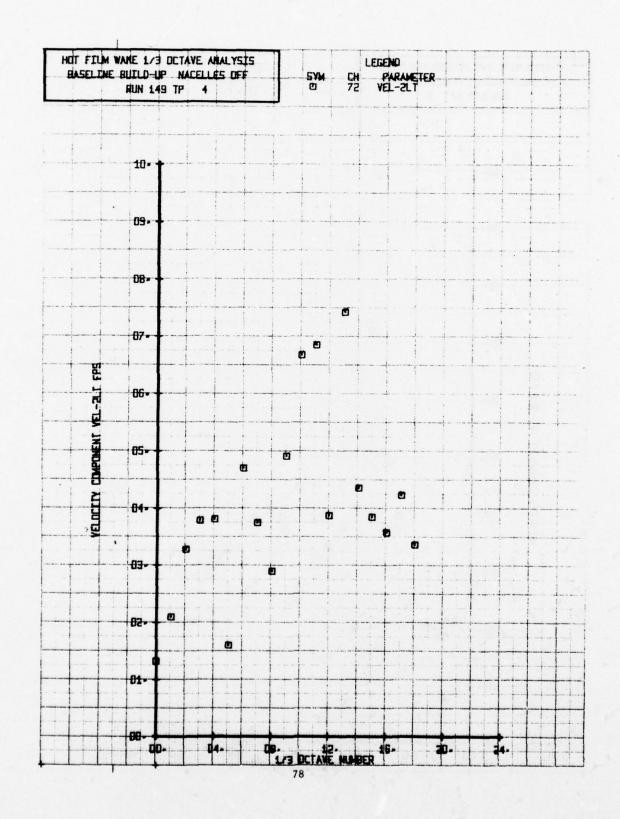


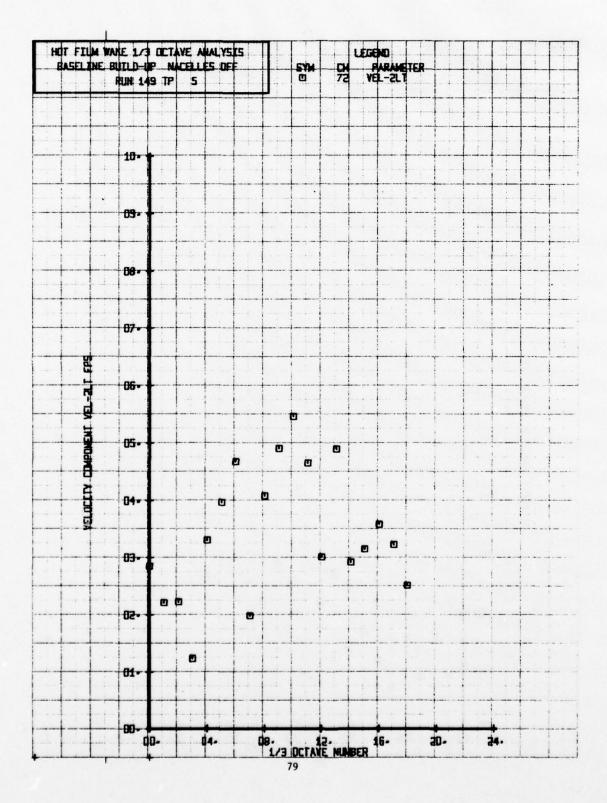


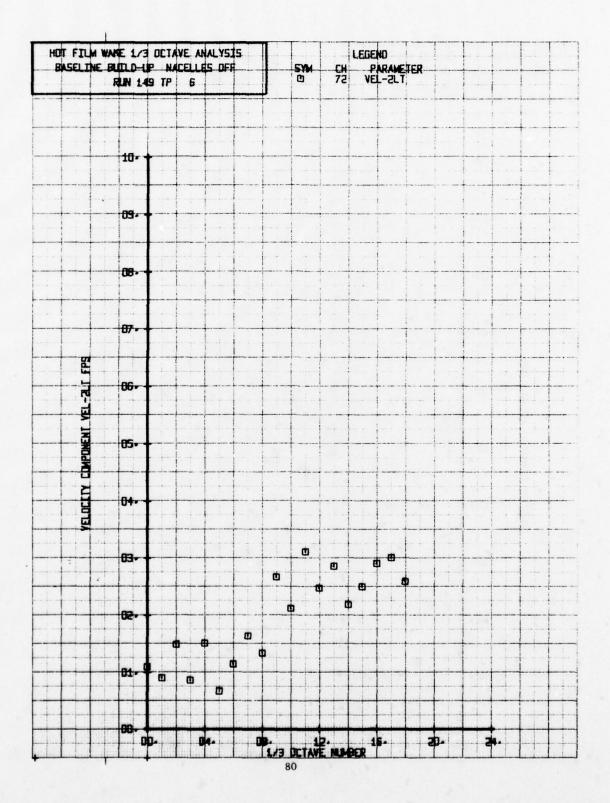


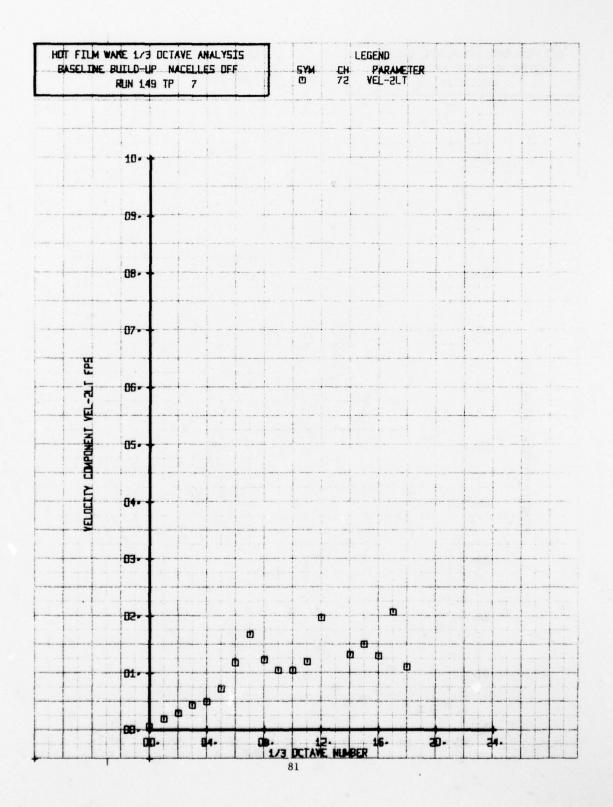


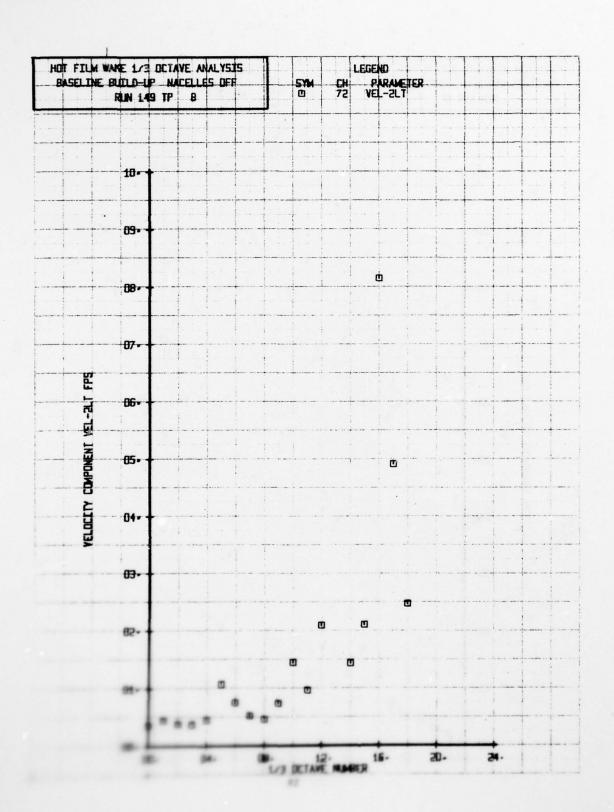


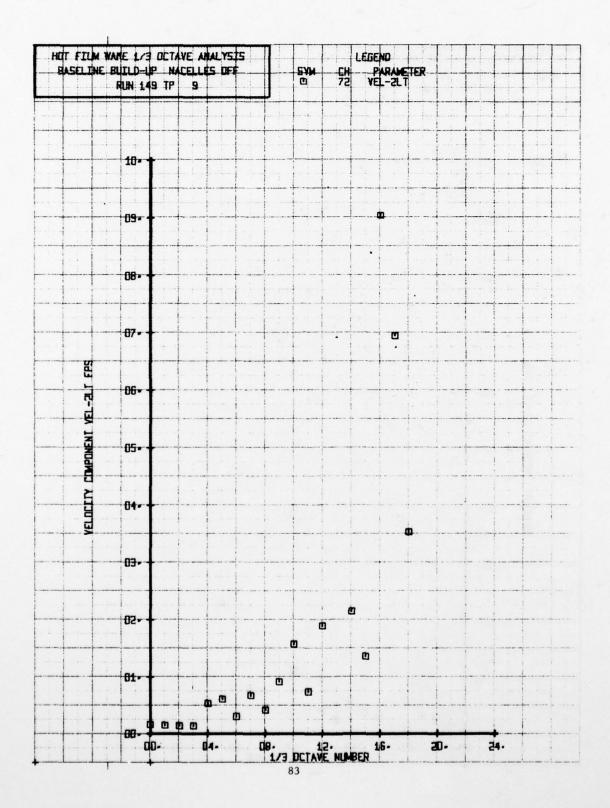


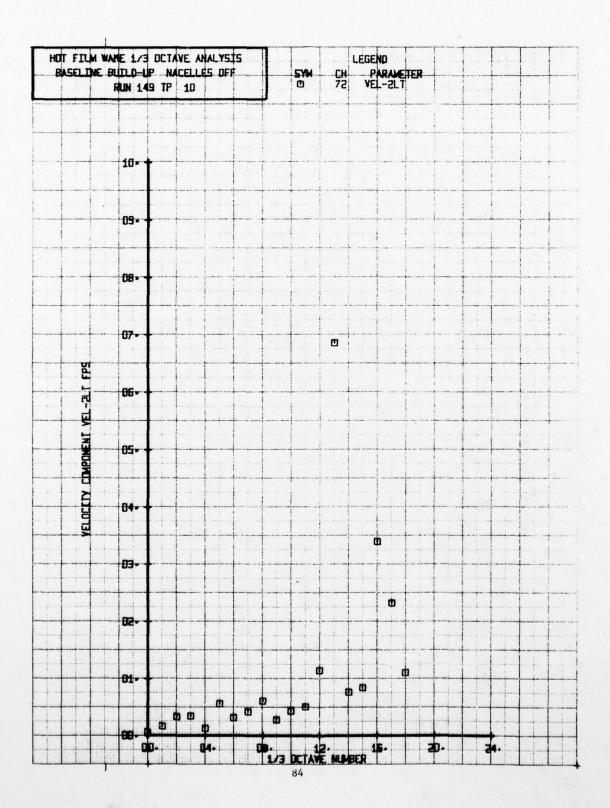


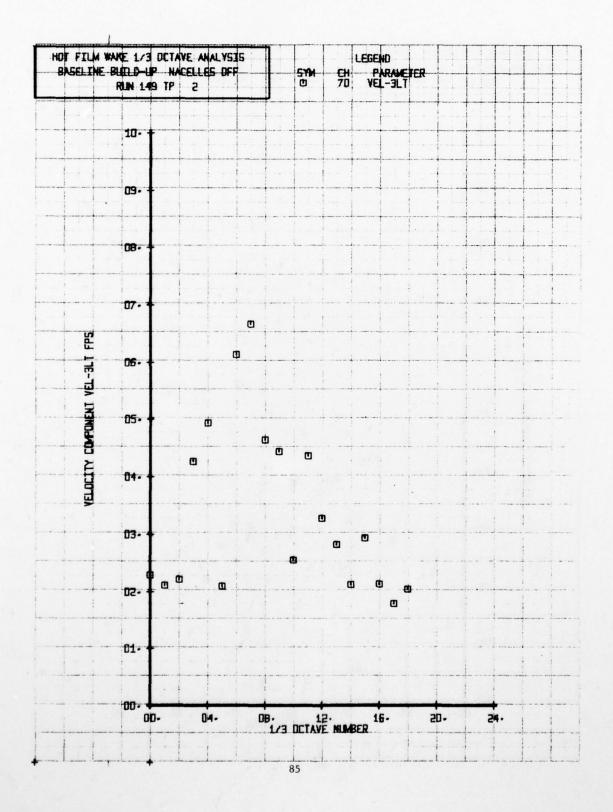


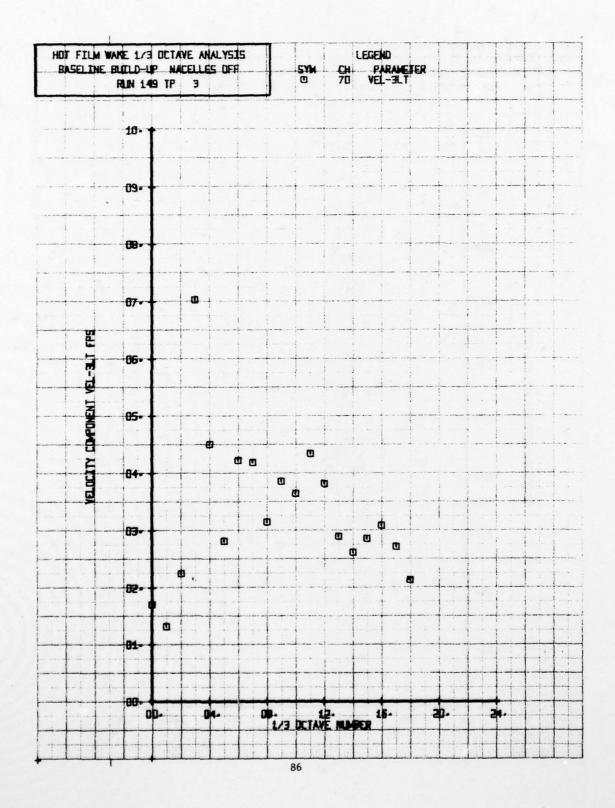


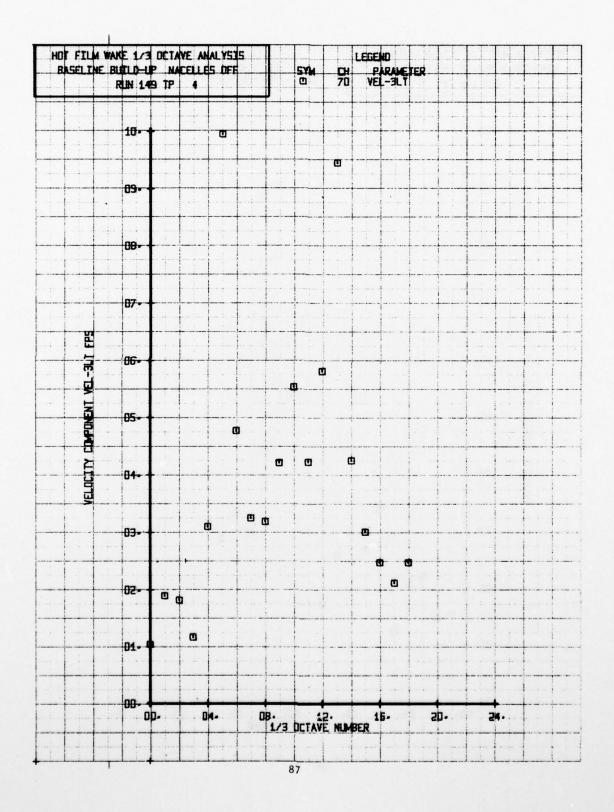


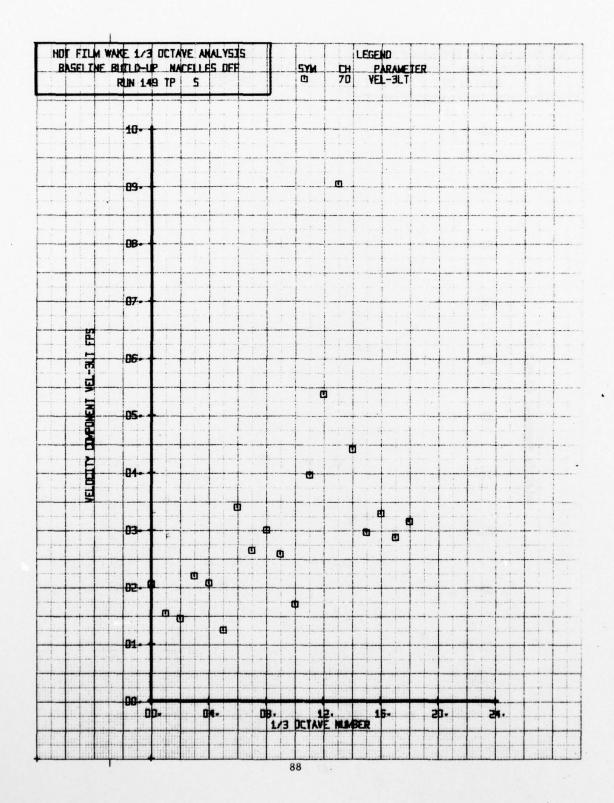


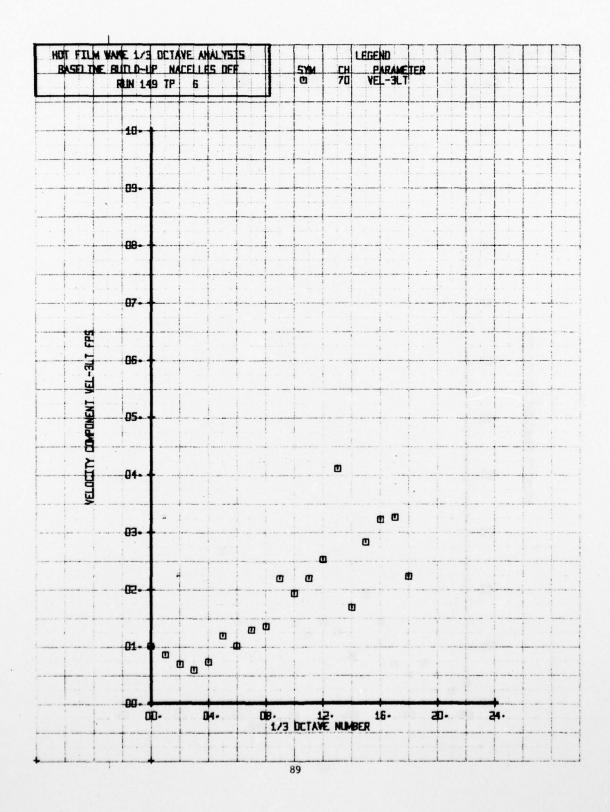


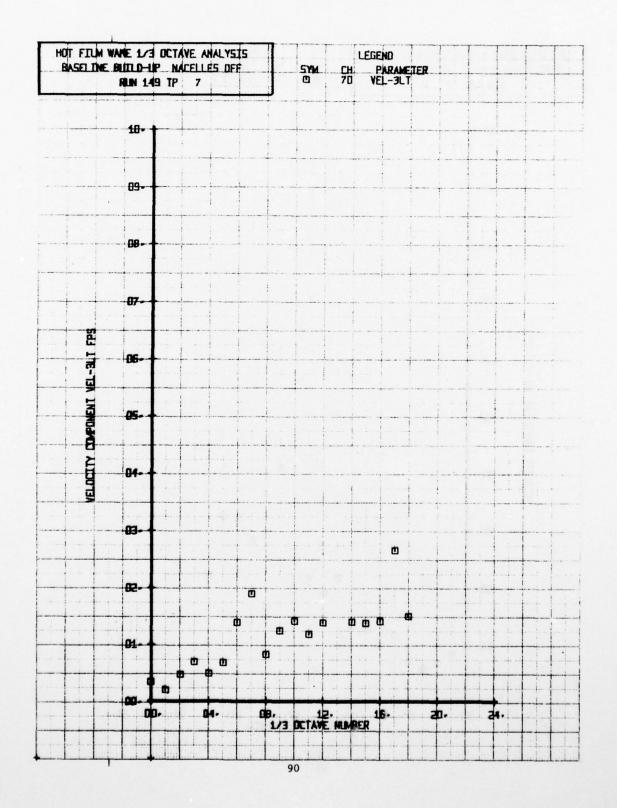


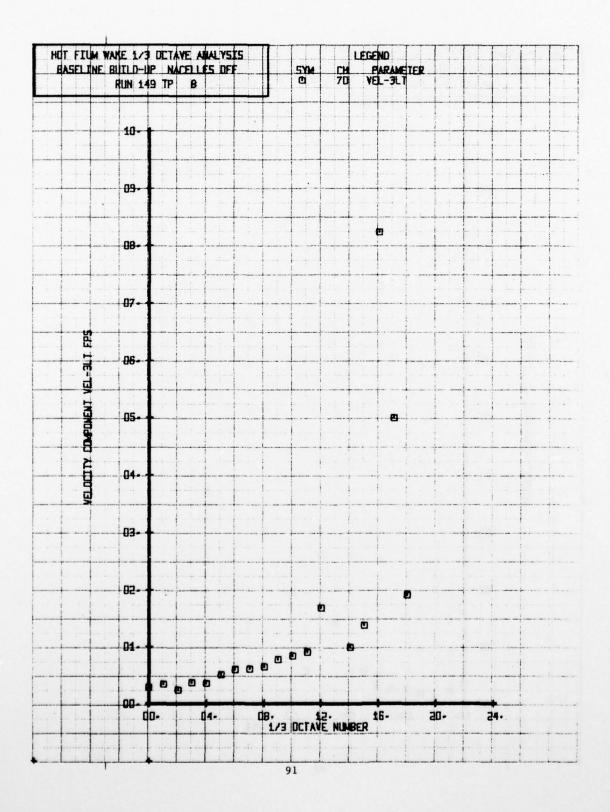


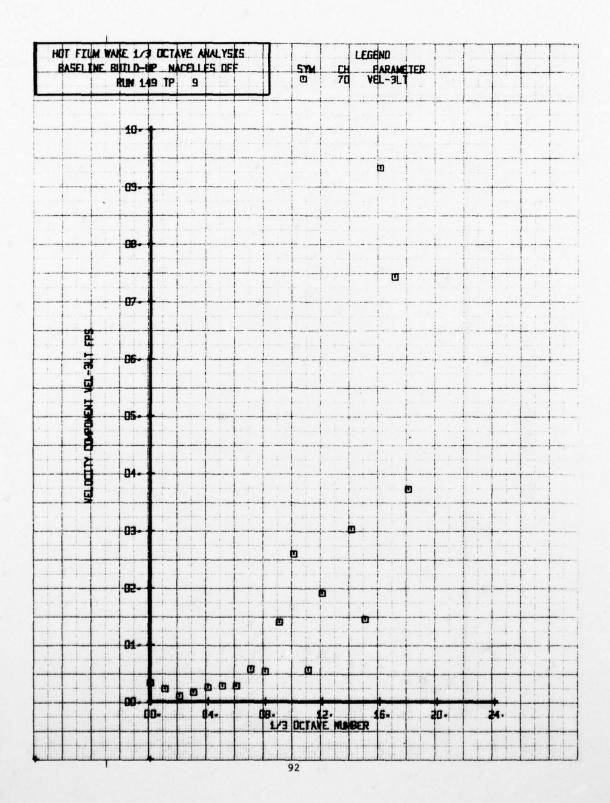


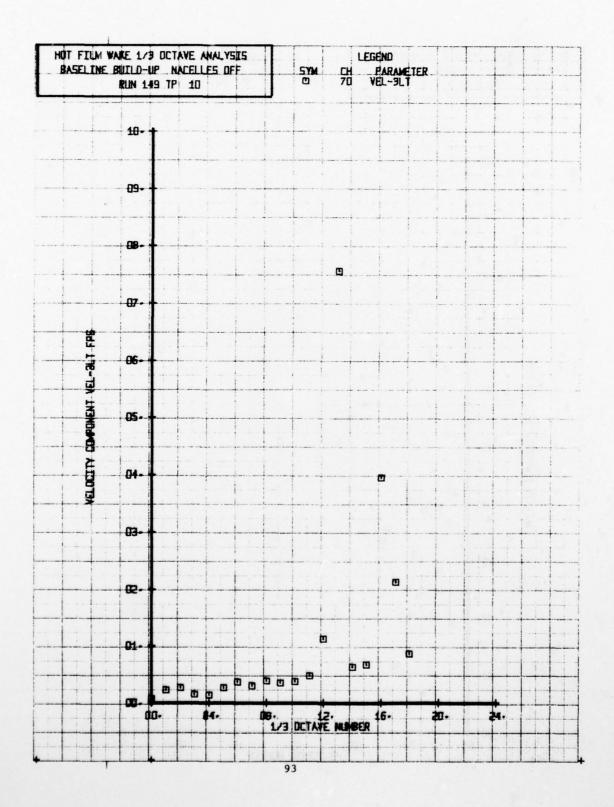


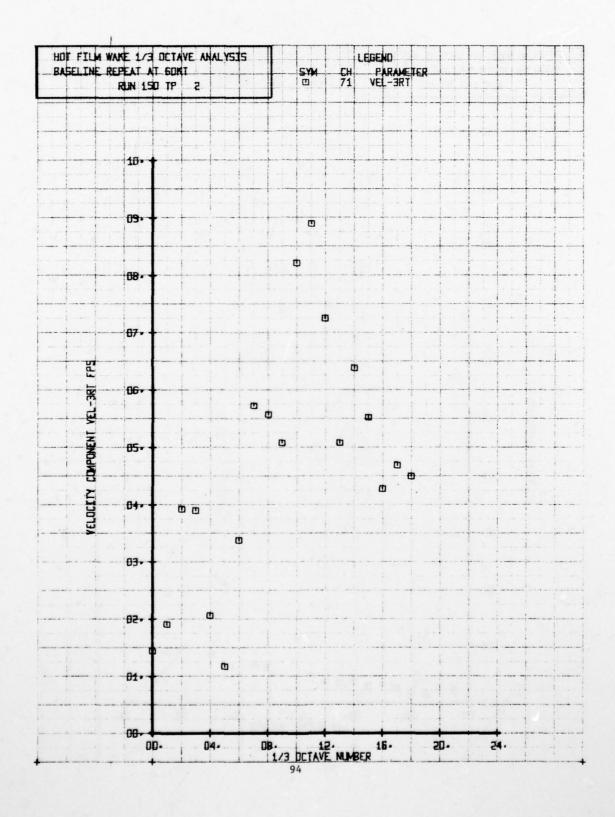


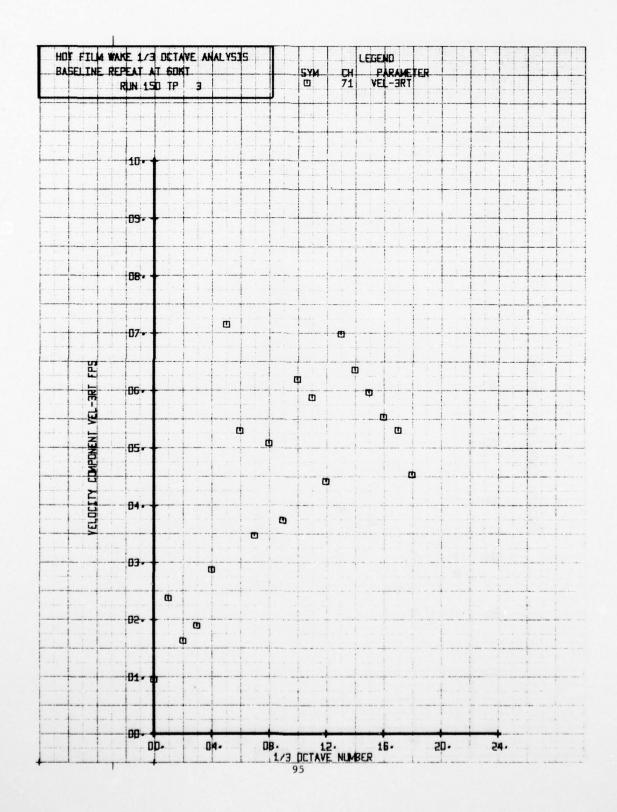




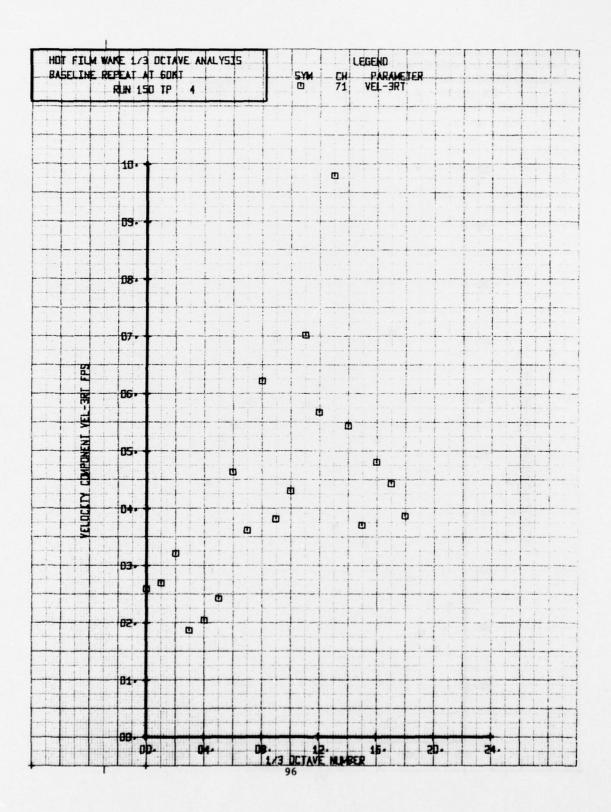


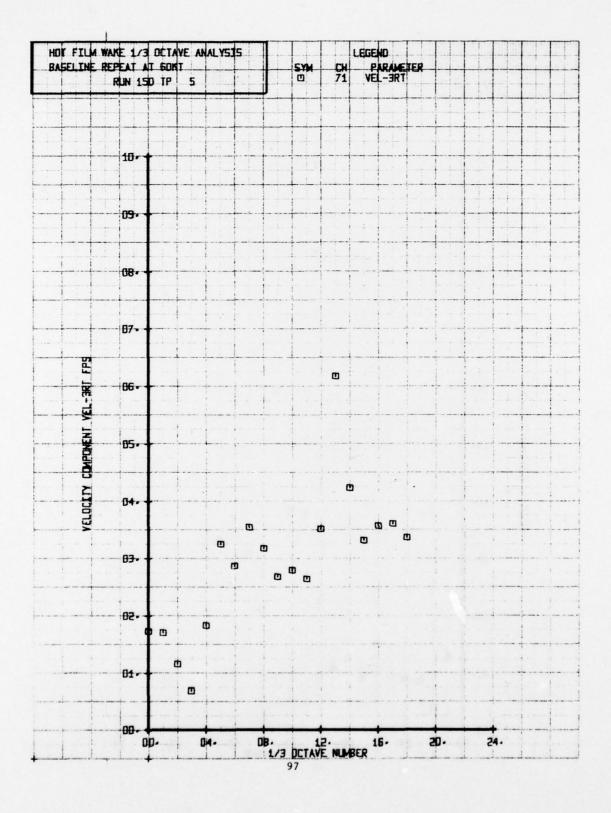


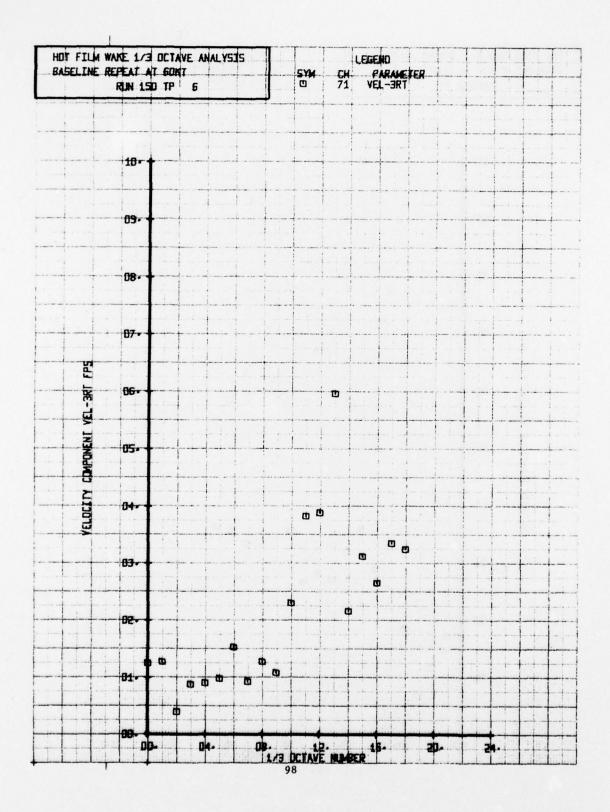


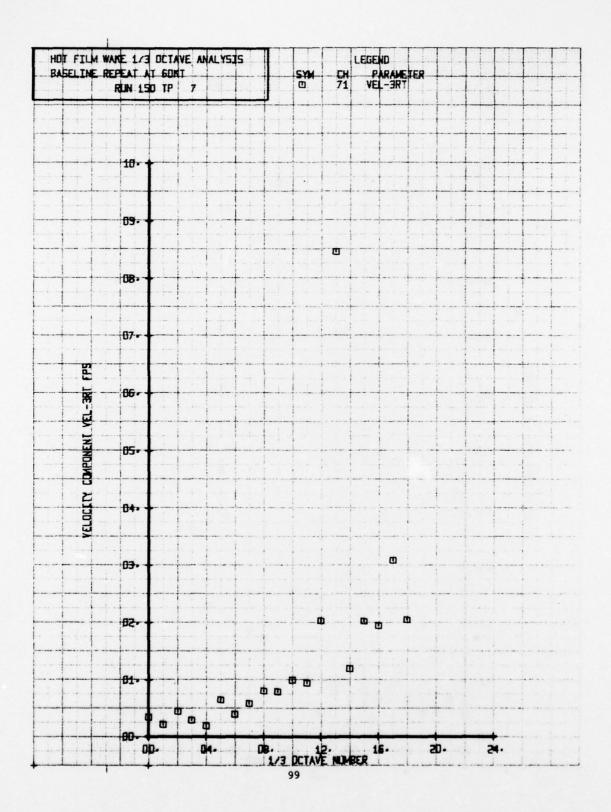


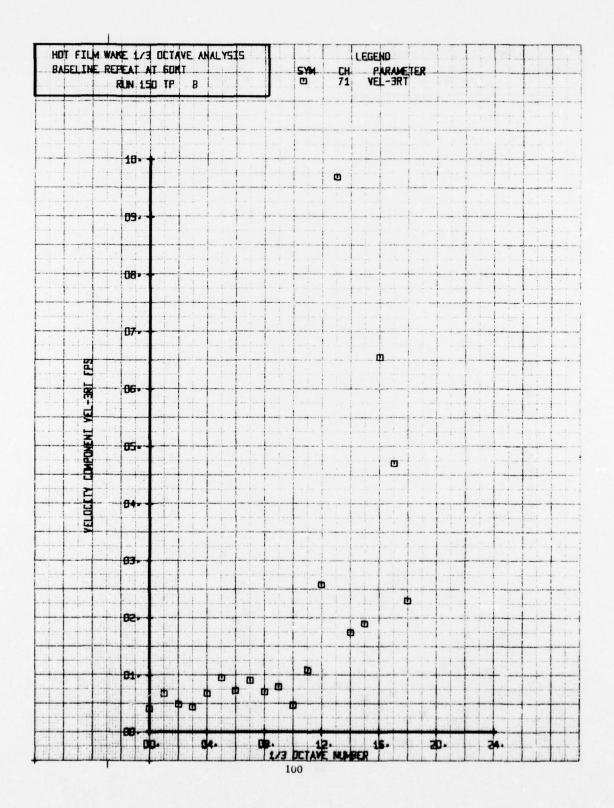
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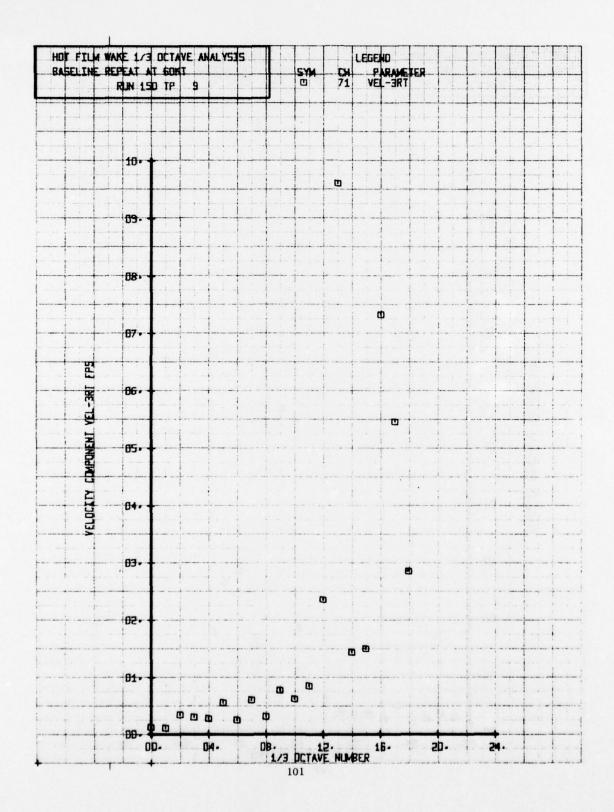


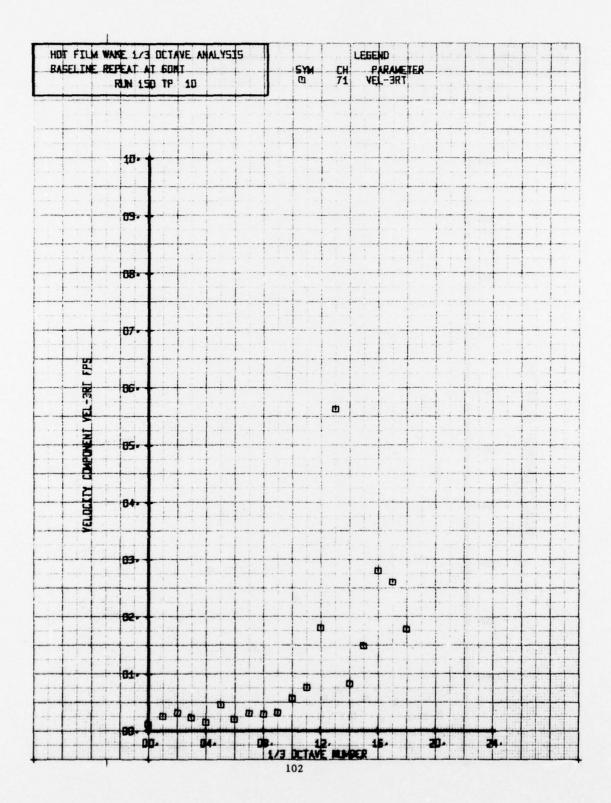


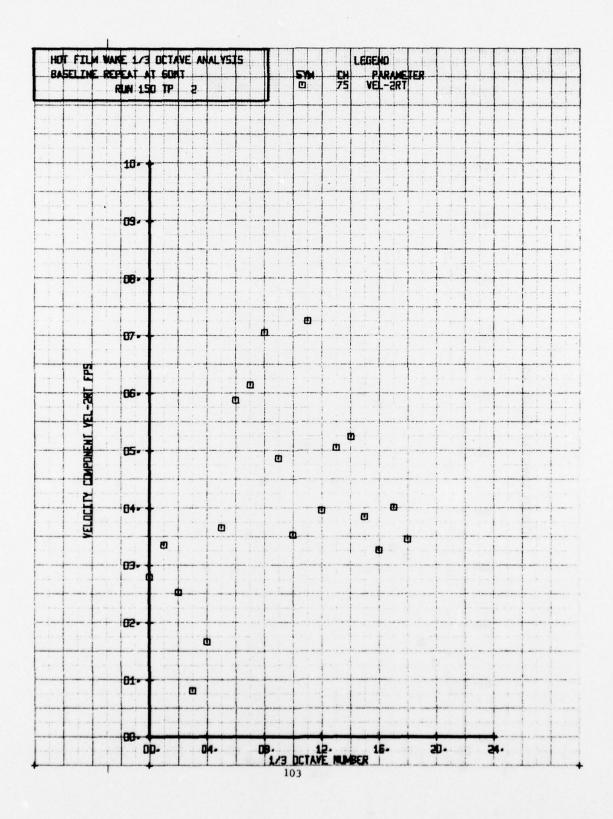


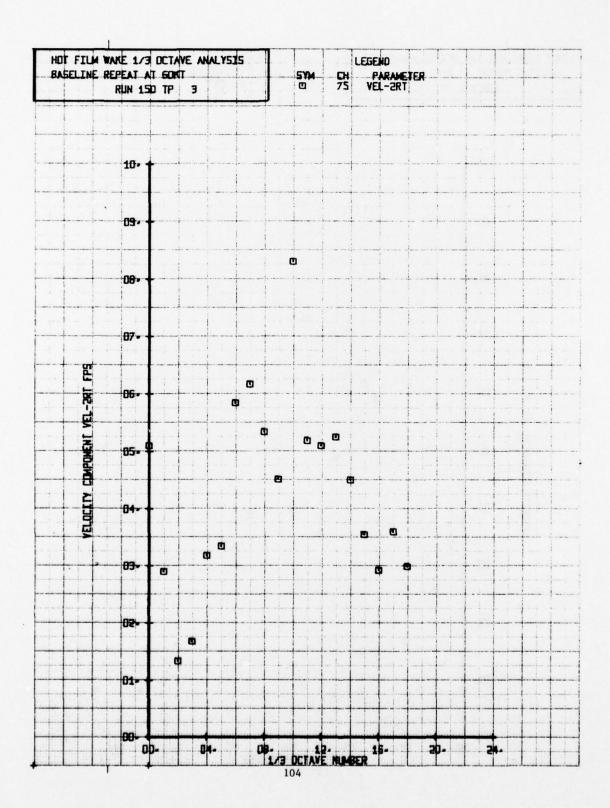


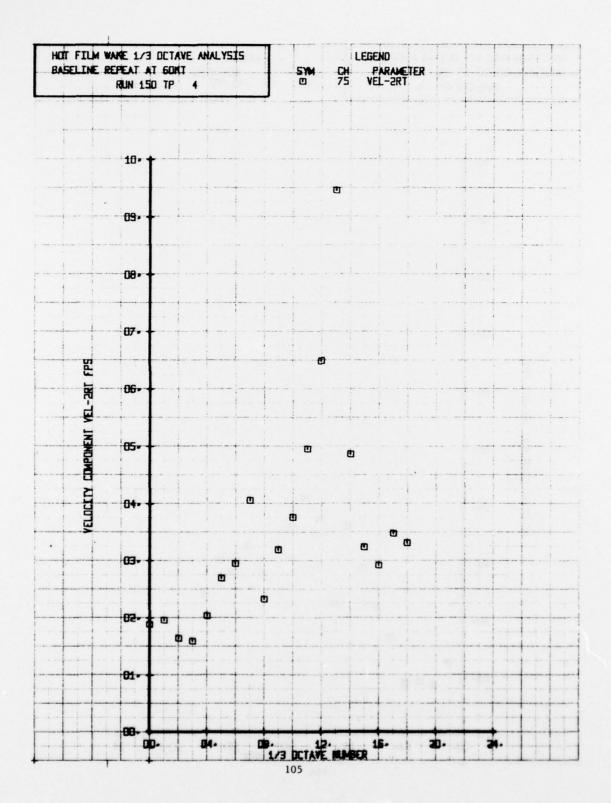


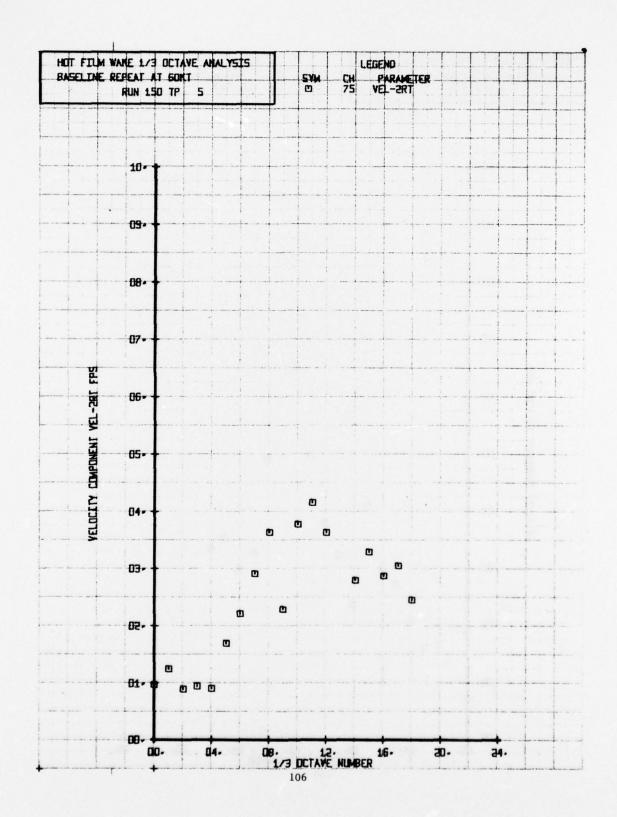


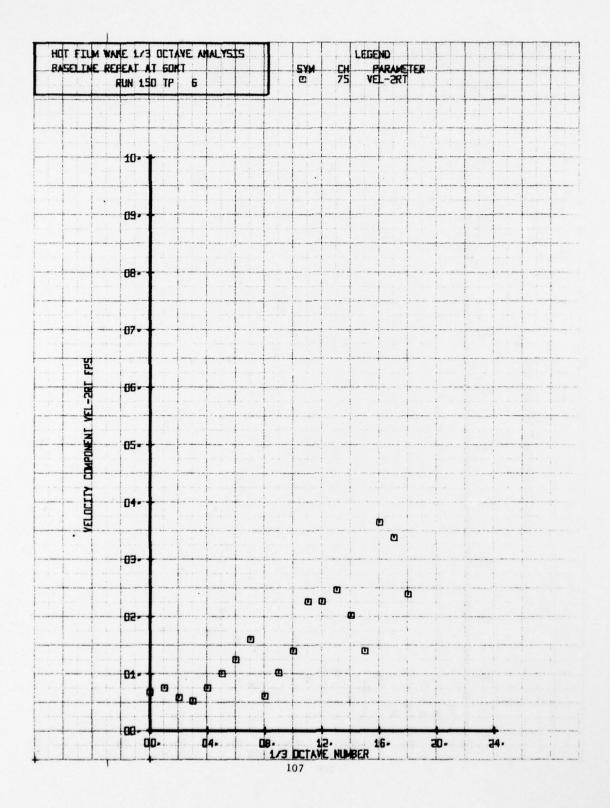


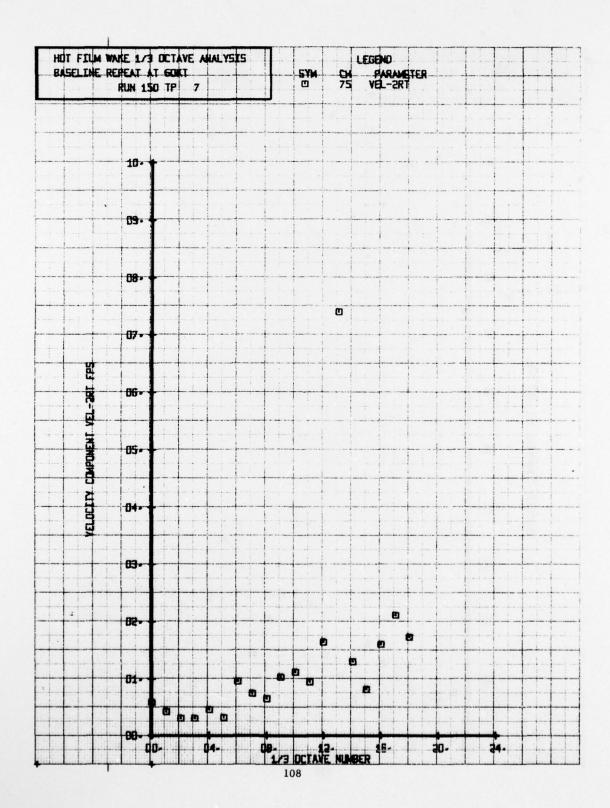


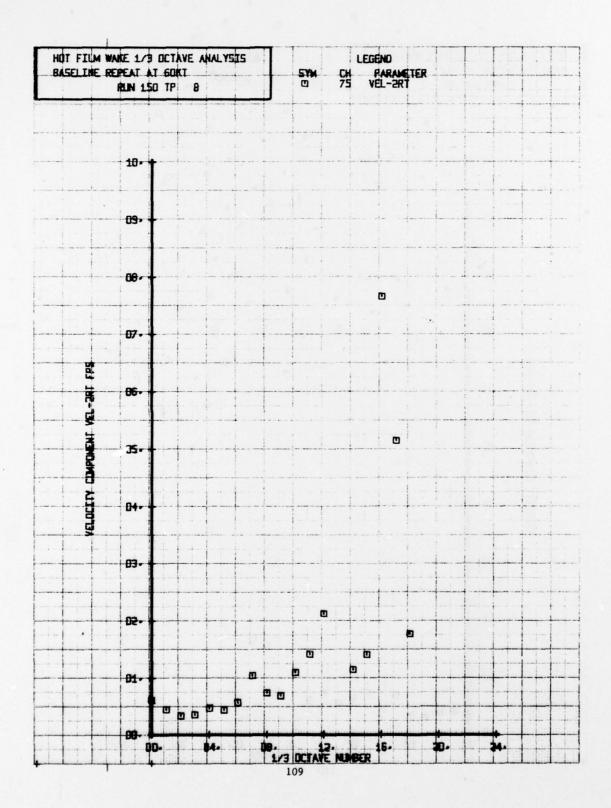


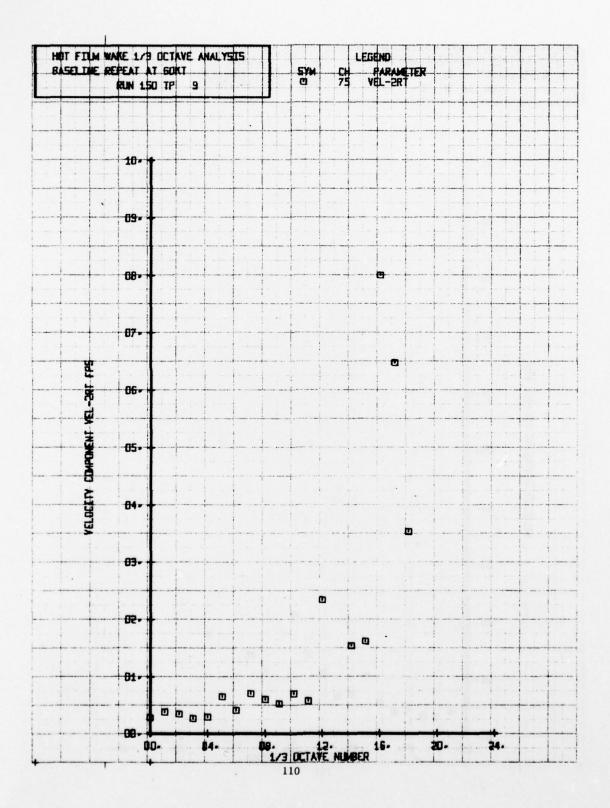


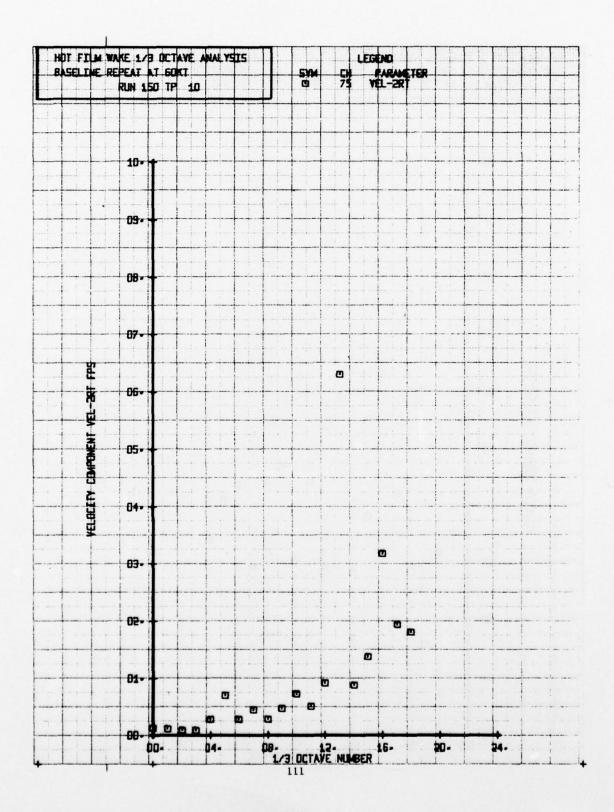


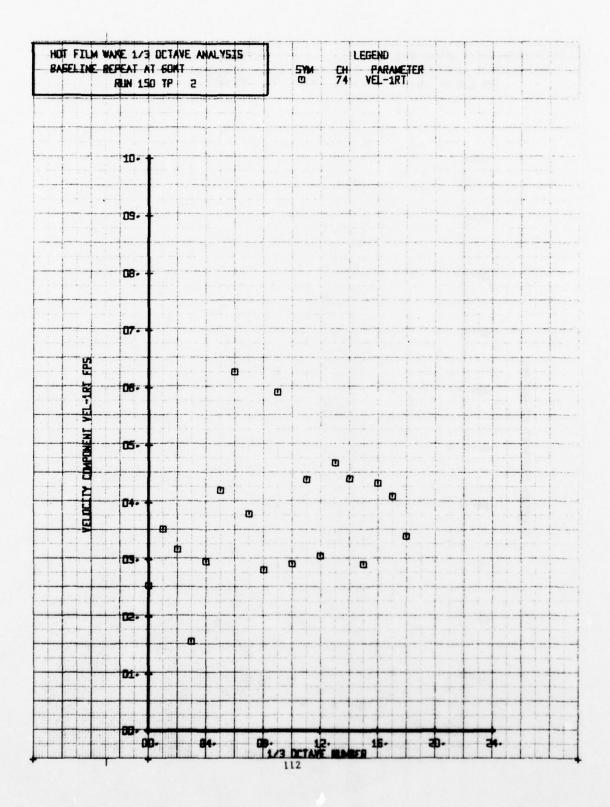


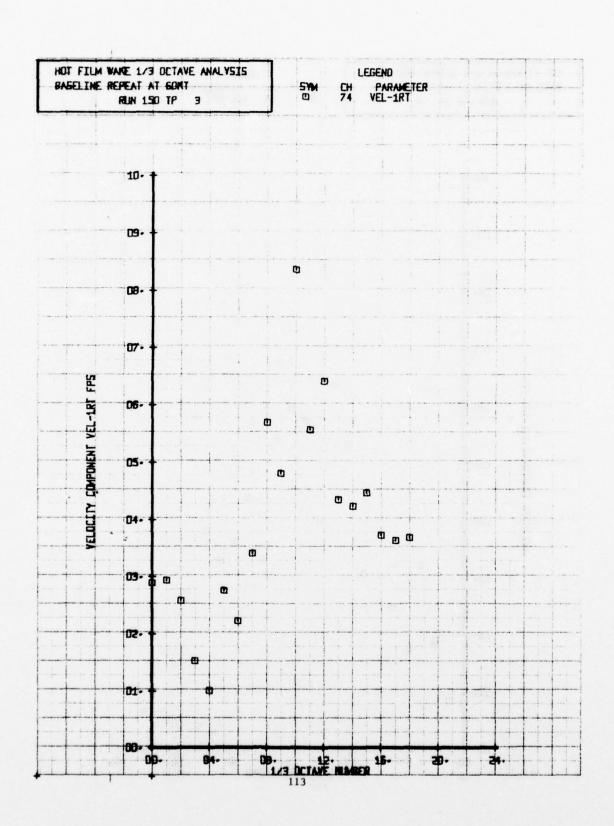


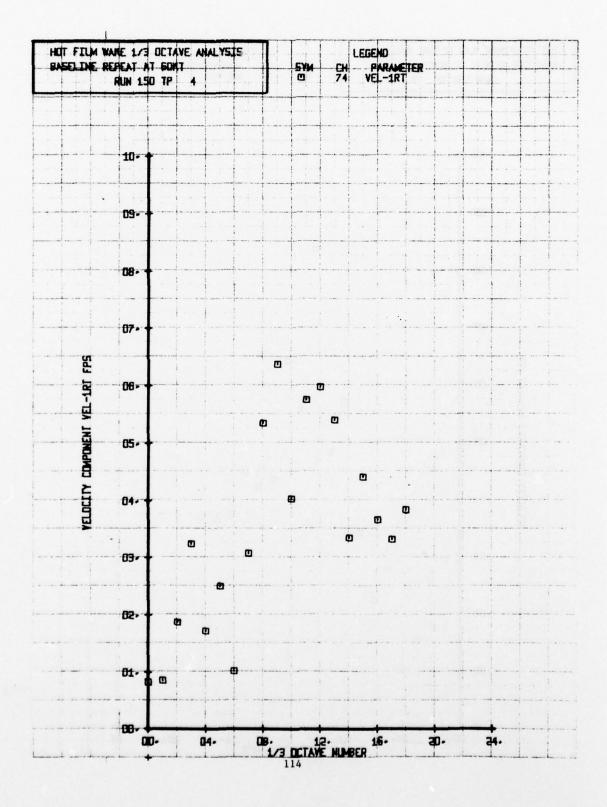


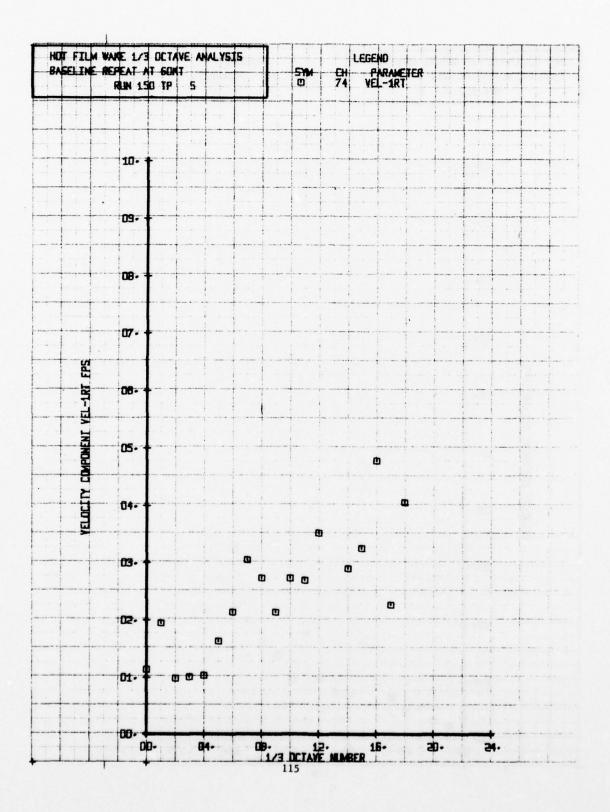


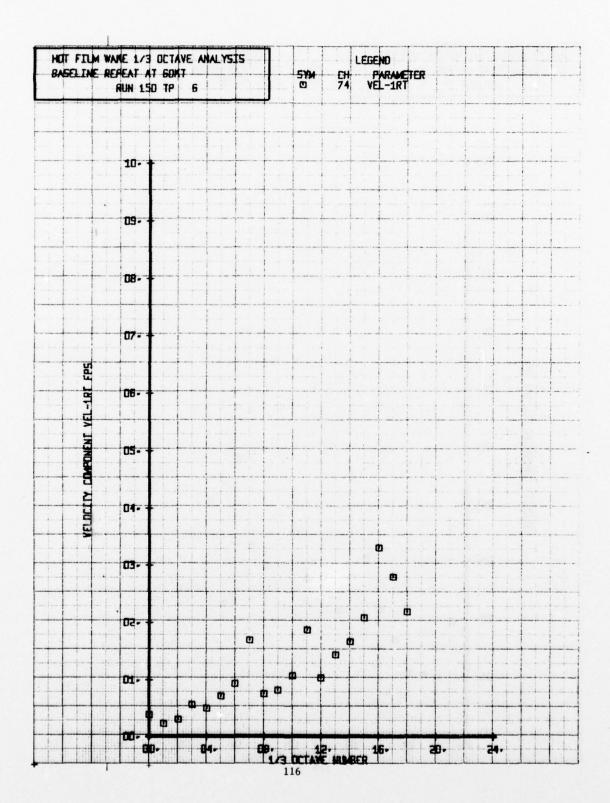


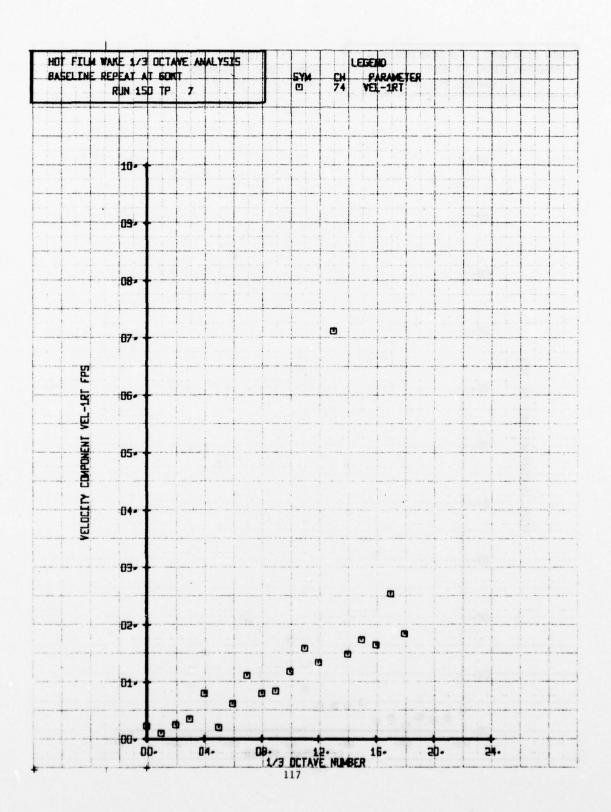


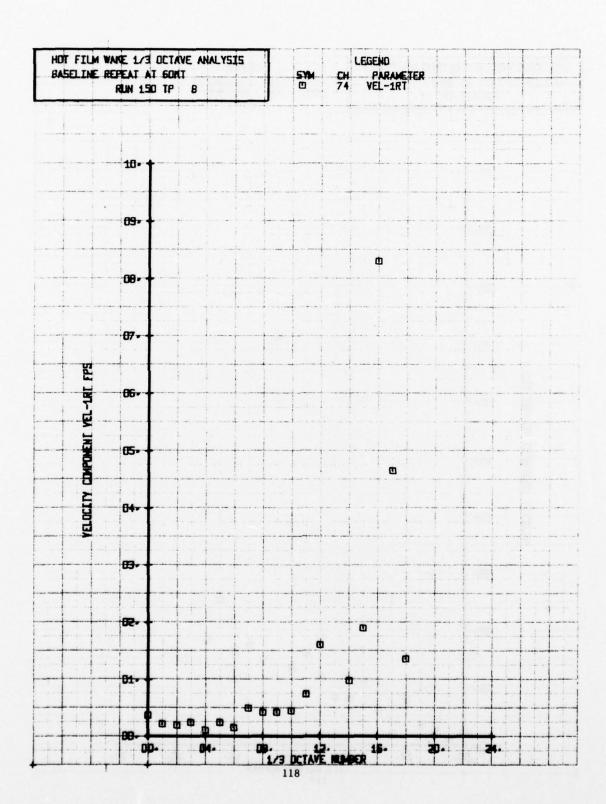


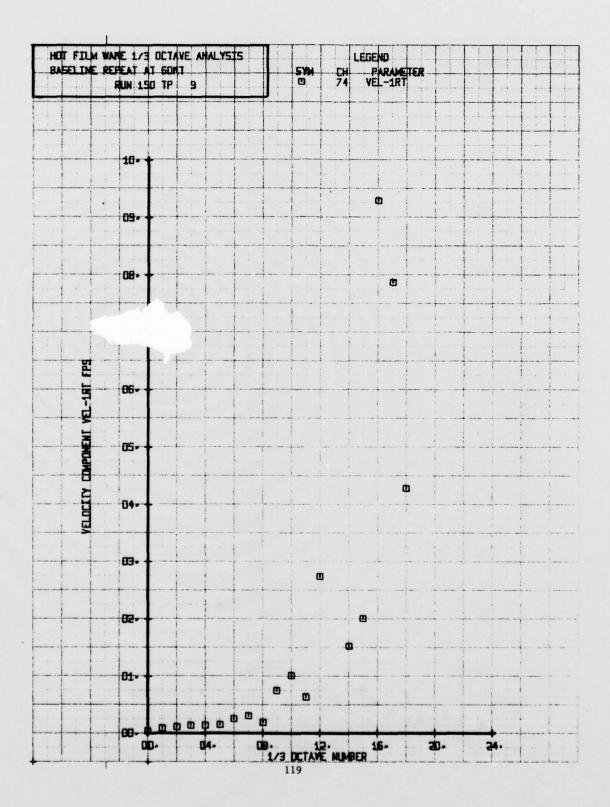


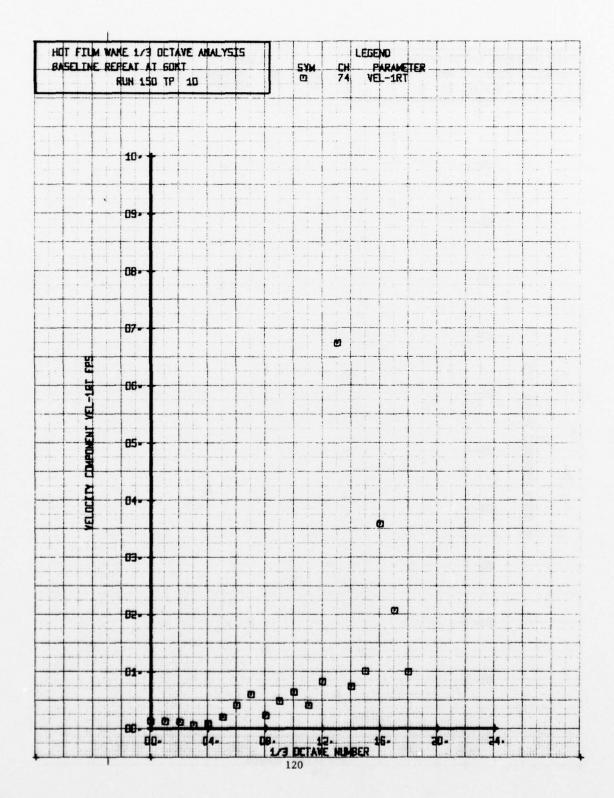


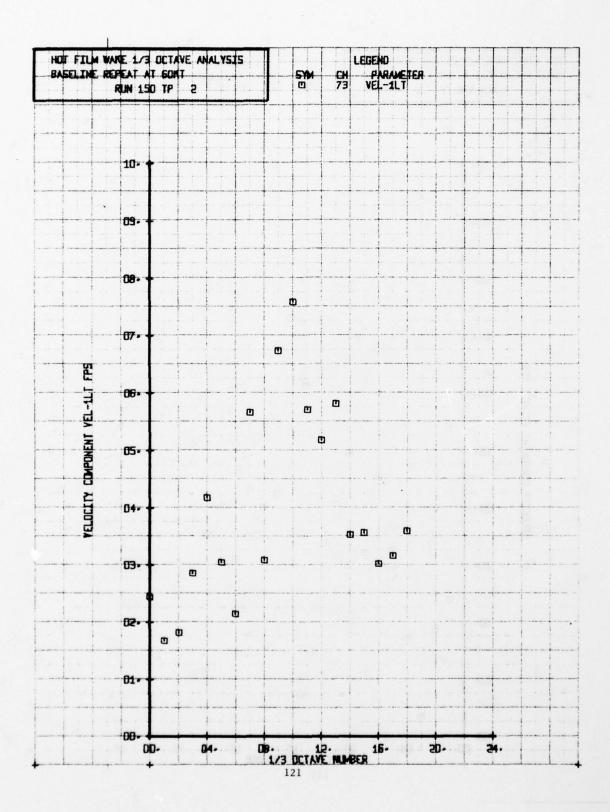


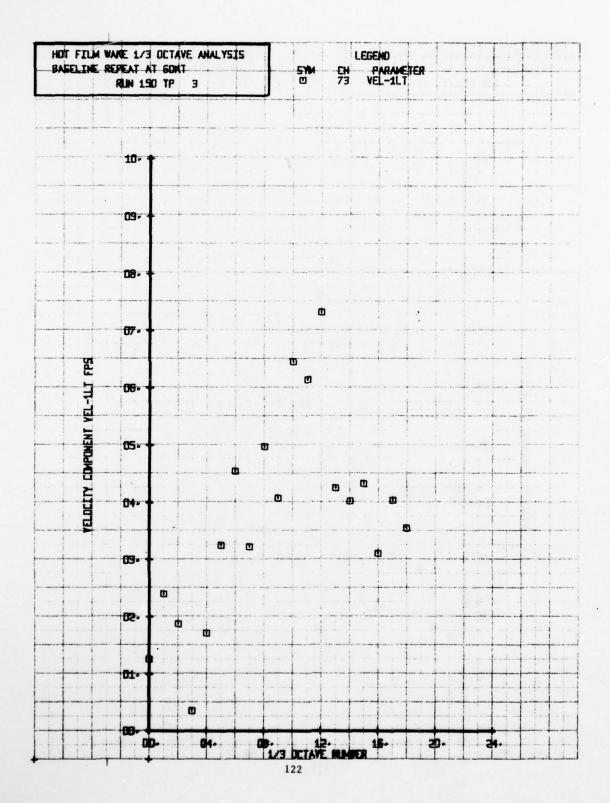


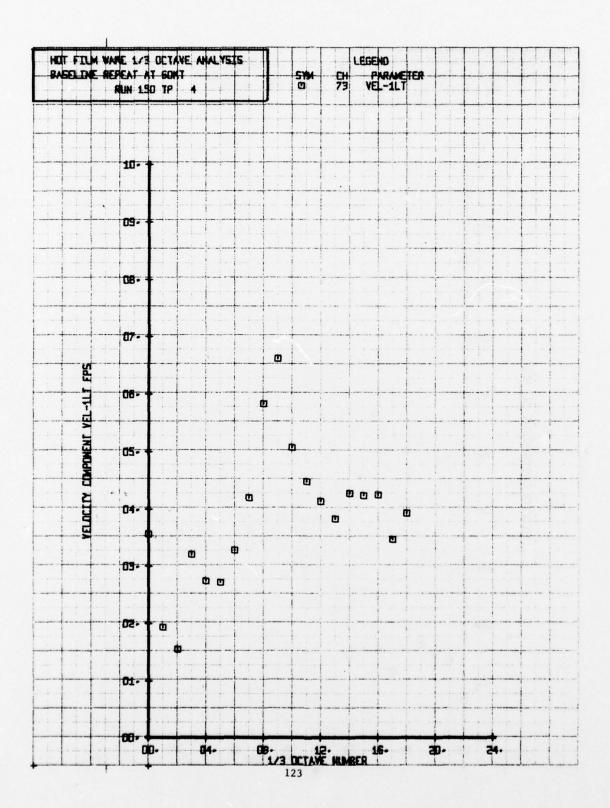


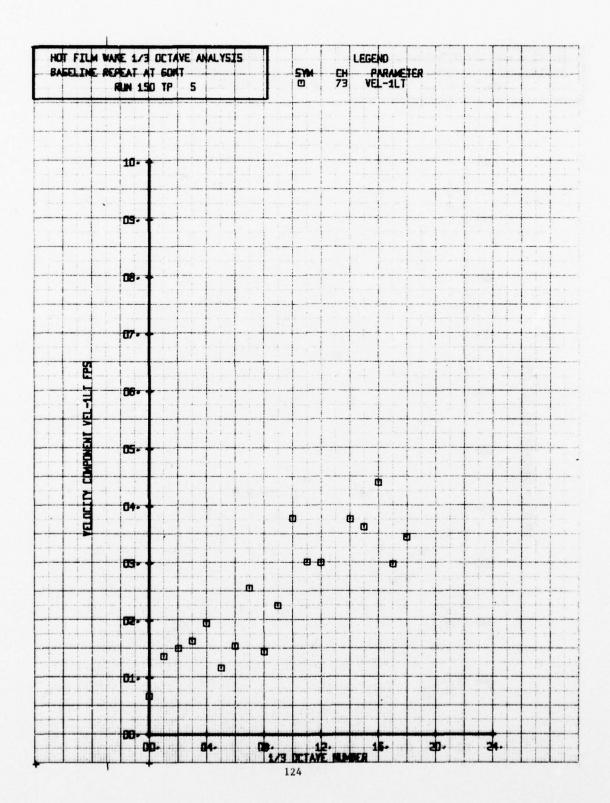


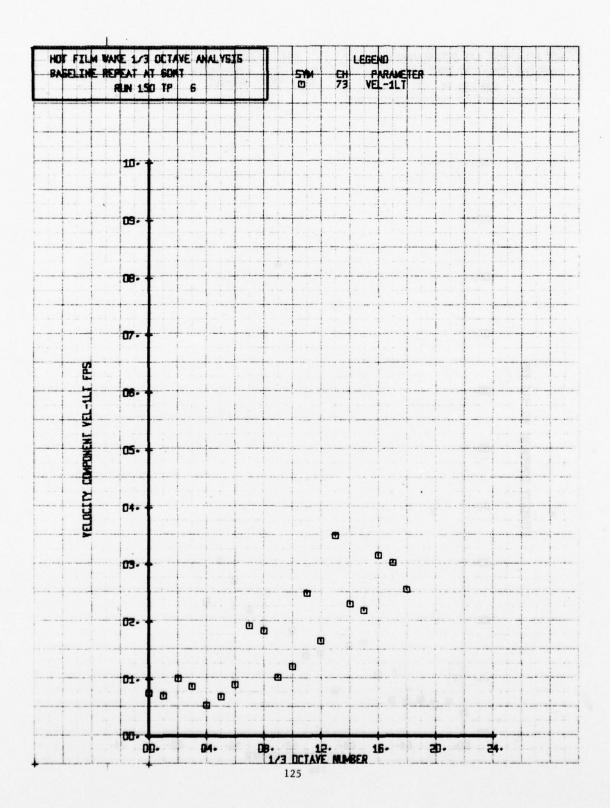


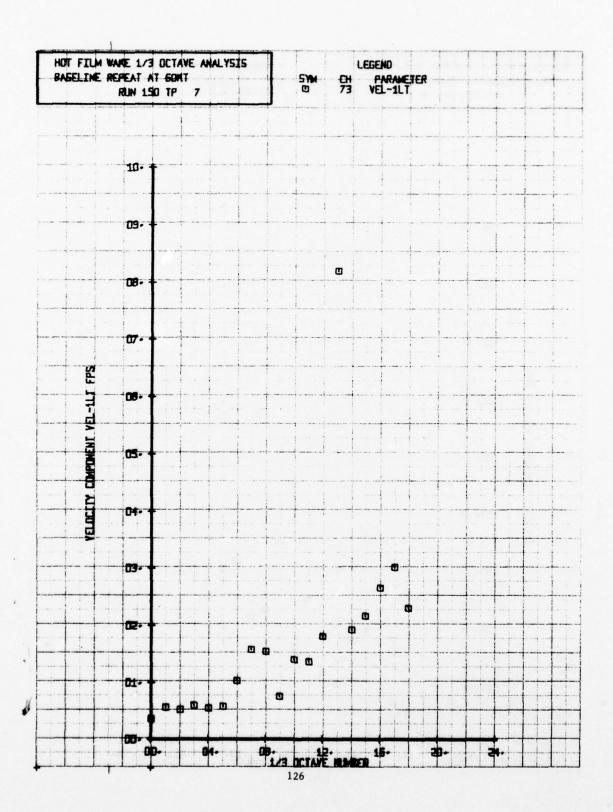


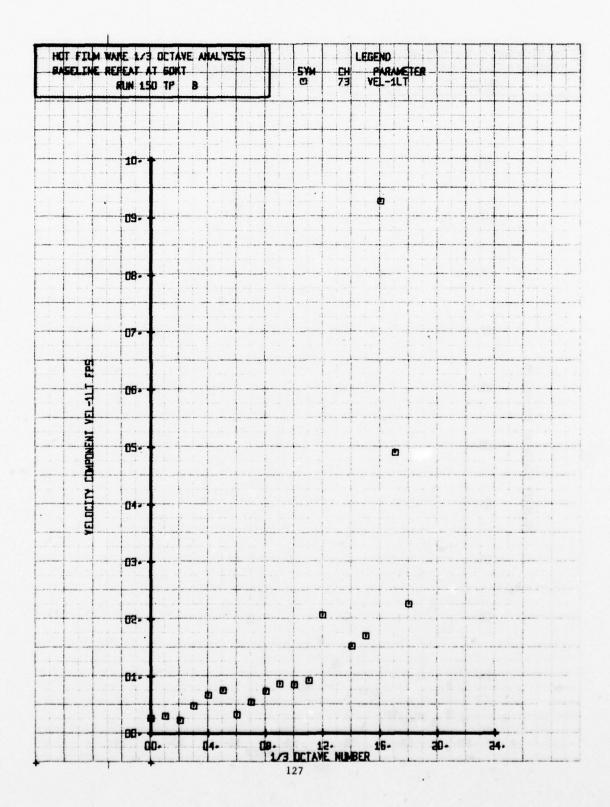


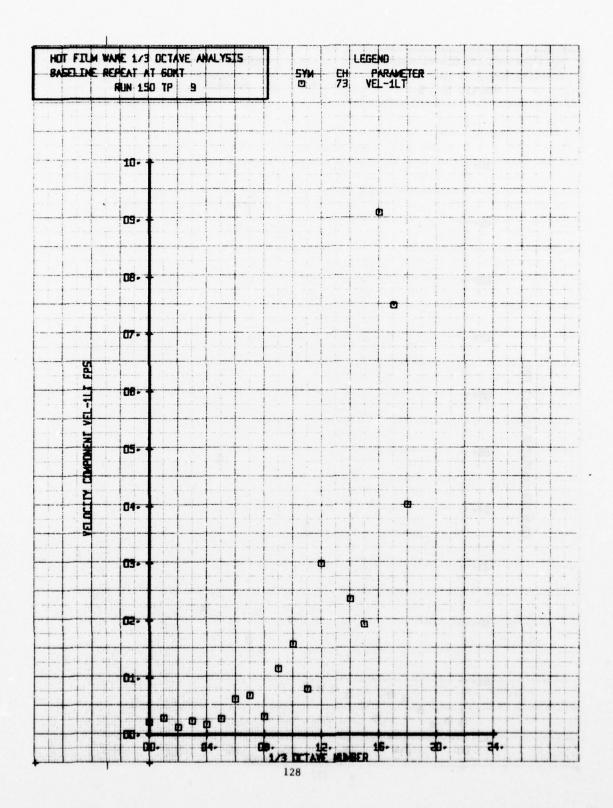


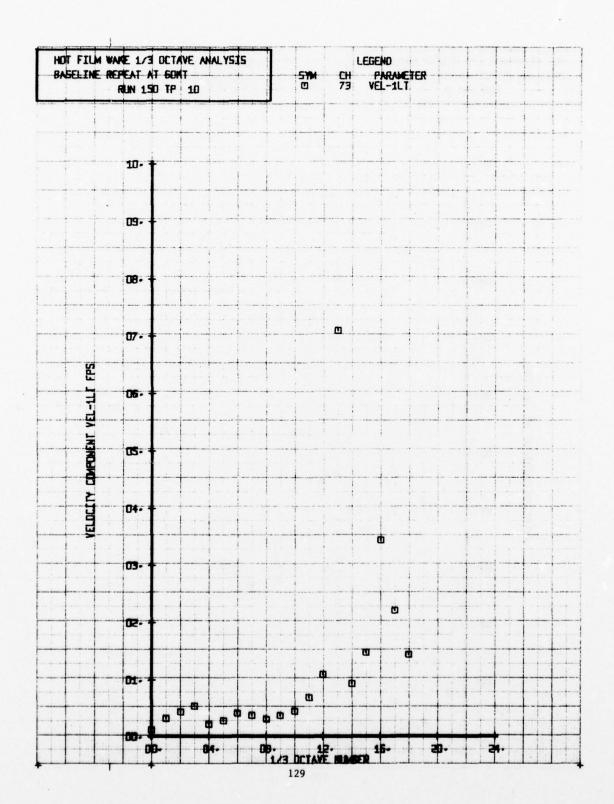


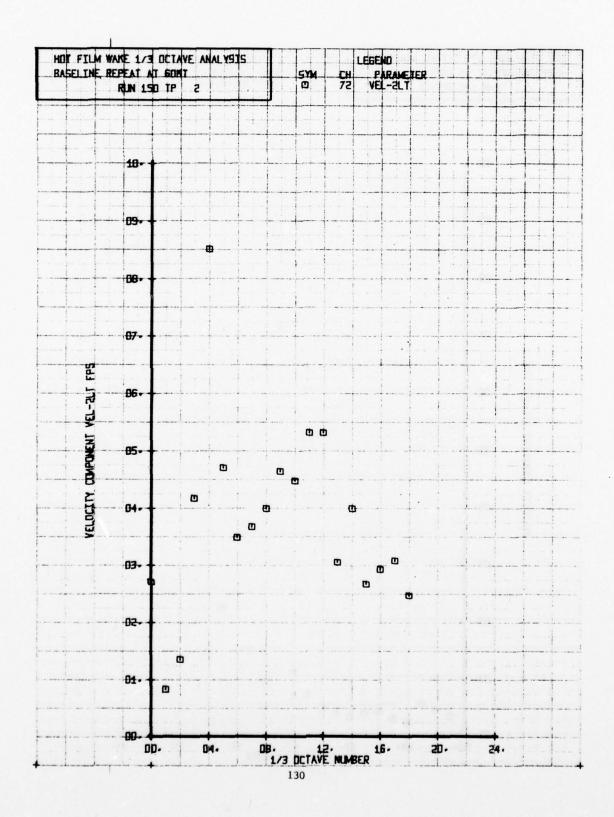


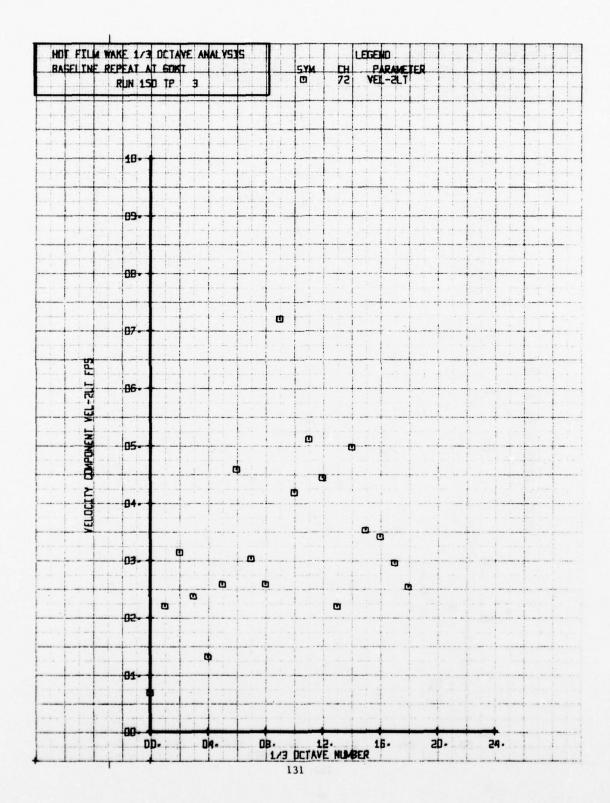


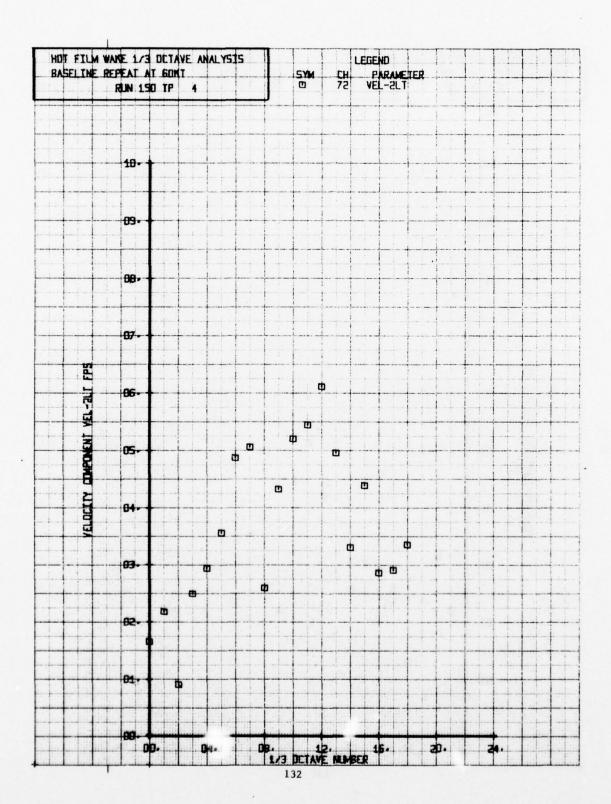


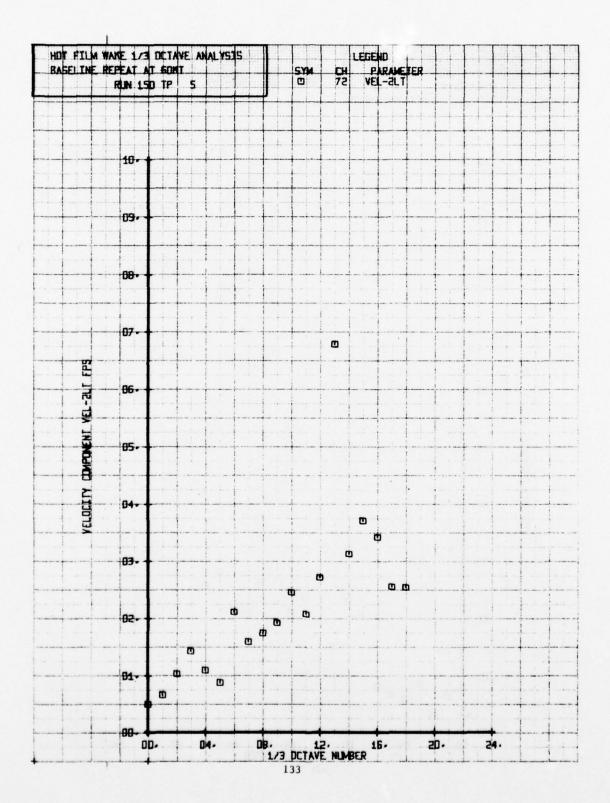


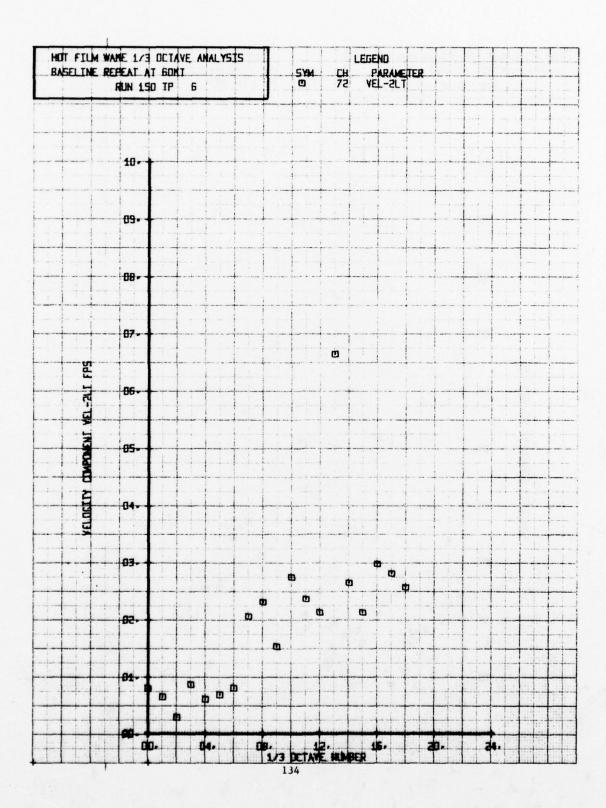


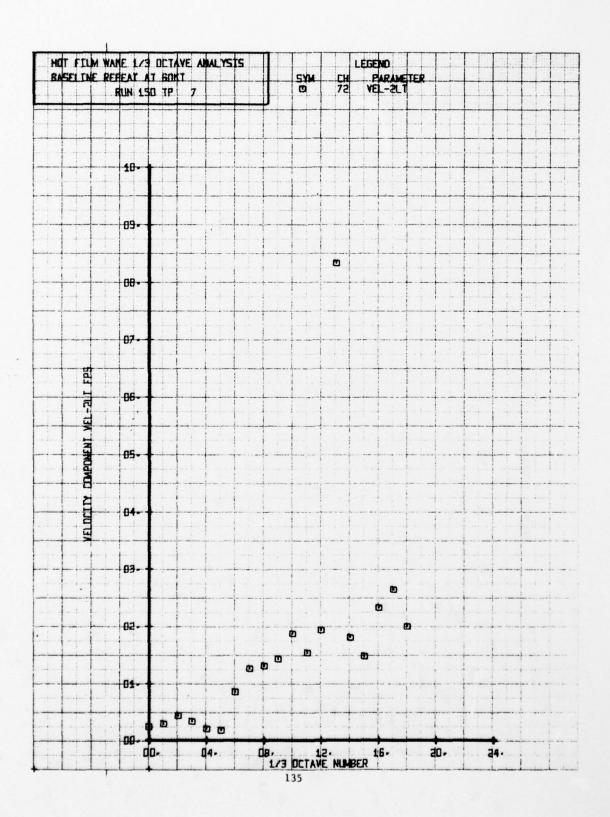


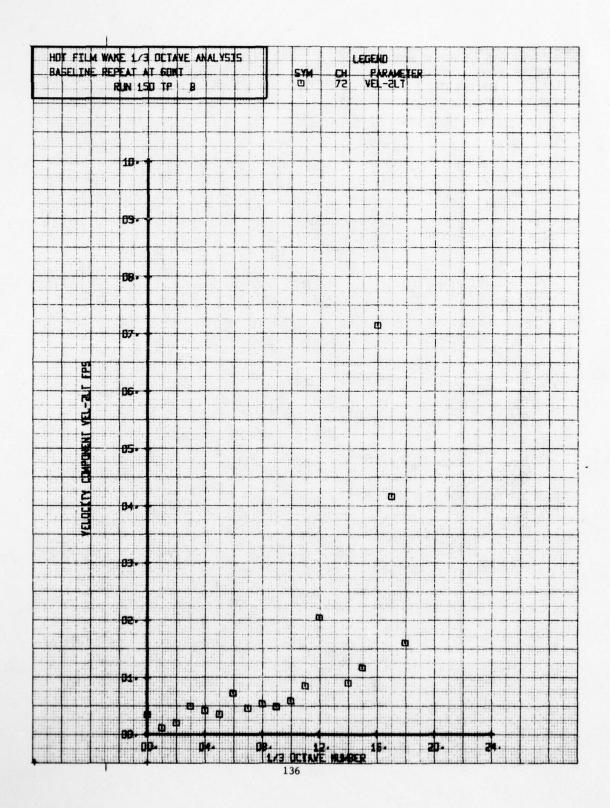


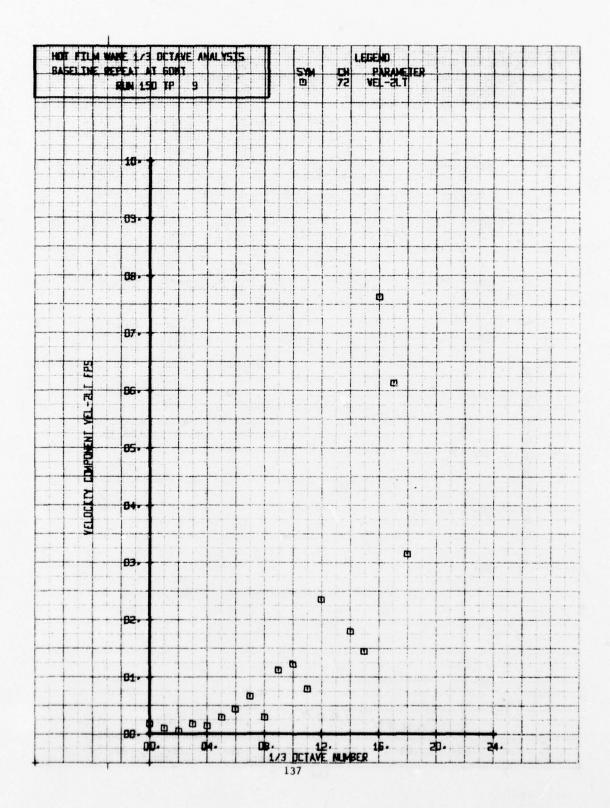


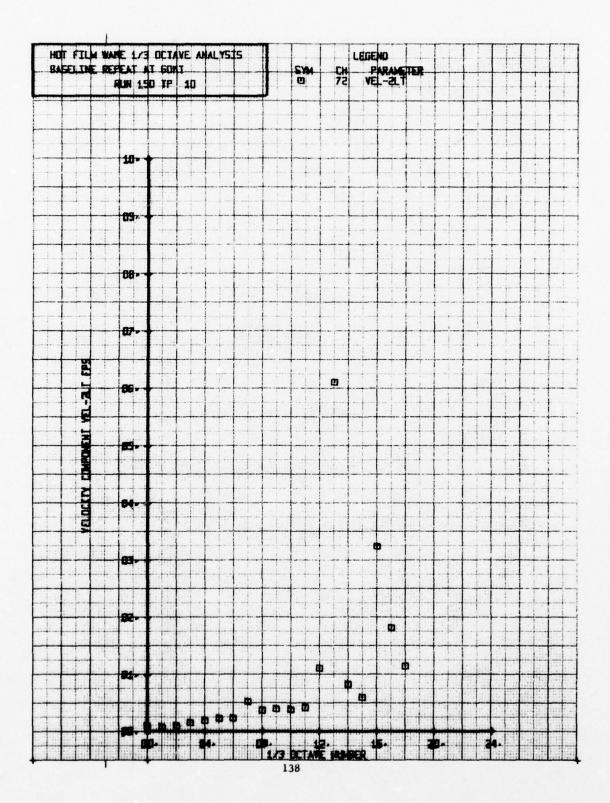


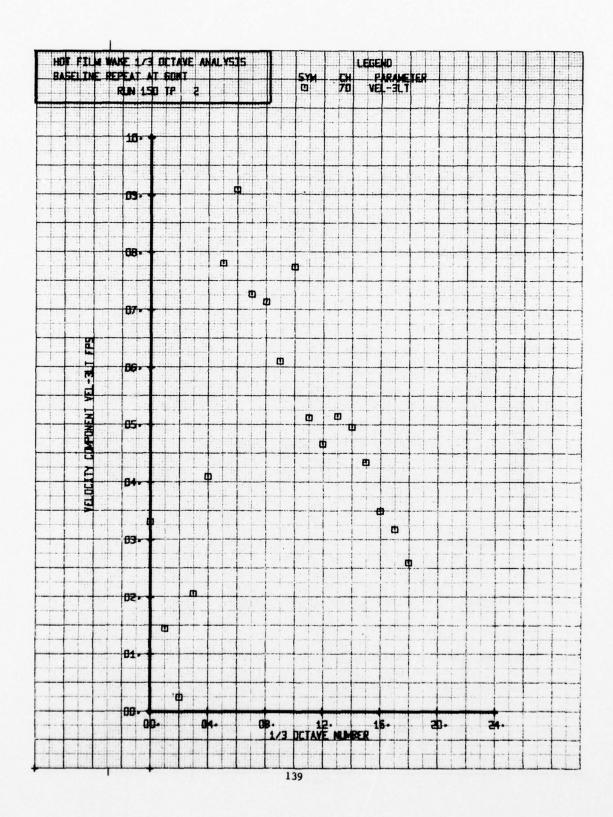


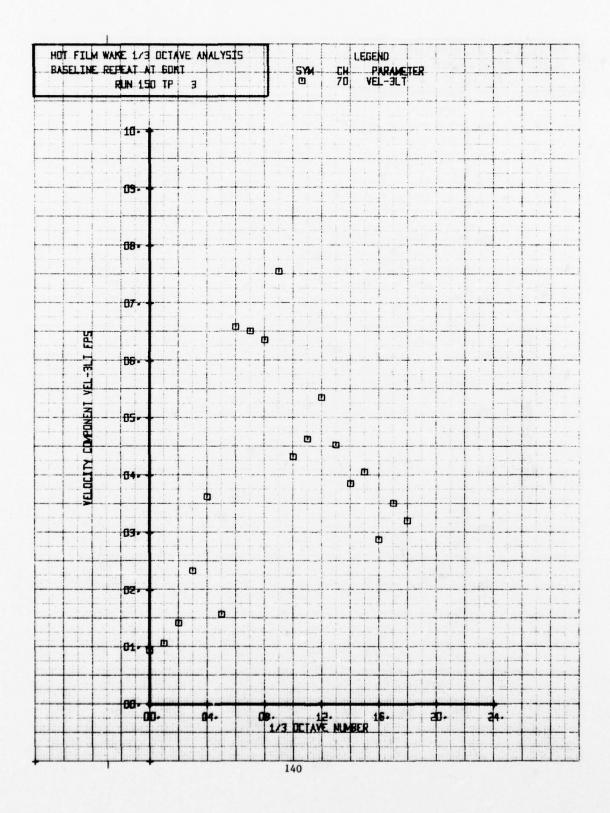


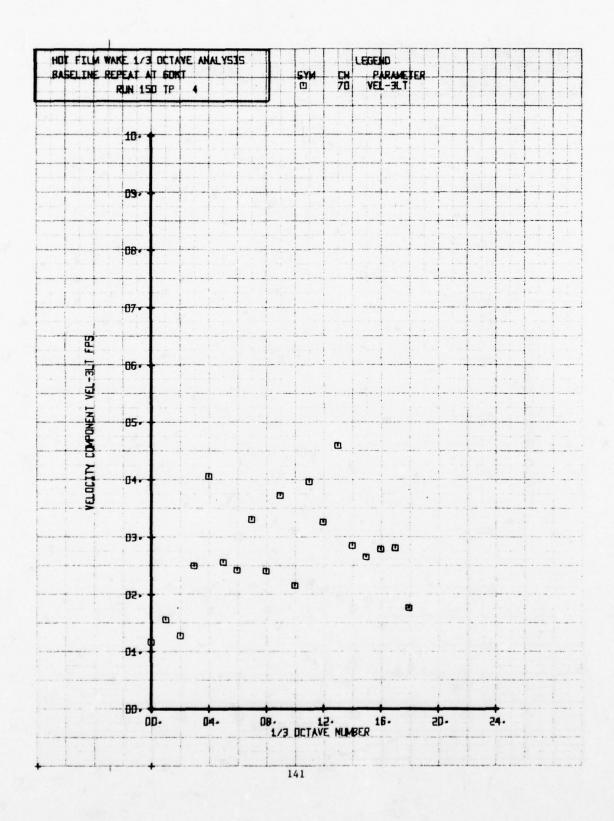


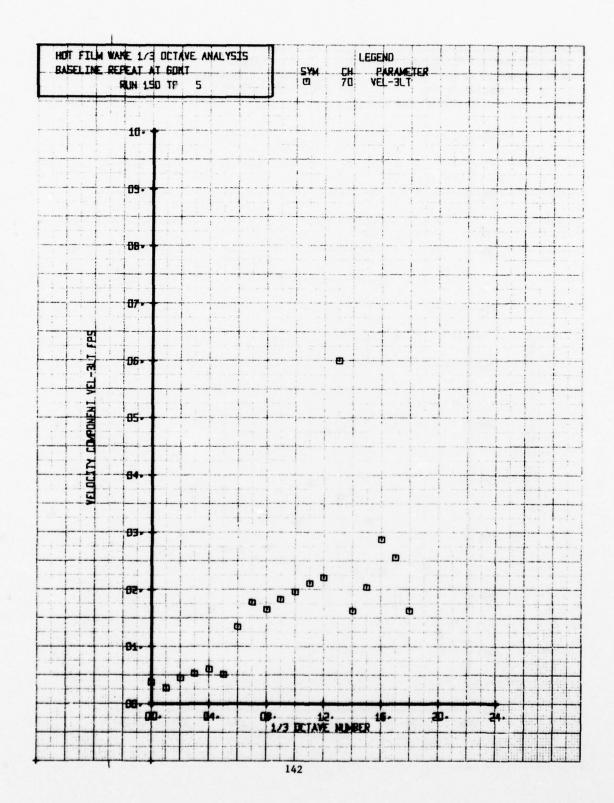


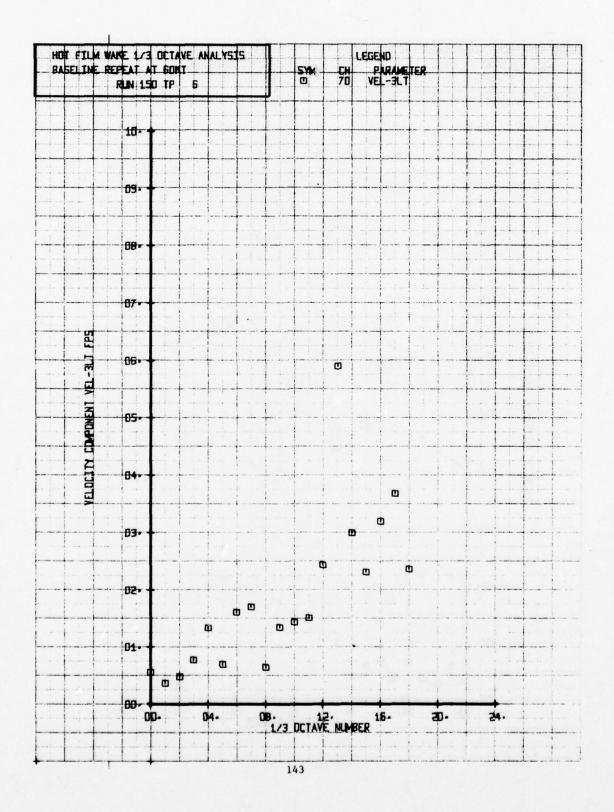


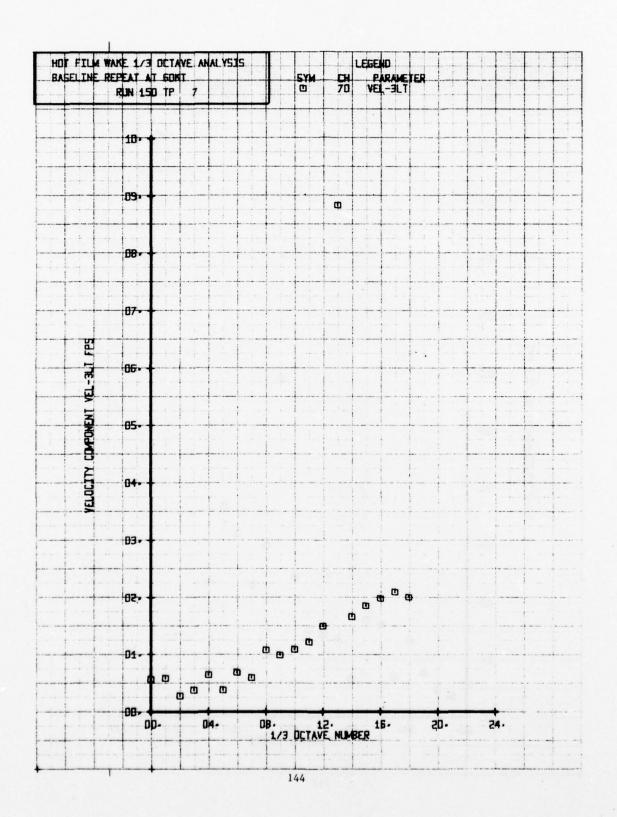


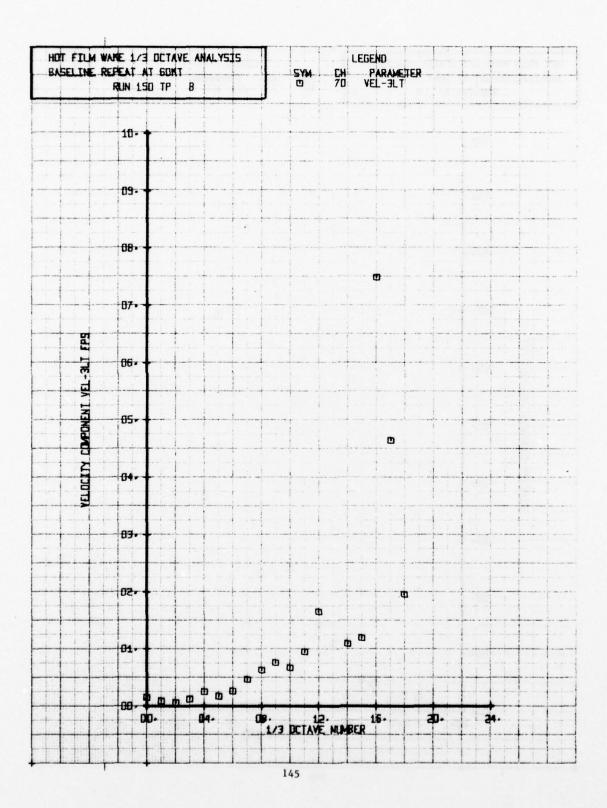


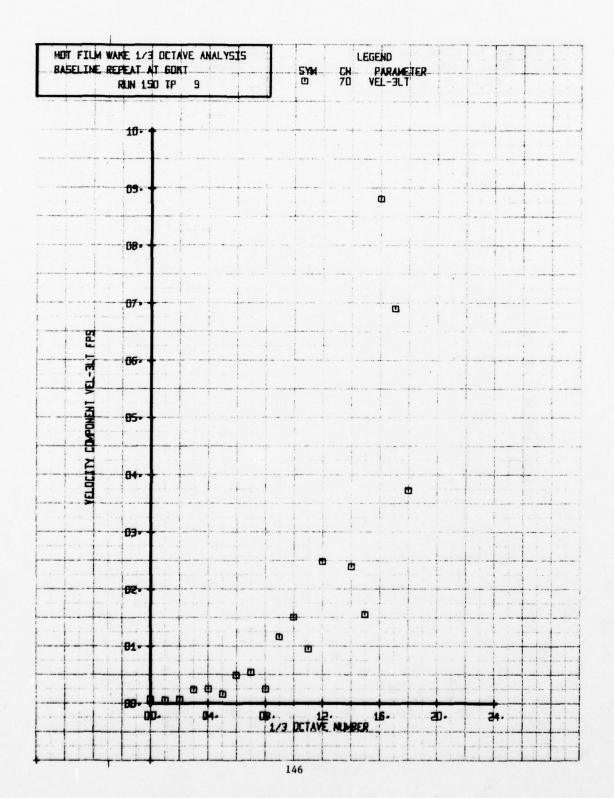


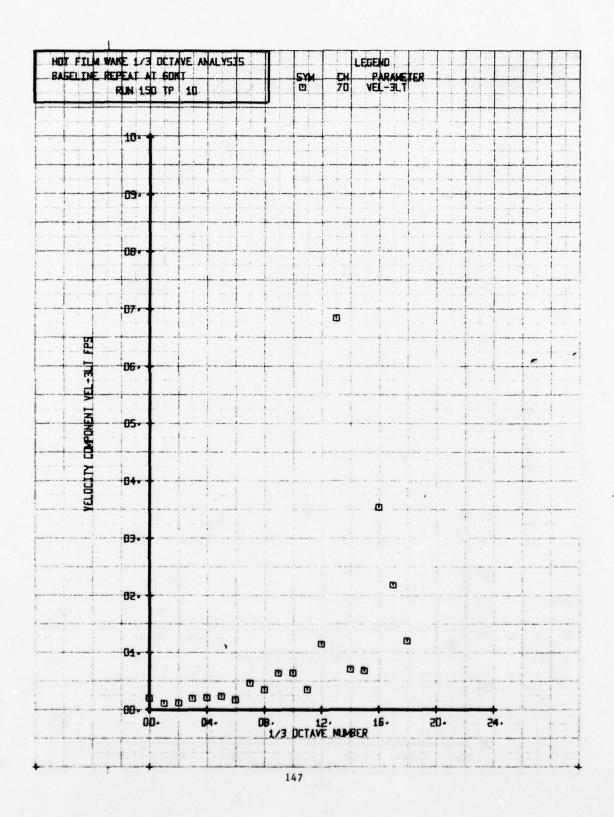


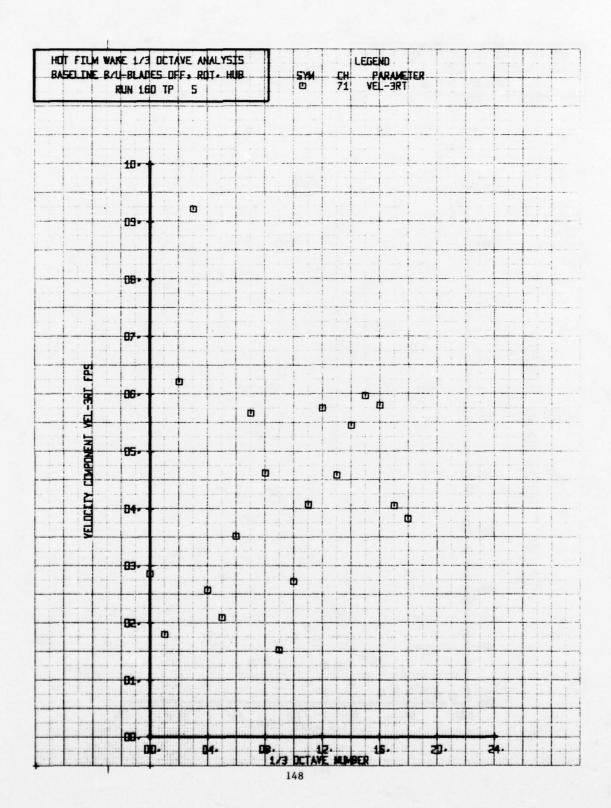


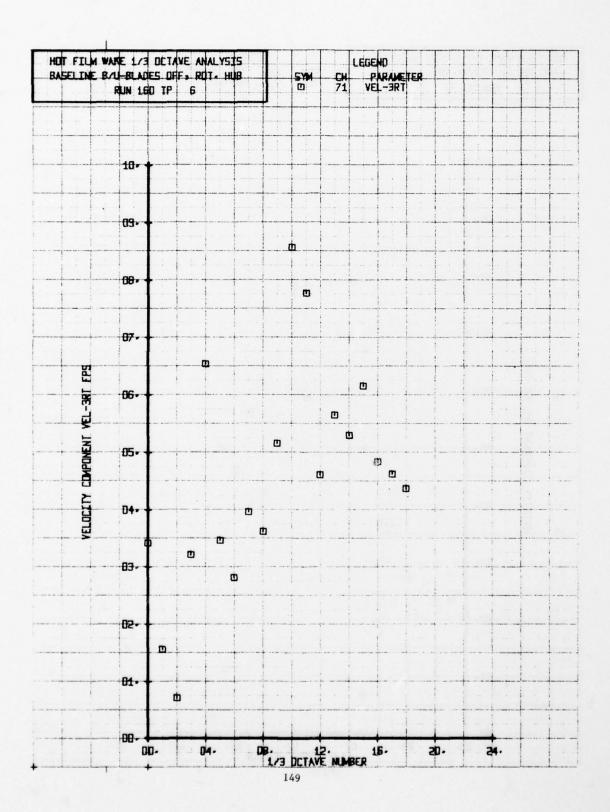


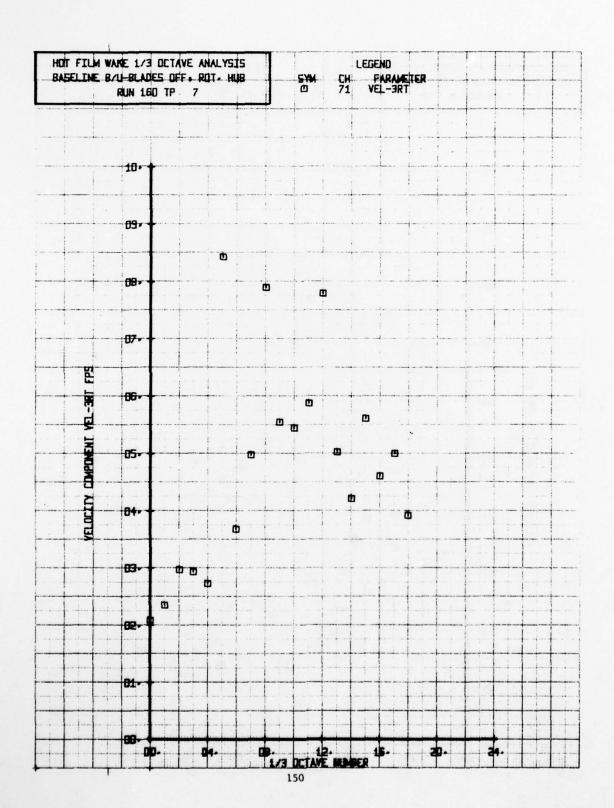


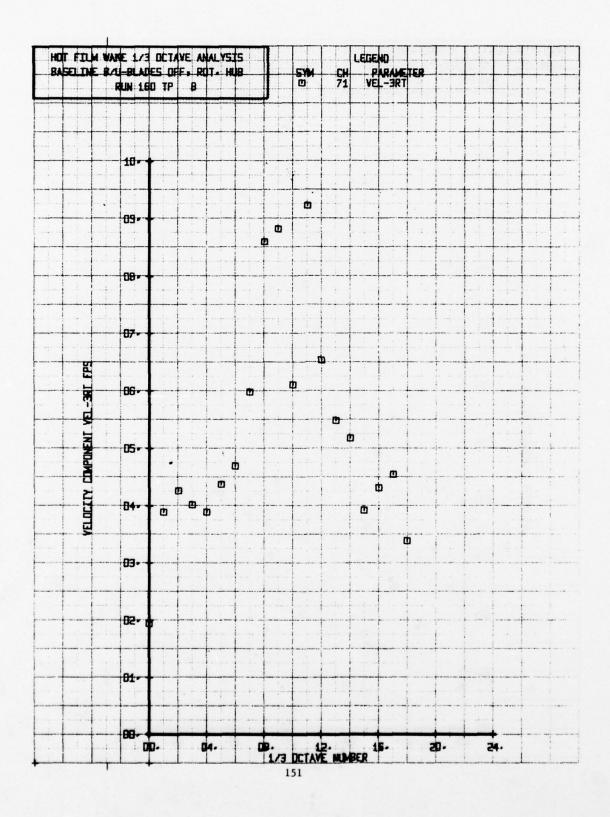


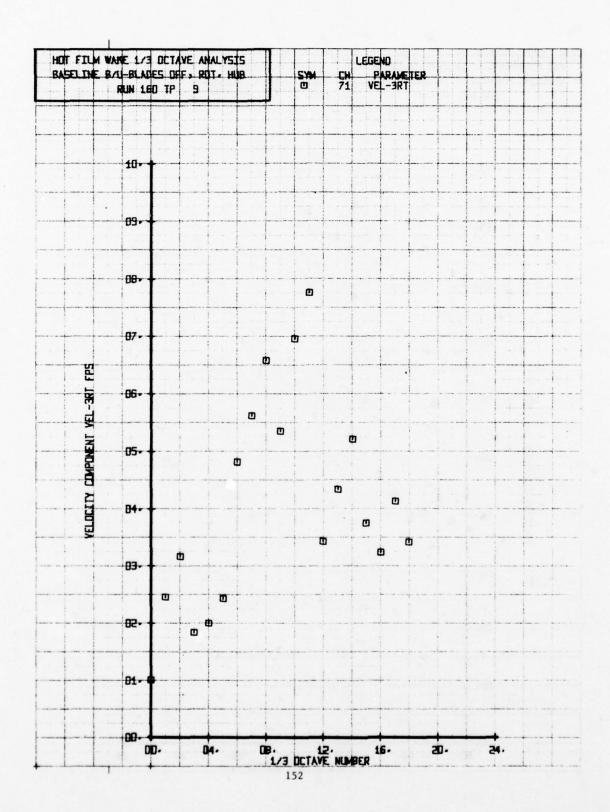


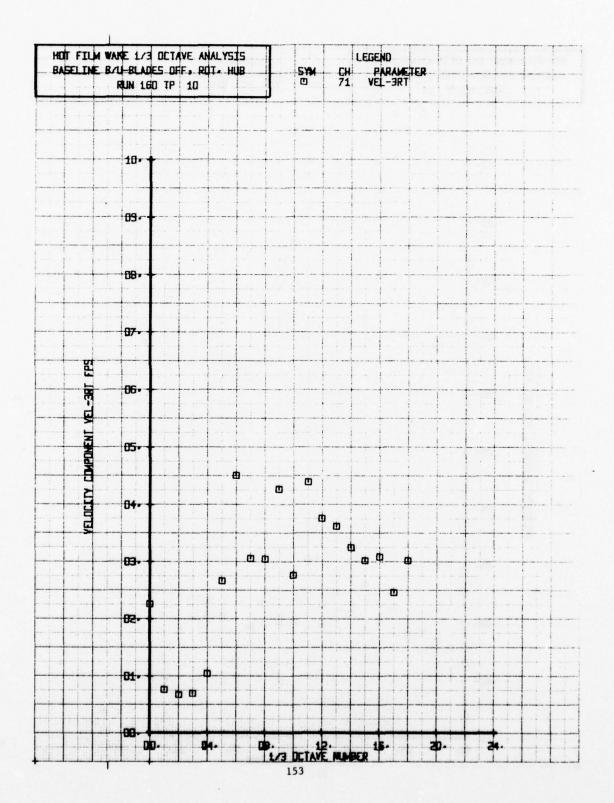


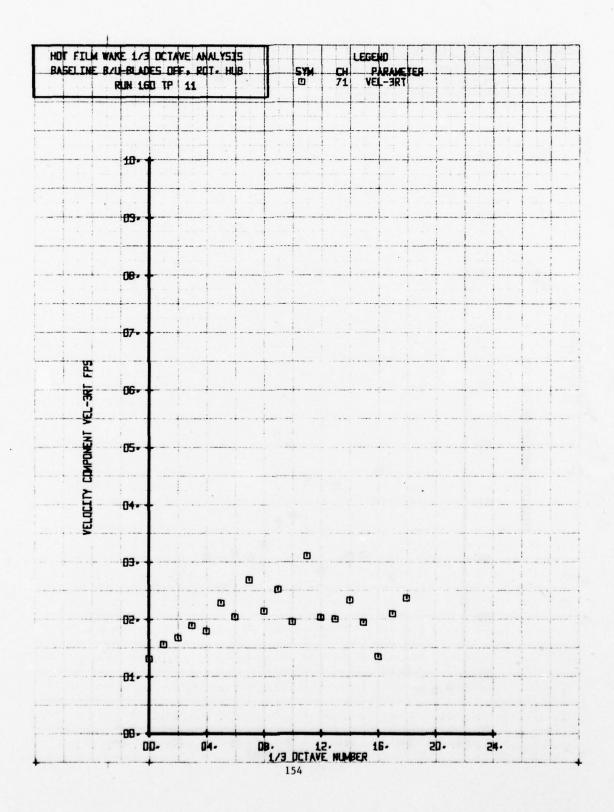


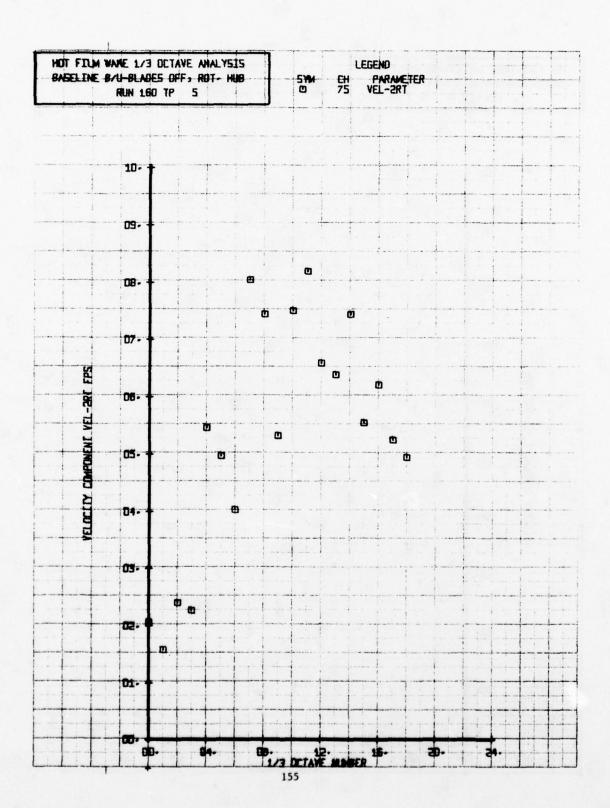


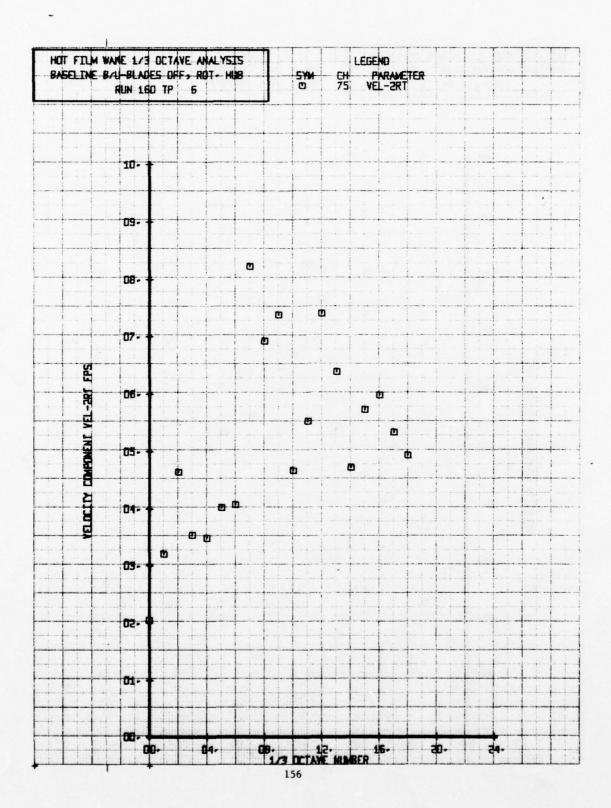


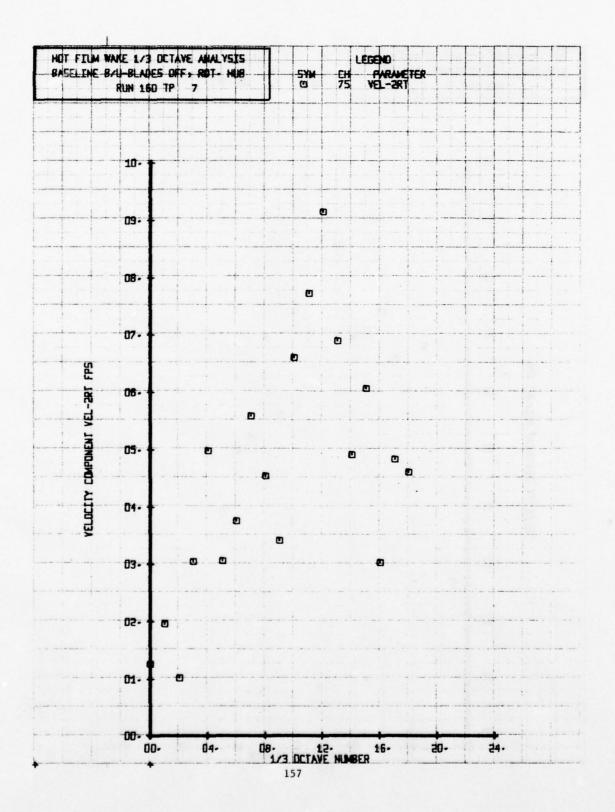


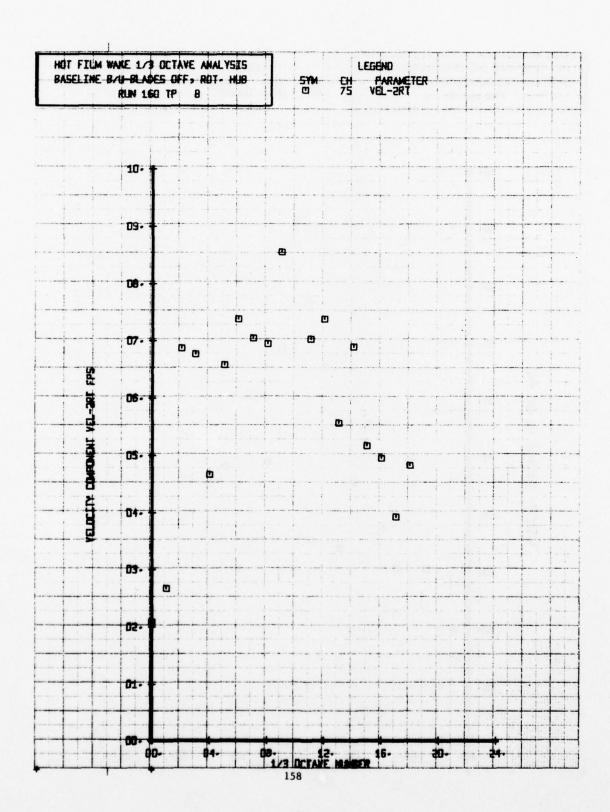


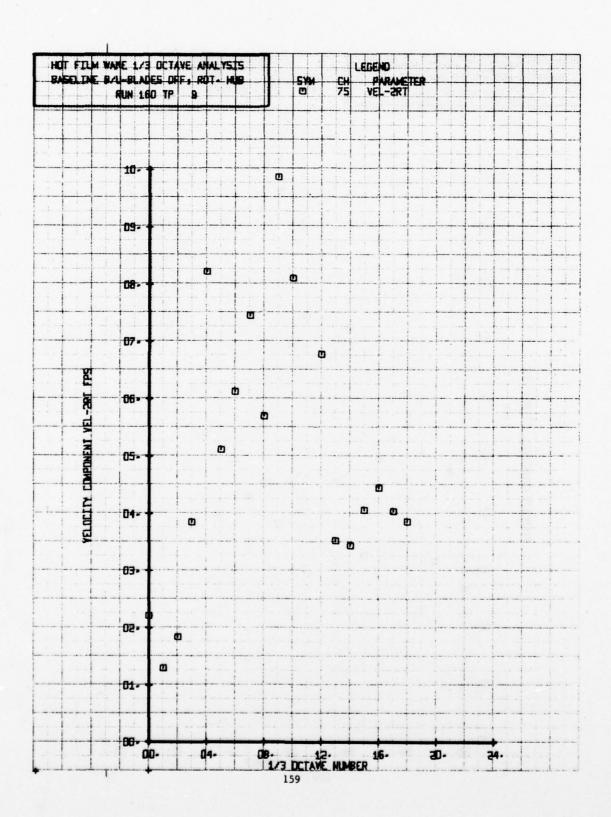


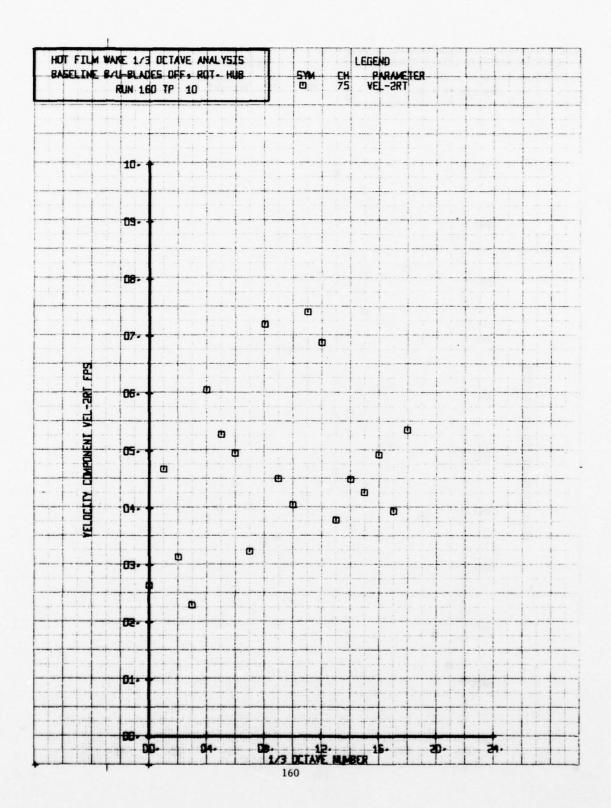


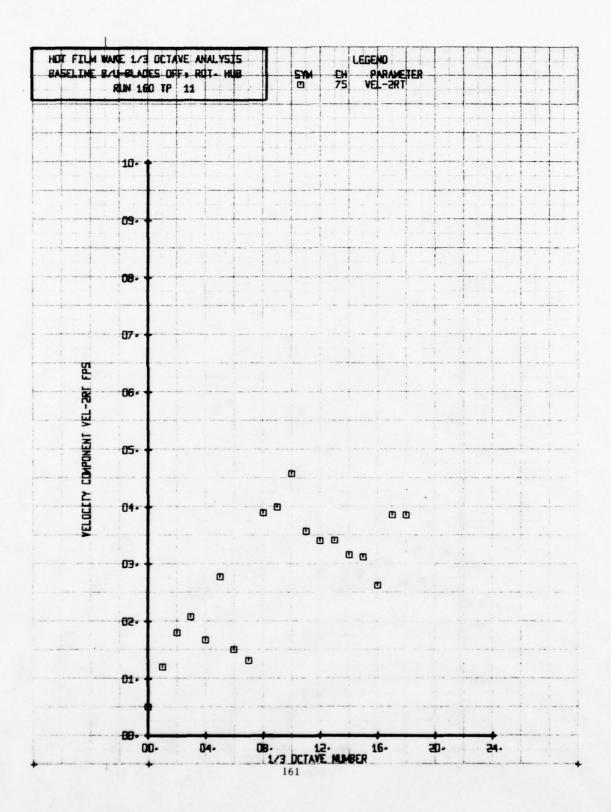


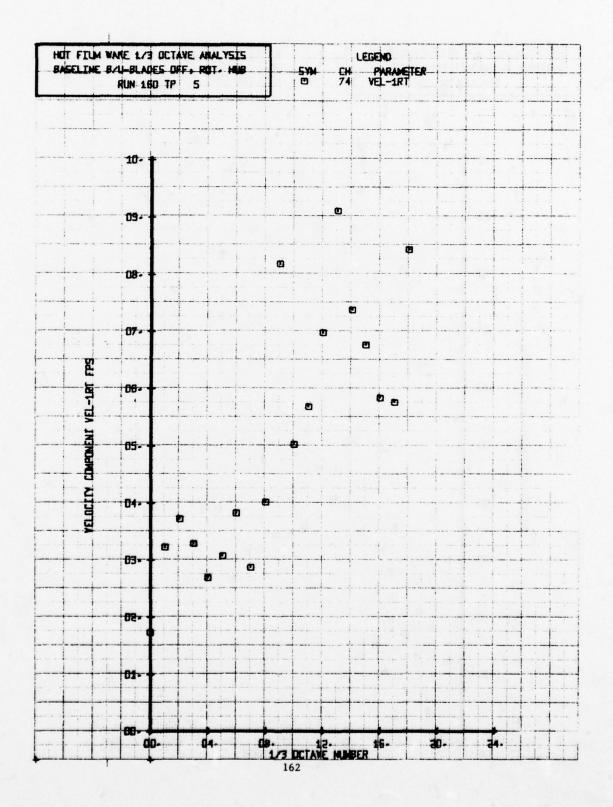


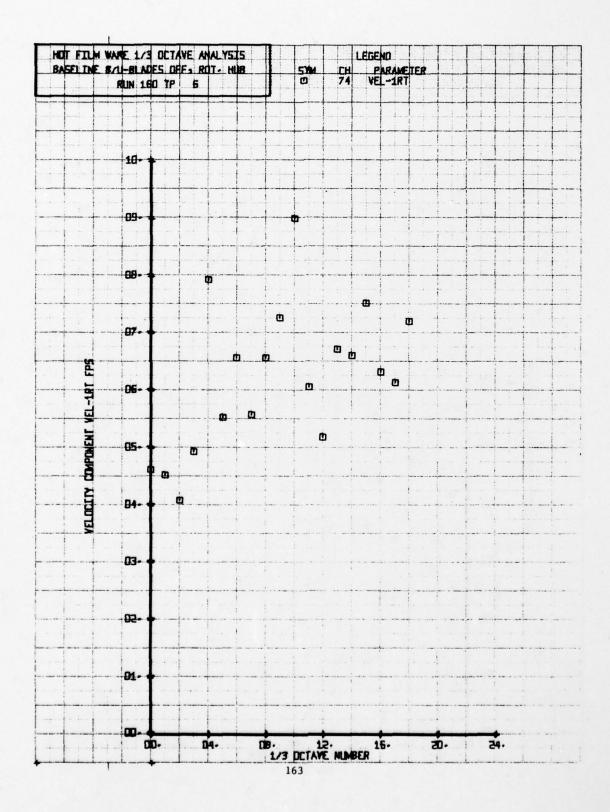


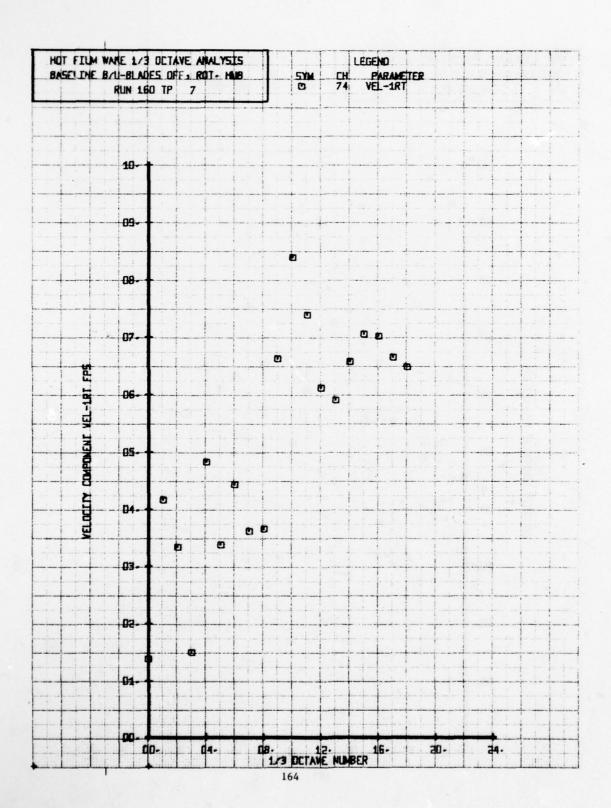


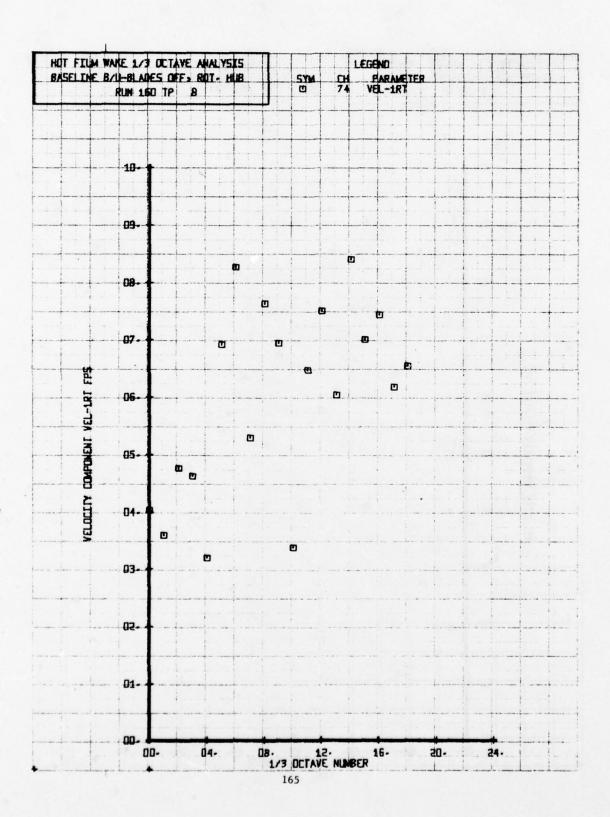


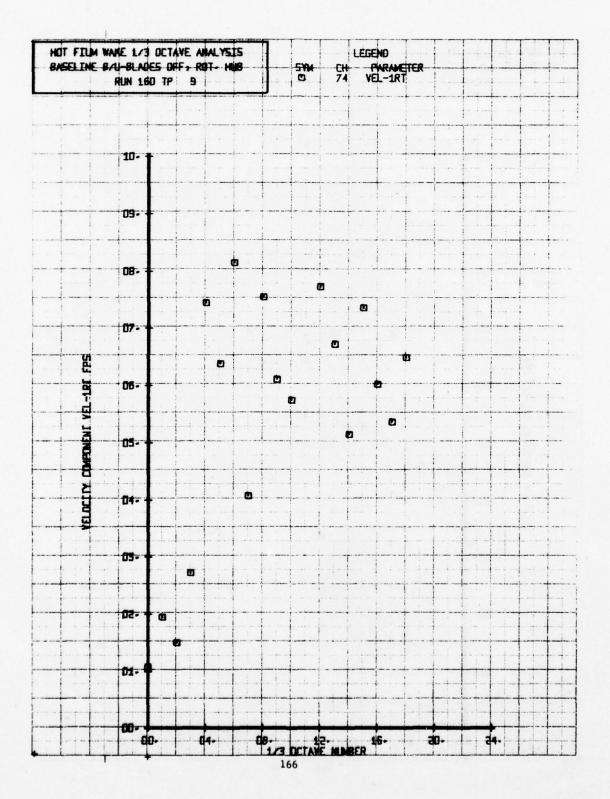


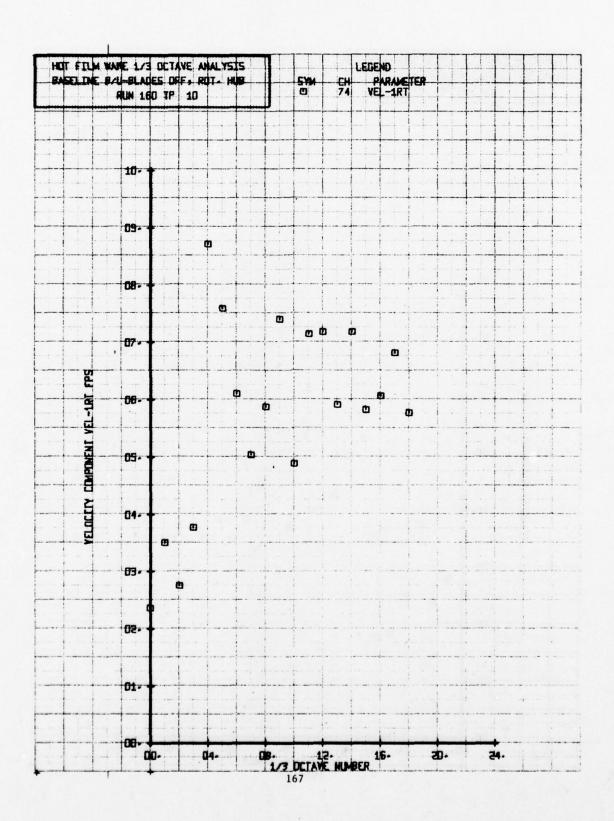


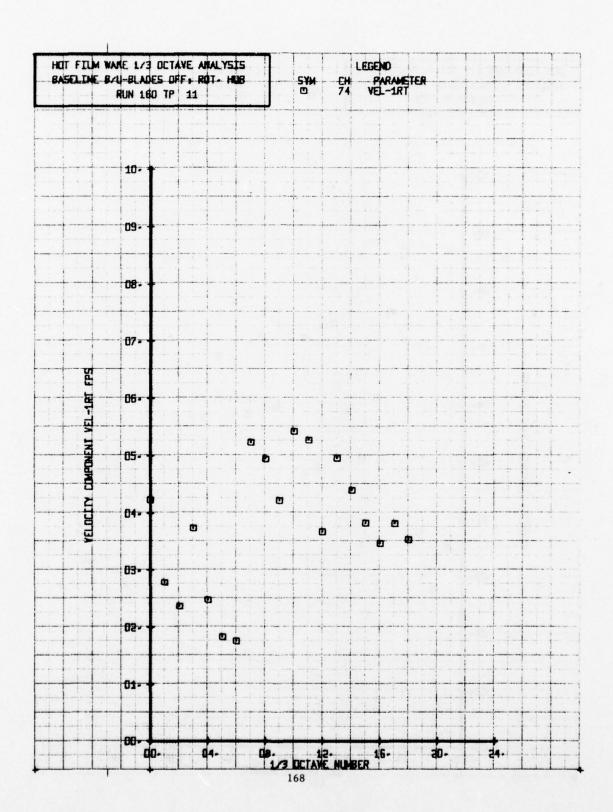


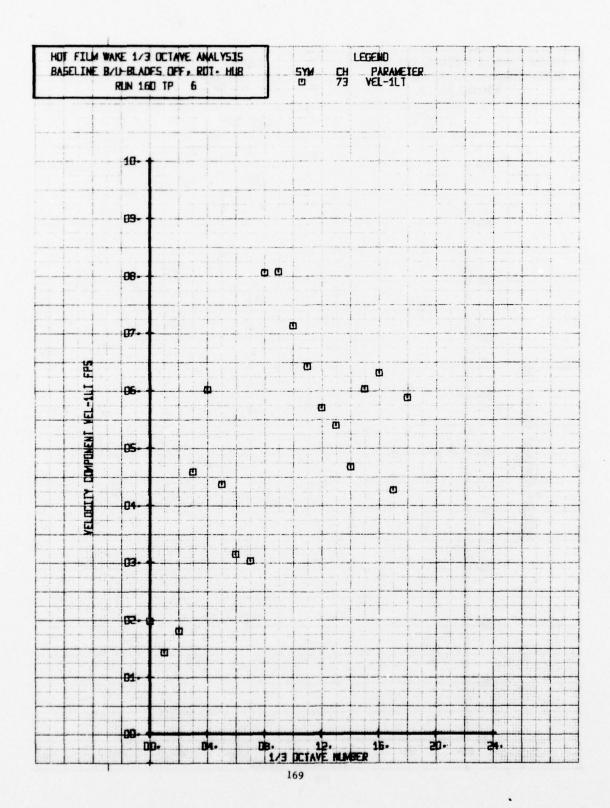


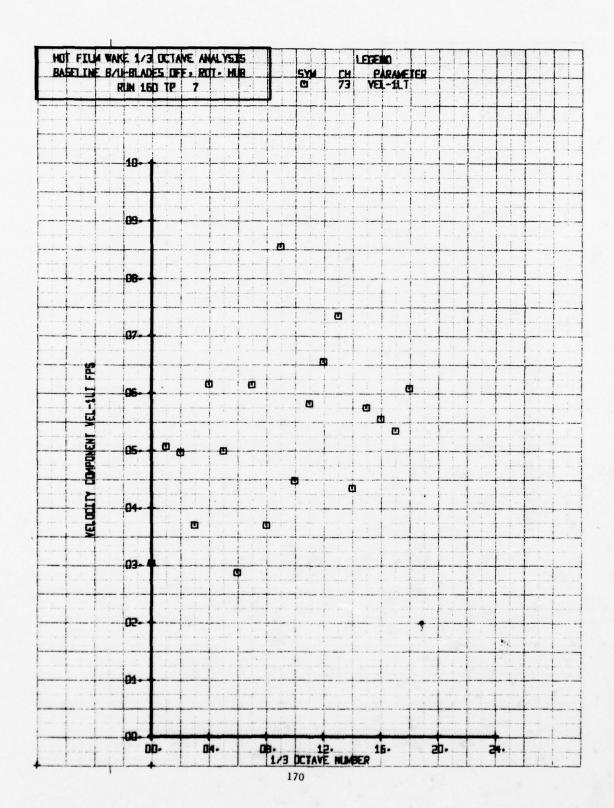


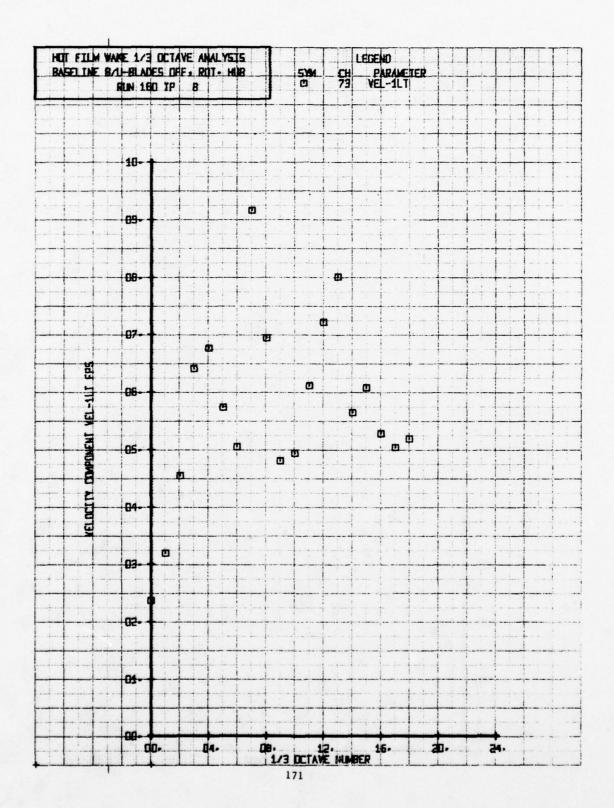


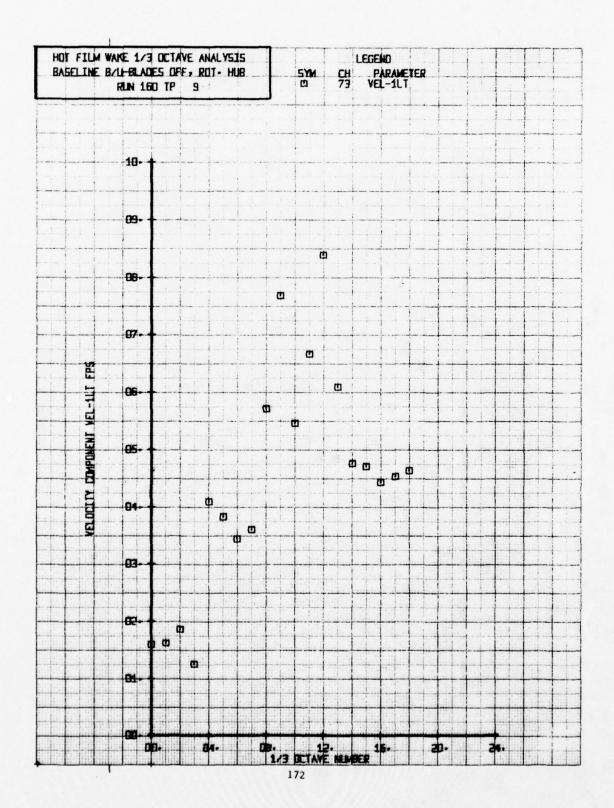


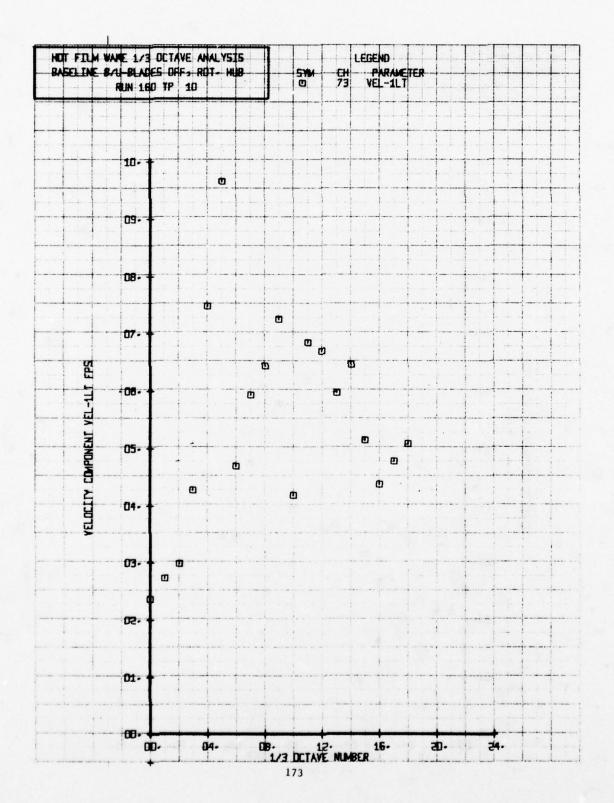


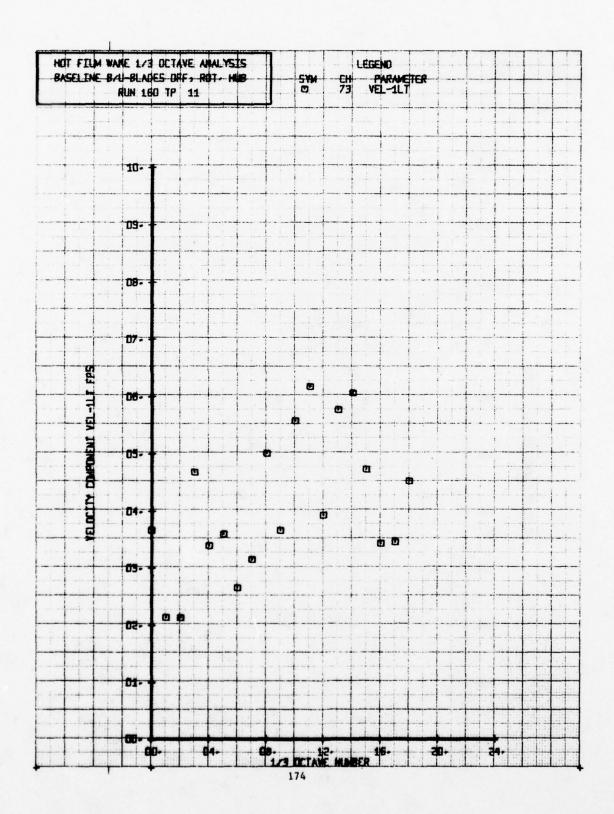


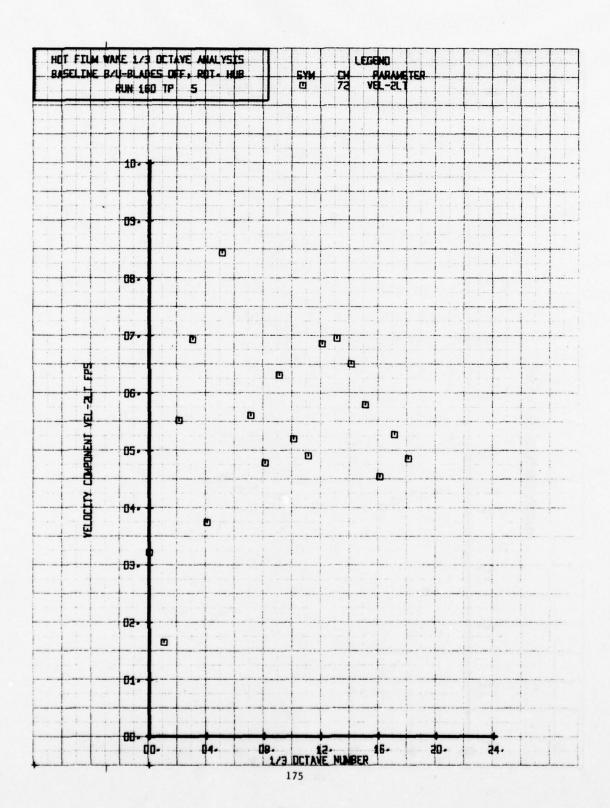


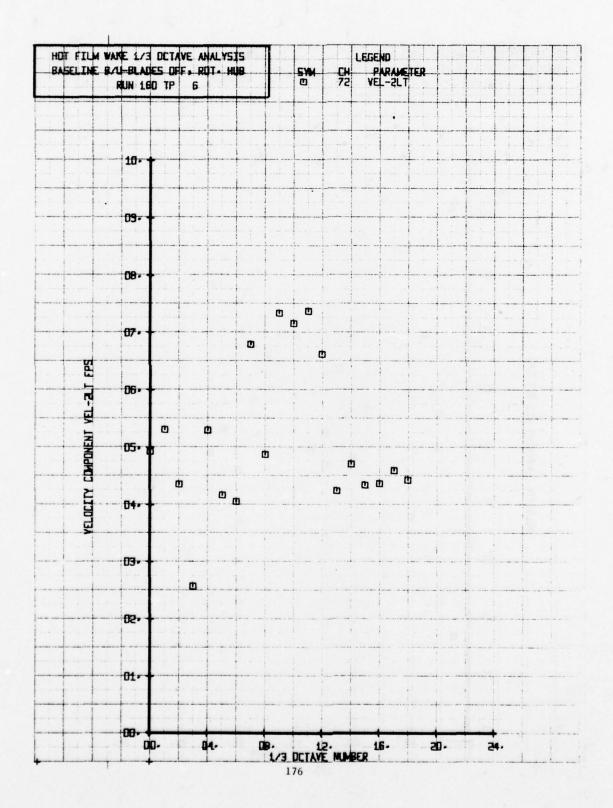


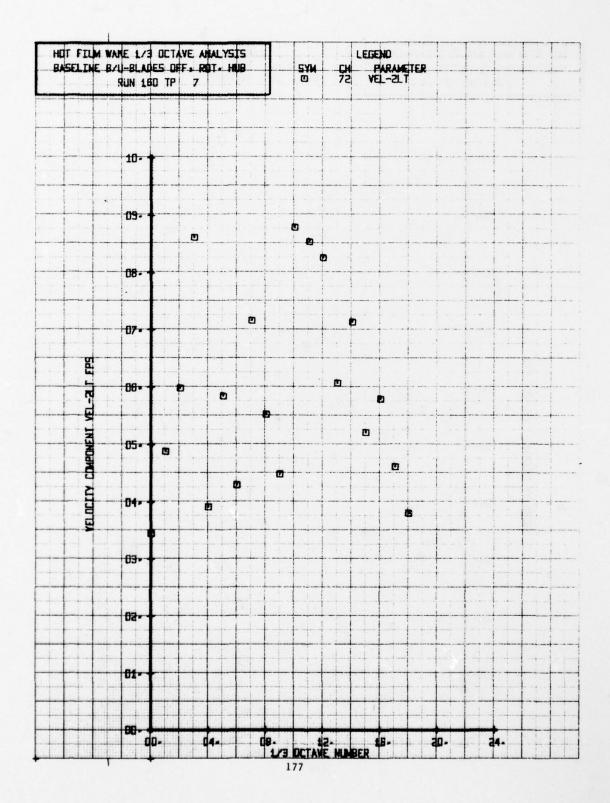


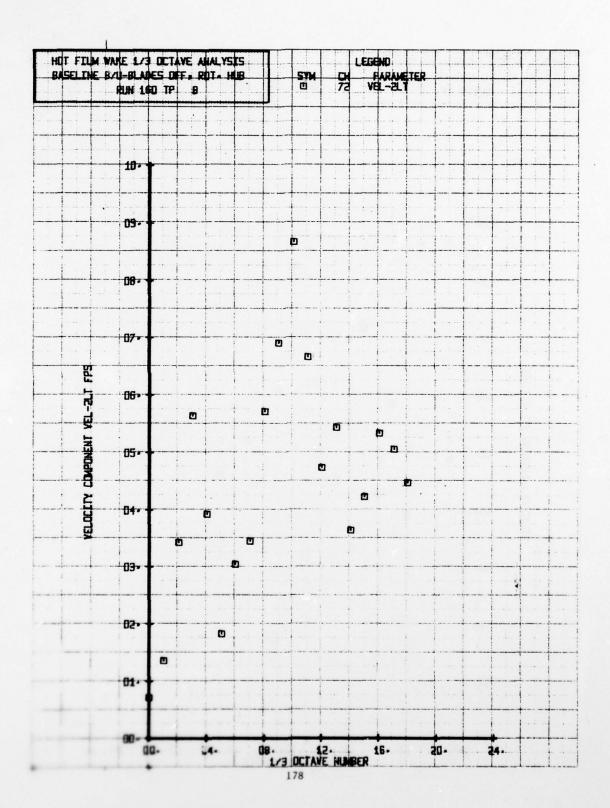


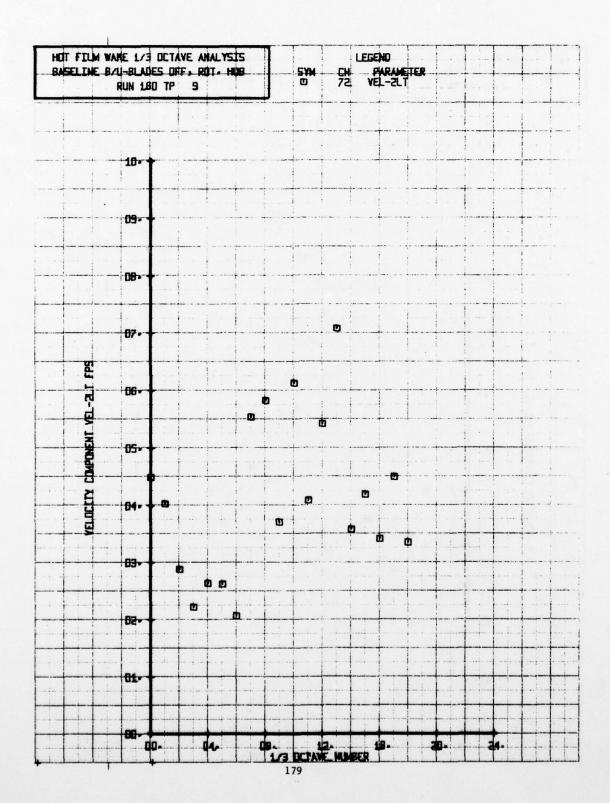


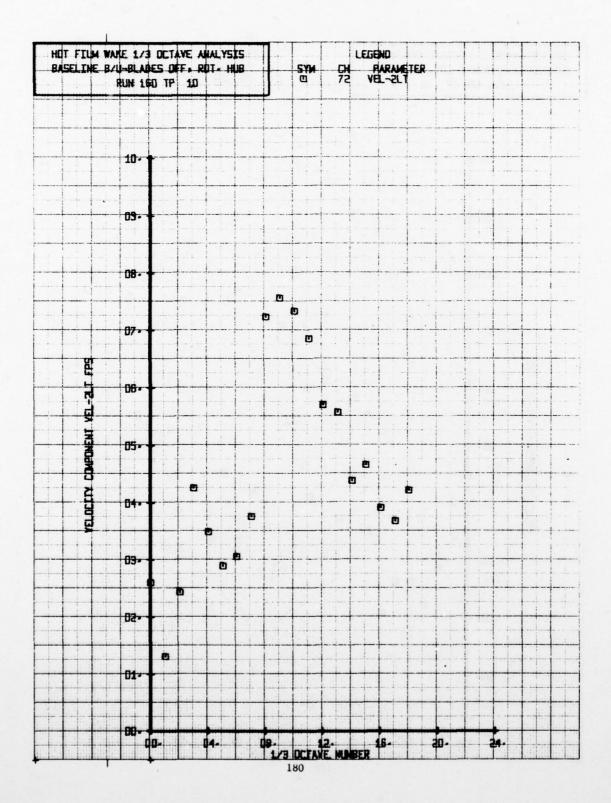


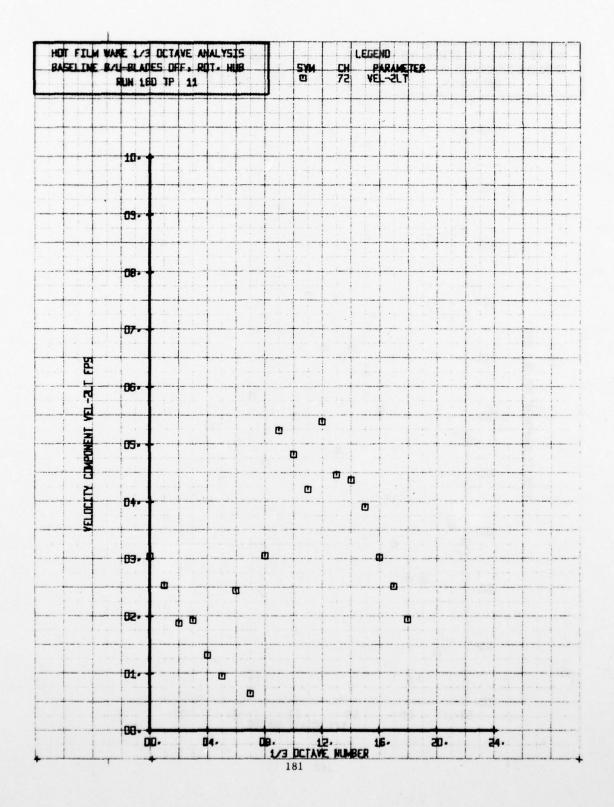


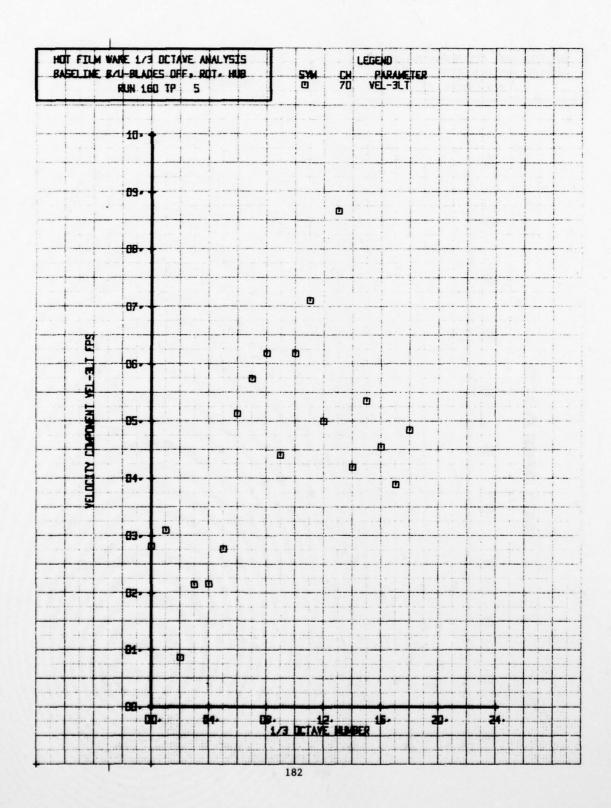


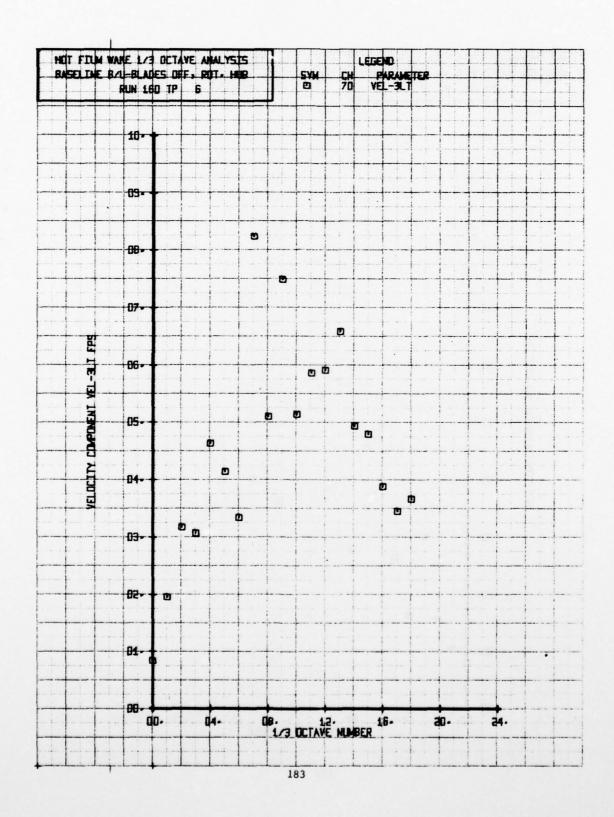


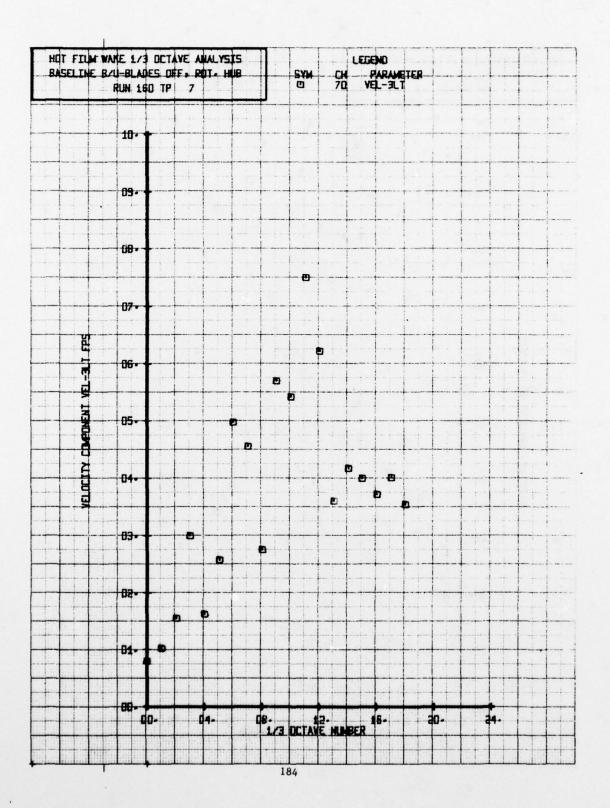


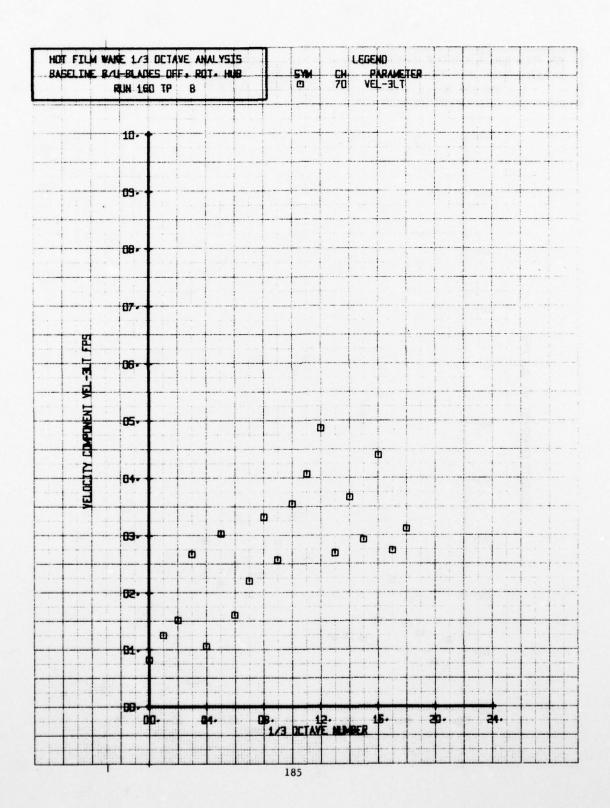


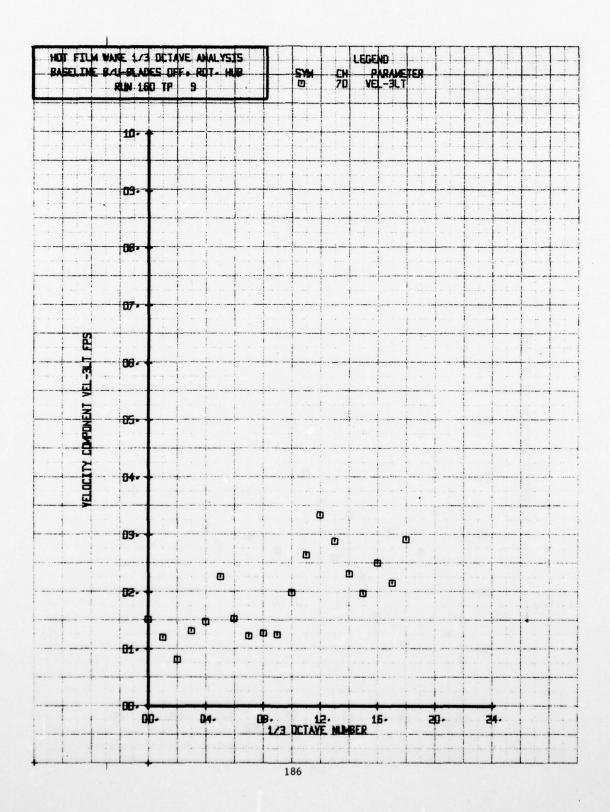


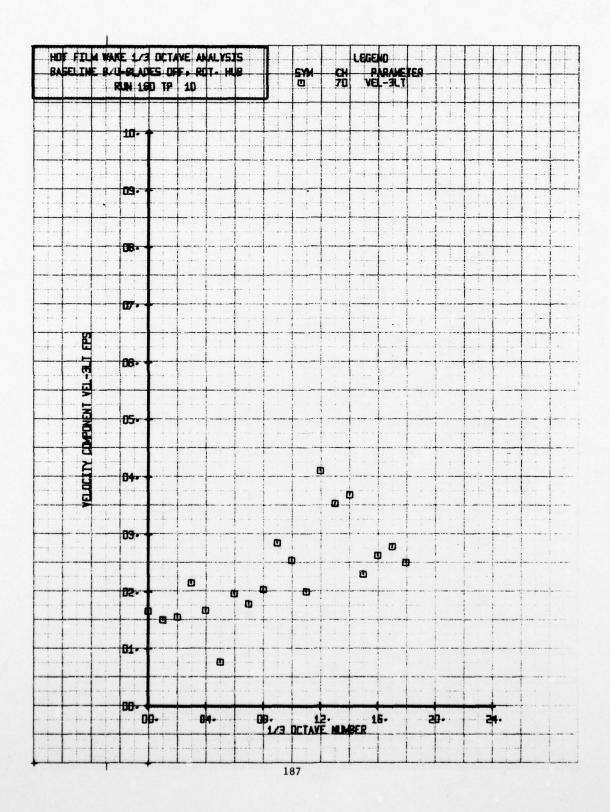


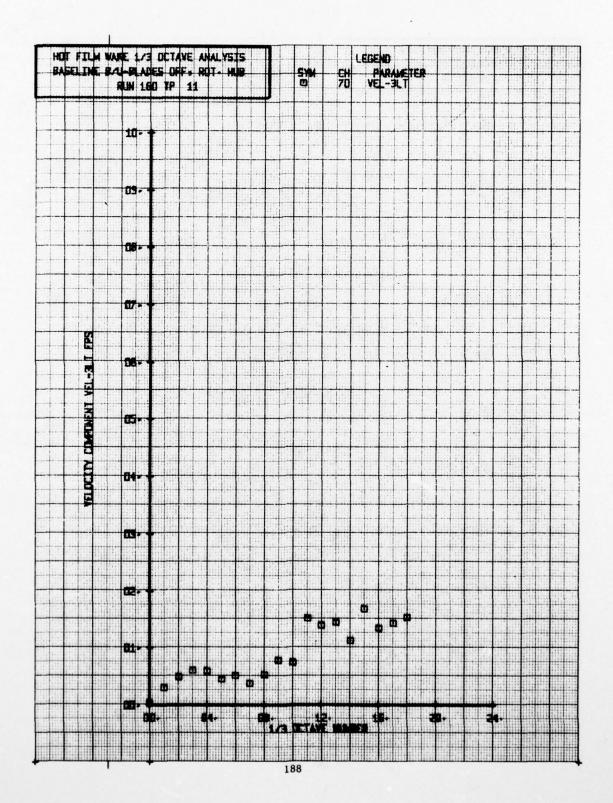


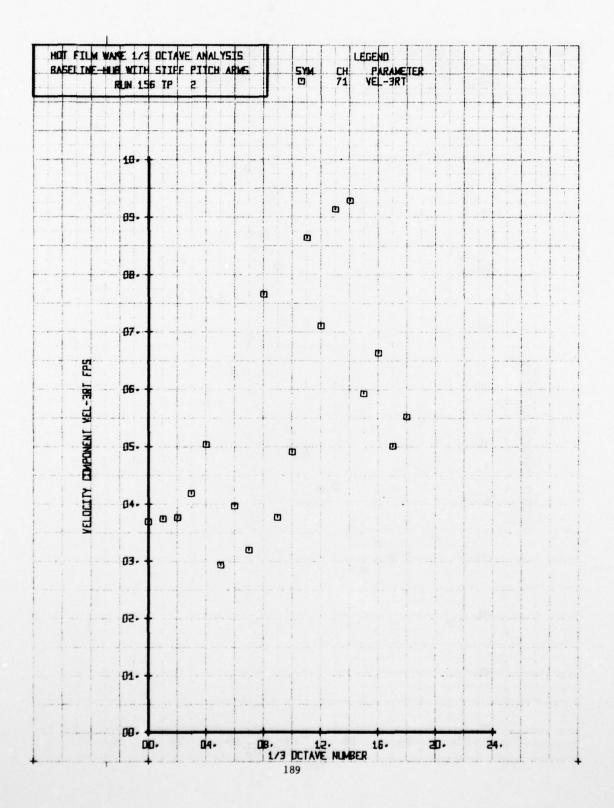


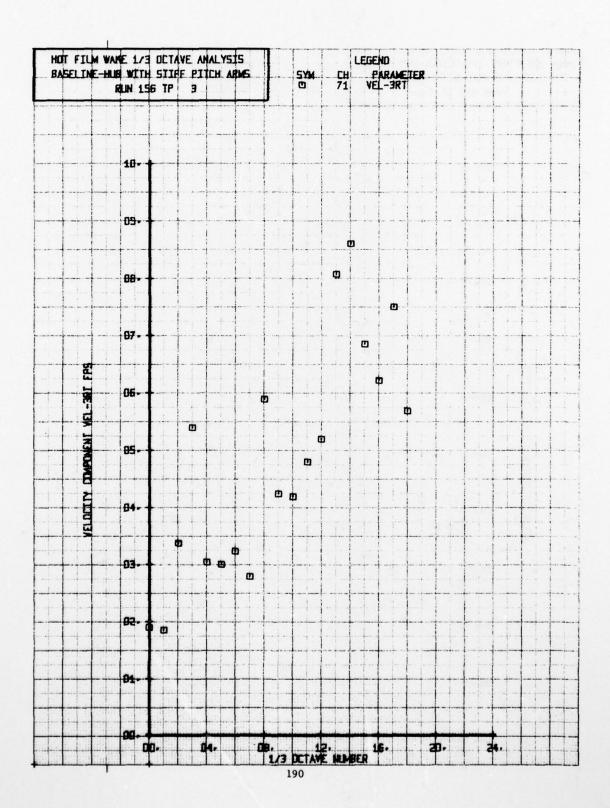


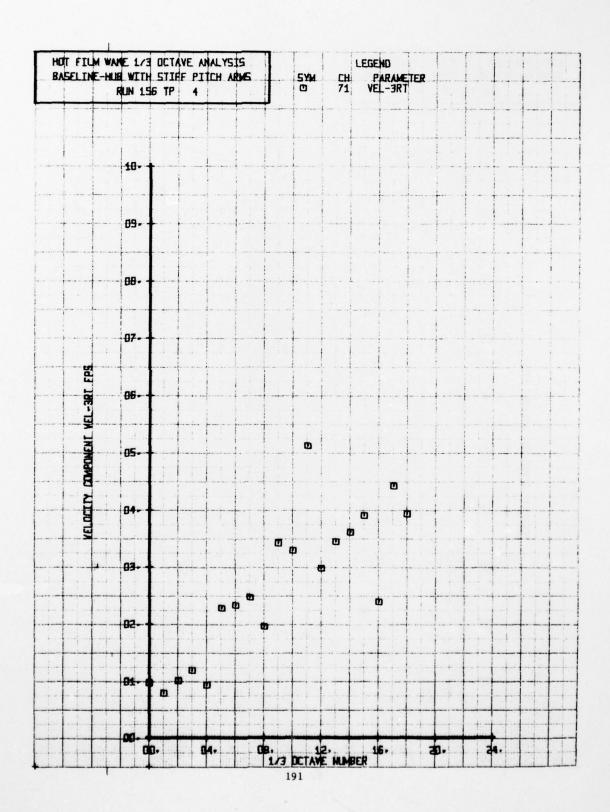












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INTERACTIONAL AERODYNAMICS OF THE SINGLE ROTOR HELICOPTER CONFI--ETC(U)
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